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# RESEARCH MEMORANDUM

for the

Bureau of Aeronautics, Department of the Navy

PRELIMINARY PERFORMANCE DATA ON WESTINGHOUSE

ELECTRONIC POWER REGULATOR OPERATING ON

J34-WE-32 TURBOJET ENGINE IN ALTITUDE

WIND TUNNEL

By James R. Ketchum, Darnold Blivas, and George J. Pack

Lewis Flight Propulsion Laboratory Cleveland, Ohio

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### NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

## RESEARCH MEMORANDUM

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PRELIMINARY PERFORMANCE DATA ON WESTINGHOUSE ELECTRONIC

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## INTRODUCTION

The behavior of the Westinghouse electronic power regulator operating on a J34-WE-32 turbojet engine was investigated in the NACA Lewis altitude wind tunnel at the request of the Bureau of Aeronautics, Department of the Navy. The object of the program was to determine the steady-state stability and transient characteristics of the engine under control at various altitudes and ram pressure ratios, without afterburning. Recordings of the response of the following parameters to step changes in power lever position throughout the available operating range of the engine were obtained: ram pressure ratio, compressor-discharge pressure, exhaust-nozzle area, engine speed, turbine-outlet temperature, fuel-valve position, jet thrust, air flow, turbine-discharge pressure, fuel flow, throttle position, and boost-pump pressure.

Representative preliminary data showing the actual time response of these variables are presented. These data are presented in the form of reproductions of oscillographic traces.

#### APPARATUS

Engine. - J34-WE-32 turbojet engine with variable-area exhaust nozzle and WEC-1 afterburner

Minimum fuel flow set to 550 pounds per hour at a windmilling speed of 1500 rpm and an altitude of 2000 feet

Control. - Westinghouse electronic power regulator (part no. 61-F-758-4; serial no. S-CZA-78, modified to correspond to part no. 61-F-758-6 insofar as temperature schedule is concerned)

The original thermocouple harness of nine paralleled short thermocouples was removed and replaced by another group of nine paralleled elements which were emersed 6 inches. The couples were also of equal resistance to insure a more nearly correct average temperature indication.

Test facilities. - Lewis altitude wind tunnel with 20-foot-diameter test section

Air supplied through a ram pipe connected directly to the engine

Instrumentation. - Transient responses of engine variables were recorded on multiple channel, direct-inking, magnetic-motor oscillographs. The recording unit in combination with its amplifier has flat frequency response to approximately 100 cycles per second. The oscillograph chart speed was 2.5 units per second. The following table lists the sensing devices used for indicating variations in the parameters:

Measured quantity	Translent instrum	Steady-state			
١	Sensor	Frequency res- ponse range	instrumentation		
The end on a second		(cycles/sec) 0-5	Chronometric		
Engine speed	Direct-current tacho-	0-5	18		
<u> </u>	meter generator	0-10	tachometer		
Compressor-	Aneroid-type		Bourdon-type		
discharge	pressure sensor with	1	gage		
pressure	strain-gage element	pressure	D3		
Turbine-	Aneroid-type	0-10	Bourdon-type		
discharge	pressure sensor with	<u>}</u>	gage		
pressure	strain-gage element	pressure	Adamas		
Ram pressure	Aneroid-type	0-10	Airspeed		
ratio	pressure sensor with		indicator		
<b>D</b>	strain-gage element	pressure	D		
Boost-pump	Aneroid-type	Indeterminent	Bourdon-type		
pressure	pressure sensor with		gage		
797 63/	strain-gage element	T-3-4	Determent		
Fuel flow (pres-	Aneroid-type	Indeterminent	Rotameter		
sure drop across	pressure sensor with				
fuel nozzles)	strain-gage element				
Air flow (velo-	Aneroid-type	0-10			
city pressure in		]			
ram pipe)	strain-gage element	pressure			
Turbine-outlet	Unshielded loop	0-1			
temperature	thermocouples (five	at sea-level	Nine thermo-		
	in series)	mass flow	couples in paral- lel connected to Brown recorder (Westinghouse con- trol thermocouple harness)		
Fuel-valve	Fuel-valve feedback	0-100	Microammeter at-		
position	potentiometer con-	1	tached to fuel		
	nected to give po-		valve feedback		
	sition indication.		potentiometer		
Exhaust-nozzle	Exhaust-nozzle-area	0-100	Microammeter at-		
area	feedback potentio-		tached to exhaust.		
	meter connected to		nozzle-area feed-		
	give position in- dication		back potentiometer		
Throttle		0-100	Selsyn indicator		
position	Wire-wound potentio- meter connected to	0-100	Deteli Innterent		
POPTATOR	give position in-		No.		
	1				
Manage of the	dication	0-100			
Thrust	Strain gage mounted on strain link at-	0~100			
	tached to forward		Supplication of the Control of the C		
	engine suspension				

#### PROCEDURE

Accelerations and decelerations, in the non-afterburning region, were produced by manually advancing or cutting back the throttle in a stepwise manner. Oscillograms of the transients were taken. In addition, steady-state readings, before and after each transient, were recorded on both steady-state and transient instrumentation. Various size steps were made throughout the operating range of the engine at the following flight conditions:

Simulated altitude (ft)	Nominal inlet temperature (°F)	Nominal ram pressure ratio
10,000	48	1.20
<b>25,000</b>	-14	1.05
25,000	-10	1.20
25,000	8	1.60
35,000	-15	1.20
40,000	-10	1.20
45,000	<b>-1</b> 0	1.20

## PRESENTATION OF DATA

The preliminary data presented are in the form of reproductions of oscillograms, which have been reduced to 71 percent of their original size. The data are representative of the operation of the controlled engine over its range of operation.

For all runs the following parameters are shown:

Ram pressure ratio
Compressor-discharge pressure
Exhaust-nozzle area
Engine speed
Turbine-outlet temperature
Fuel-valve position

For some of the runs the following additional parameters are shown:

Jet thrust
Air flow
Turbine-discharge pressure
Fuel flow
Throttle position
Boost-pump pressure

The average steady-state value for the preceding parameters are indicated for calibration purposes on the individual oscillograms, except for the jet-thrust and air-flow records. These traces are shown only to indicate this variation during a transient. Because of the techniques employed in measuring jet thrust, variations in ram pressure influence the thrust trace. This effect of ram variation on thrust can be observed by noting the thrust trace when the ram varies.

As an aid in the understanding and interpretation of the transient runs, a nominal power control lever schedule for static sea-level operation is presented in figure 1(a). This figure shows the temperatures and speeds scheduled for each throttle position. A calibration of nozzle area to its panel reading is presented in figure 1(b).

The oscillograms of the transient data are presented in figures 2 to 63 and are indexed in table I according to altitude, nominal ram pressure ratio, throttle position, engine speed, and turbine-outlet temperature. The figures chosen are representative of the operation of the controlled engine throughout the range of throttle settings.

Lewis Flight Propulsion Laboratory,
National Advisory Committee for Aeronautics,
Cleveland, Ohio, October 10, 1950.

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## PRELIMINARY PERFORMANCE DATA ON WESTINGHOUSE

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WIND TUNNKL

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Scientist.

TABLE I - INDEX TO TRANSIENT RUNS

Fig- ure	Altitude (ft)	ram pressure	Throttle position (deg)		Engine speed (rpm)		Turbine outlet temp- erature, OF	
-		ratio	Initial		Initial		Initial	Final
2(a) 2(b)	10,000	1.2	22 30	30 30	5,210 5,780	5,780 5,210	365 <sub>1</sub> 405	405 365
3(a)			22.5	29.5	5,490	5,640	440	430
3(b)			29.5	22.5	5,640	5,490	440	430
4(a)			35.0	35.5	7,840	8,000	445	450
4(b)	,		35.5	35.0	8,000	7,840	450	445
5(a) 5(b)	ł .		43.5 49.0	49.0 43.5	10,900	10,900	460 475	475 460
6(a)			51	62	11,360		480	515
6(b)	į		62	51	11,930		515	480
7(a)	1	·	65	70.5		12,570	580	810
7(b)			70.5	65	12,570		810	580
8(a)			65	84	12,280	12,500	610	1,215
8(b)			84	65		12,280	1,215	610
9(a) 9(b)			22 35	35 22	5,150 7,620	7,620 5,150	380 450	450 <b>3</b> 80
10			42	22	9,950	5,410	450	400
11(a)			22	49	5,420	11,180	380	480
11(b)			49	22	11,180	5,420	480	480
12(a)			22	65	5,410	12,420	405	690
12(b)			65	22	12,420	5,410	690	570
13(a)			22	84		12,580	415	1,210
13(b)			84	52	12,580	5,420	1,210	430
14(a)	25,000	1.05	22	35	6,750		725	550
14(b)			35	22	7,590	6,840	550	725
15(a)			36.5	37.5	7,930	8,320	520	520 520
15(b)   16(a)		ł .	37.5 41.5	36.5 43	8,320 9,970	7,930	520 460	520
16(b)	1		43	41.5	10,450	9,970	470	
17		<b>l</b> ,	42.5	51	10,730	11,400	480	505
18		, ,	62	70	11,940	12,520	550	690
19(a)			6 <b>2</b>	83	12,520		560	1,190
19(b)	1		83	62	12,680	12,520	1,190	560
20		<u> </u>	22	42	6,890	9,980	710	490
21	1	1	22	52		11,020	700	510
22		[	22.5	70 85	7,140	12,520 12,690	720 740	675 1,190
23(a) 23(b)			22 85	22	12,690	7,050	1,190	740
120(0)				22	12,000	,,500	19100	. =0
	<u> </u>				<u> </u>	<b></b>	l	A COLUMN THE PARTY OF THE PARTY

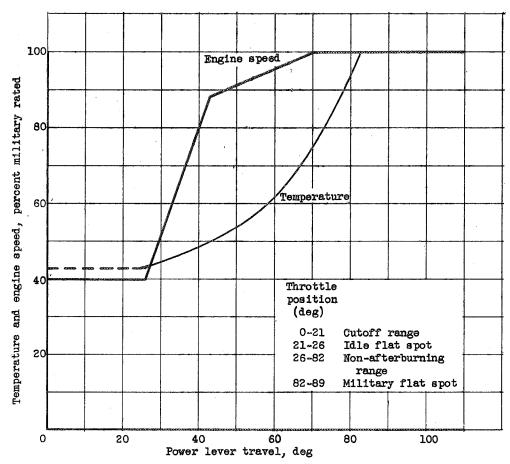
TABLE I - INDEX TO TRANSIENT RUNS - Continued

Fig-	Altitude	Nominal	Throttle		Engine speed (rpm)		Turbine outlet temp- erature, OF	
ure	(ft)	, ram pressure						
		ratio	Initial	Final	Initial		Initial	Final
24	25,000	1.2	22	42	California de la companya de la comp	10,000	540	cali- bration
25(a)			26 72	72 26	6,900	12,500	490 770	770 <b>4</b> 90
25(b) 26(a)			31.5	33.5·	12,500 6,700	6,900 cali-	540	cali- bration
26(b) 27(a)			33.5 34	31 . 5 37	7,140 7,810	bration 6,700 8,290	435 390	540 cali-
	·		37	34	8,290	7,810	380	bration cali-
27(b)			35	35 <sub>.</sub> 5		•	425	bration 405
28 29(a)			35 85	85 35		7,540	475	1,172
29(b) 30(a)			41 85	85 41	12,560		1,172 375	475 1,215 375
30(b) 31			43.5 43.5	44 55	12,540	10,910	1,215	430 445
32(a) 32(b) 33			55 55	43.5 85	11,460	11,460 10,950 12,540	405 445 435	calib. 1,215
34 35(a)			70 20 <sub>•</sub> 5	8 <b>5</b> 35	12,540 6,340	12,550	780 505	1,210
35(b) 36(a)			35 20.5	20.5 83	7,360	6,340	425 600	505 1,180
36(b)			83	20.5	12,520	7,280	1,180	600
37(a) 37(b)	25,000	1.6	21 36.5	36.5 21	7,500 8,790		370 340	340 370
38(a) 38(b)			33 41	41 33	7,760 9,630	9,630	410 305	305 410
39(a) 39(b)			33 62	62 33	7,900 11,940	11,940	415 460	<b>46</b> 0 415
40(a) 40(b)			40 41	41 40	9,600 10,030	10,030	280 300	<b>300</b> 280
41(a) 41(b)		:	62 71	71 62	12,000 12,960	12,960	465 800	800 465
42(a) 42(b)			70 84	84 70		bration		1,220 805
43(a) 43(b)			34 84	84 34	8,380 12,580	12,580 8,380	1,220	1,220 440
				<u></u>	<u> </u>		<u> </u>	

TABLE I - INDEX TO TRANSIENT RUNS - Concluded

Fig- ure	Altitude (ft)	Nominal ram pressure	Throttle position (deg)		Engine speed (rpm)		Turbine outlet temp= erature, OF	
		ratio	Initial		Initial		Initial	Final
44(a) 44(b) 45	35,000	1.2	37 40 41.5	40 37 44 <sub>*</sub> 5	10,000	10,000 9,220 11,100	720 570 cali- bration	570 720 <b>44</b> 0
46(a) 46(b) 47(a) 47(b) 48 49(a) 49(b) 50(a) 50(b) 51(a) 51(b)			46.5 64.5 60.5 64 61 84 47 84 36.5 84 84	64.5 46.5 64 60.5 70 84 61 84 47 84 36.5 29	11,860 11,860 12,550 11,240 12,580	11,140 12,100 11,850	450 500 480 510 500 500 1,210 485 1,220 780 1,210 1,150	500 450 510 480 685 1,210 500 1,220 485 1,210 780 840
53 54(a) 54(b) 55(a) 55(b) 56(a) 56(b) 57(a) 57(b) 58 59 60	40,000	1.2	42 42 50.5 46.5 55 47 66 50 62 84 41 84	43.5 50.5 42 55 46.5 66 47 62 50 47 84 39	11,940	11,900 11,420 11,940 11,620 12,490 11,670 12,430 11,940 11,660 12,620	670 650 700 770 820 800 840 710 760 1,205 660 1,220	700 650 cali- bration cali- bration 760 710 810 1,220 735
61 62 63	45,000 45,000 <sup>a</sup> 45,000 <sup>a</sup>	1.2	47 80 80	47 80 84	11,400 12,500 12,500	cali- bration 12,600		1,260 1,135 1,135

a Engine-inlet temperature, +47° F.



(a) Power control lever schedule for static sea-level operation.

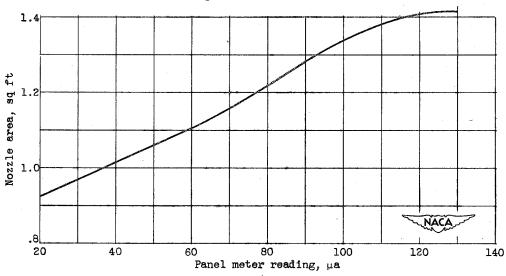
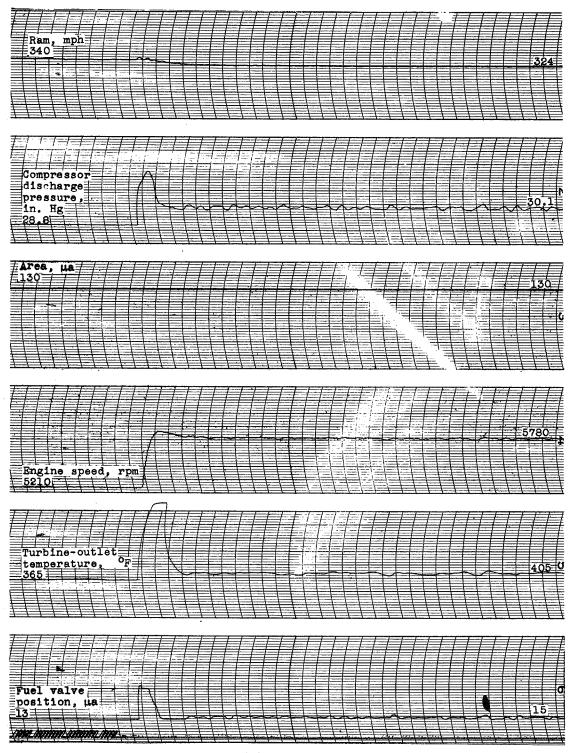


Figure 1. - Power lever and area indicator calibration curves.

(b) Area indicator.



(a) Acceleration.

Figure 2. - Transient operation of automatically-controlled engine. Throttle position, 22° to 30°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.

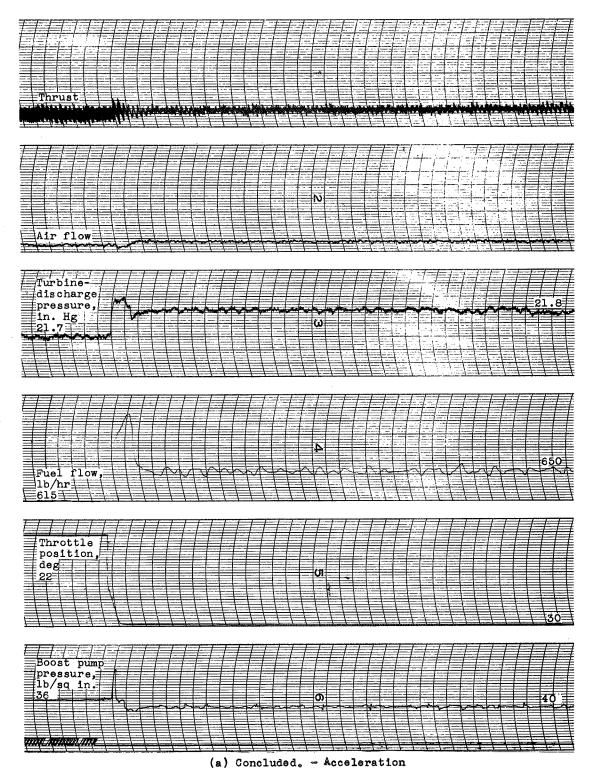
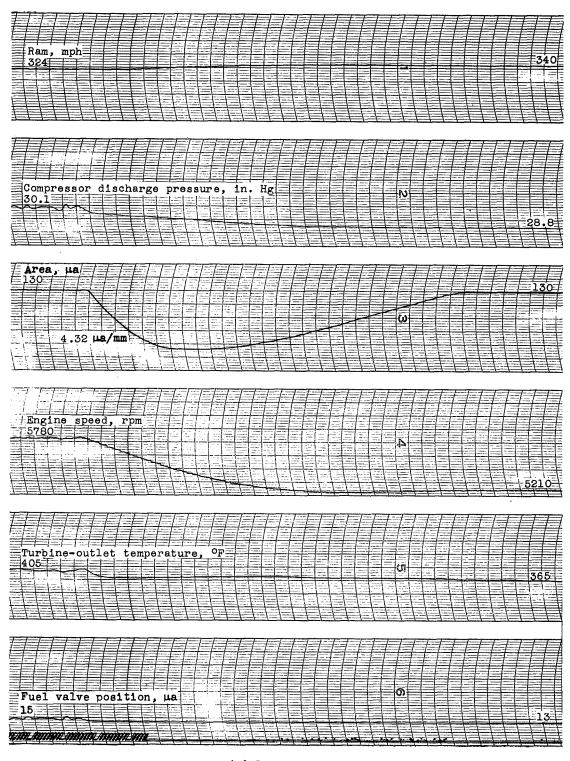


Figure 2. - Continued. Transient operation of automatically-controlled engine. Throttle position, 22° to 30°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.

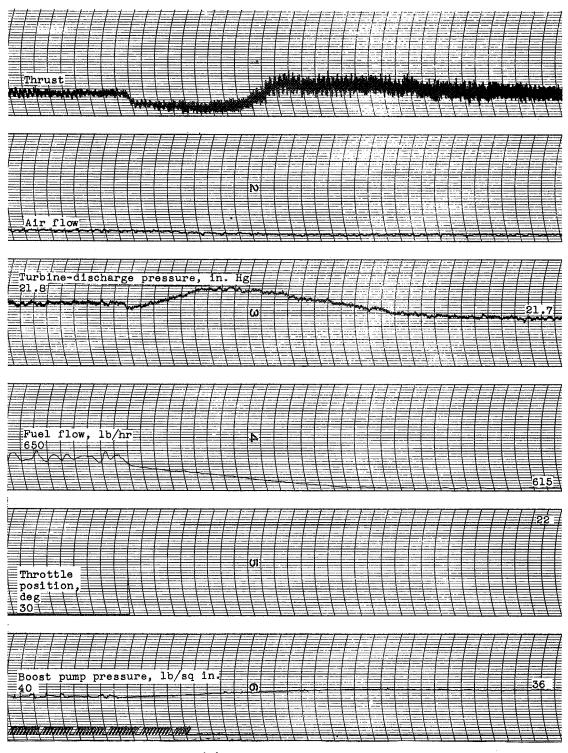
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(b) Deceleration.

Figure 2. - Continued. Transient operation of automatically-controlled engine. Throttle position, 22° to 30°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.





(b) Concluded. - Deceleration.

Figure 2. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 22° to 30°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.



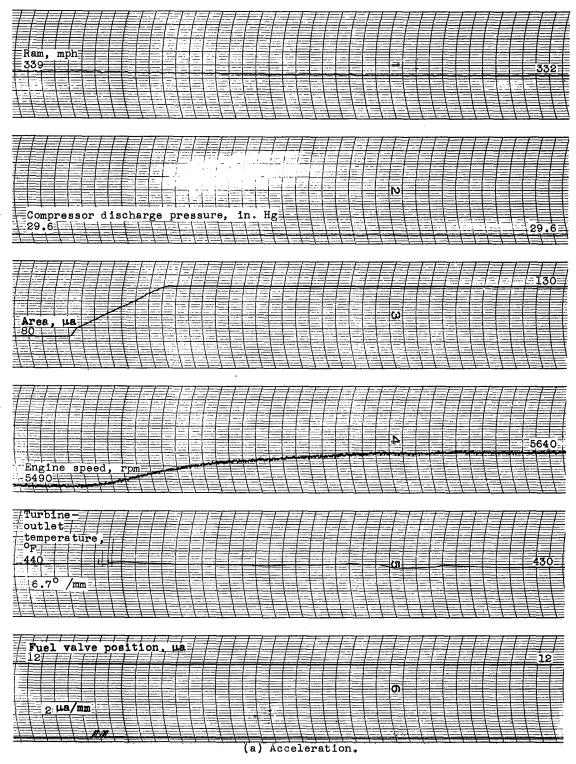


Figure 3. - Transient operation of automatically-controlled engine. Throttle position, 22.5° to 29.5°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.



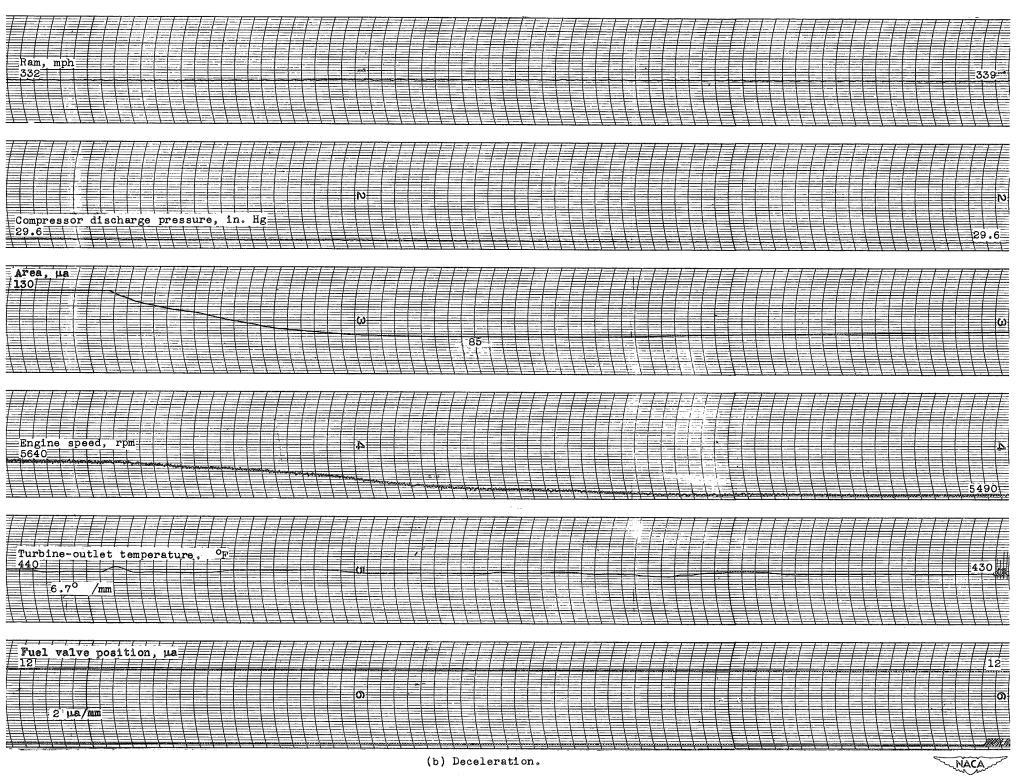
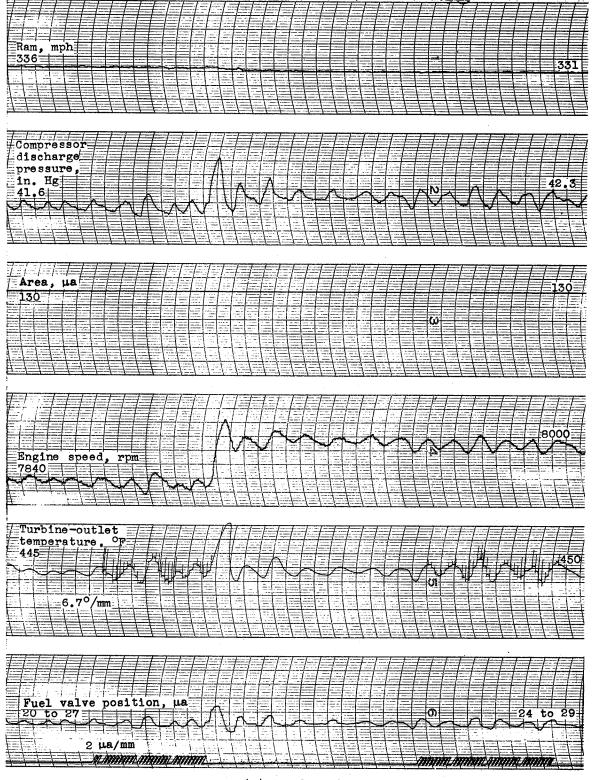


Figure 3. ~ Concluded. Transient operation of automatically~controlled engine. Throttle position, 22.5° to 29.5°; altitude, 10,000 feet; nominal ram~pressure ratio, 1.2.

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(a) Acceleration.

Figure 4. - Transient operation of automatically-controlled engine. Throttle position,  $35.0^{\circ}$  to  $35.5^{\circ}$ ; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.



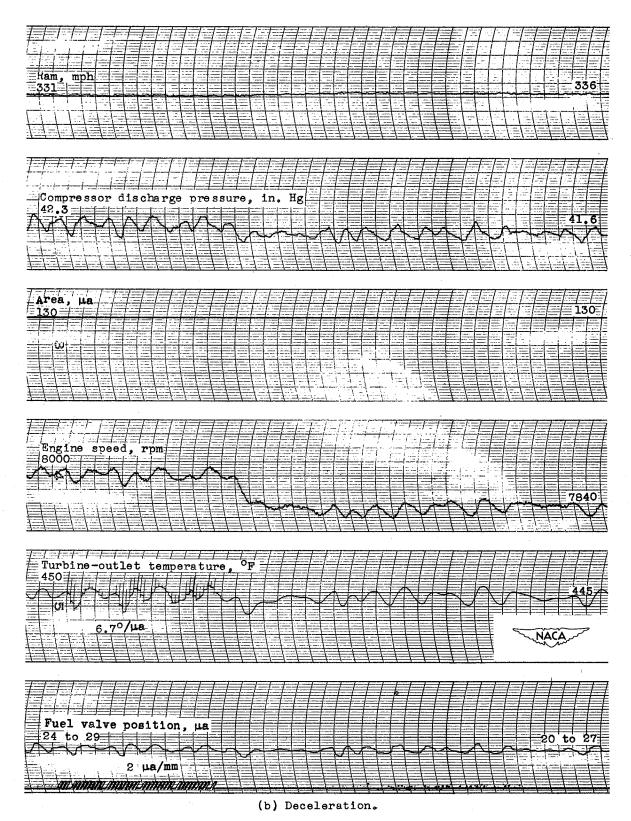


Figure 4. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 35.0° to 35.5°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.

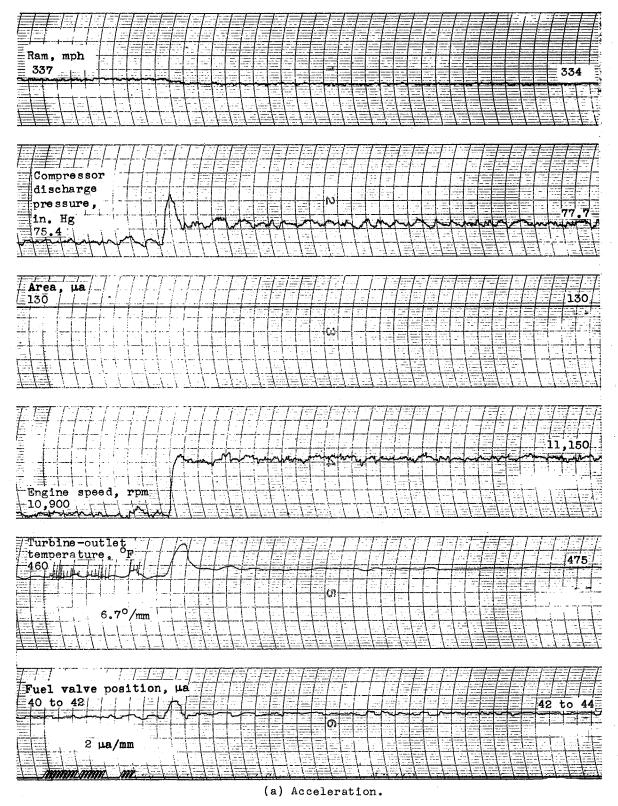
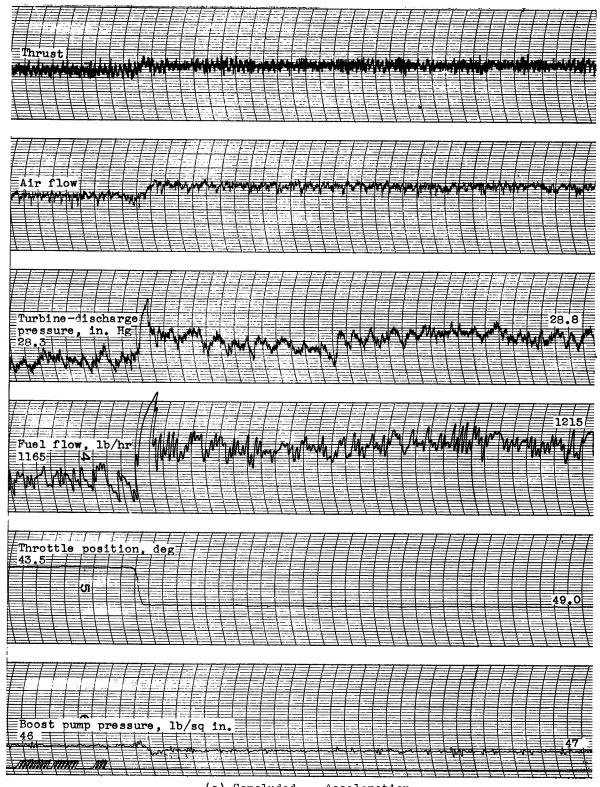


Figure 5. - Transient operation of automatically-controlled engine. Throttle position, 43.5° to 49.0°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.





(a) Concluded. - Acceleration.

Figure 5. - Continued. Transient operation of automatically-controlled engine. Throttle position, 43.50 to 49.00; altitude, 10,000 feet, nominal ram-pressure ratio, 1.2.



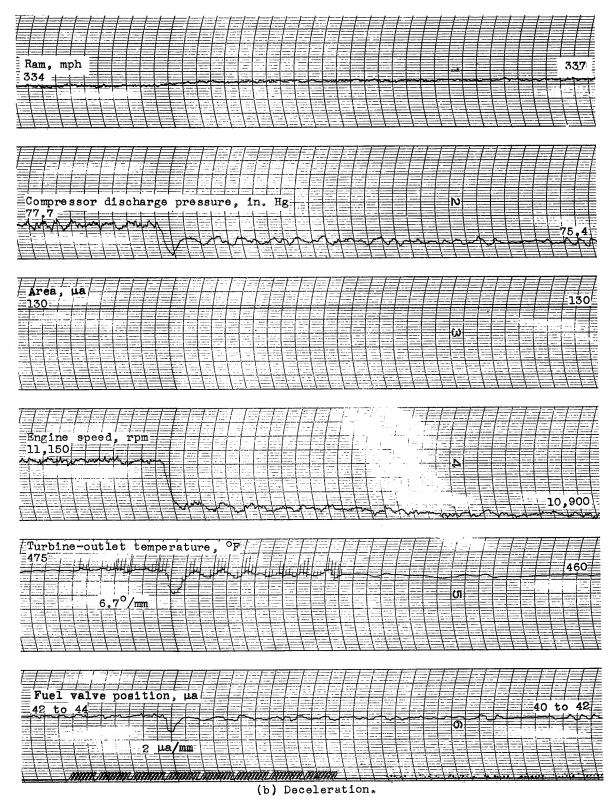


Figure 5. - Continued. Transient operation of automatically∞controlled engine. Throttle position, 43.5° to 49.0°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.

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Figure 5. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 43.5° to 49.0°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.

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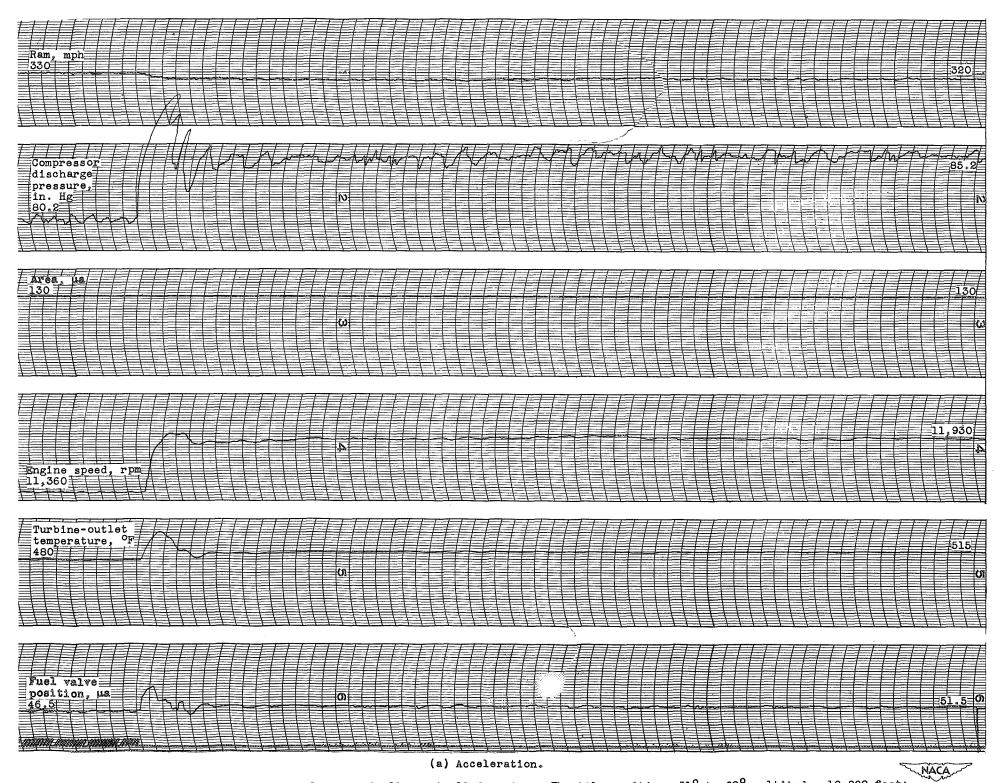


Figure 6. - Transient operation of automatically-controlled engine. Throttle position, 51° to 62°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.

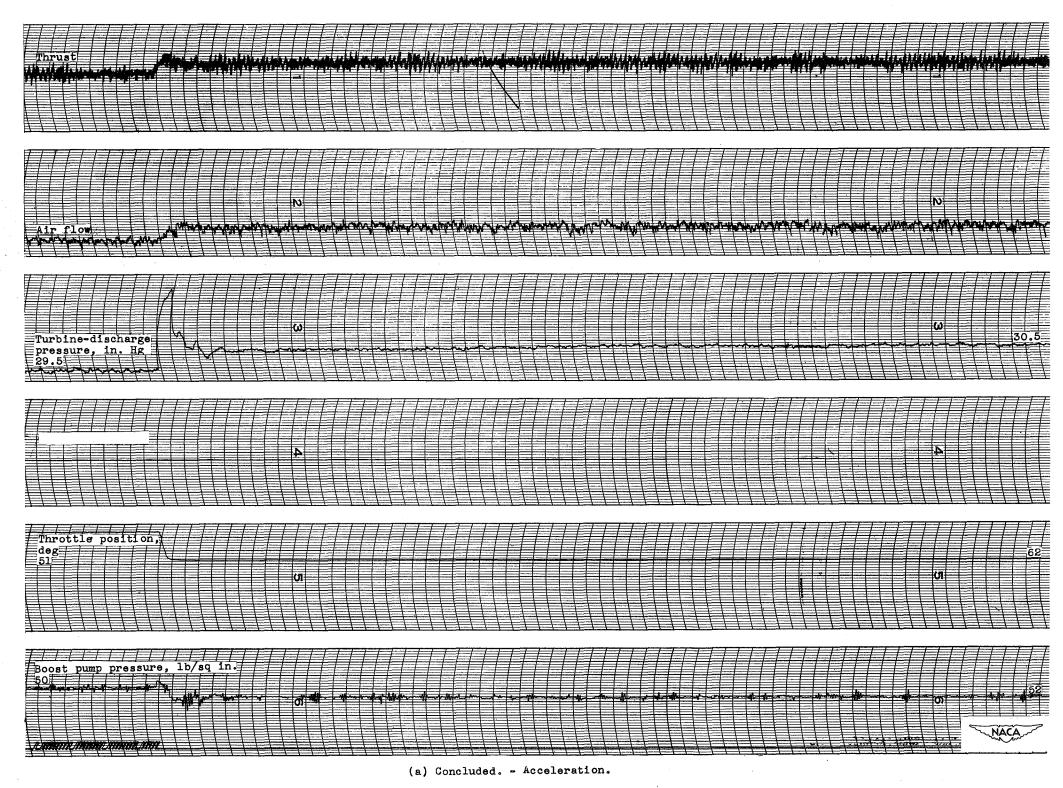


Figure 6. - Continued. Transient operation of automatically-controlled engine. Throttle position, 51° to 62°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.

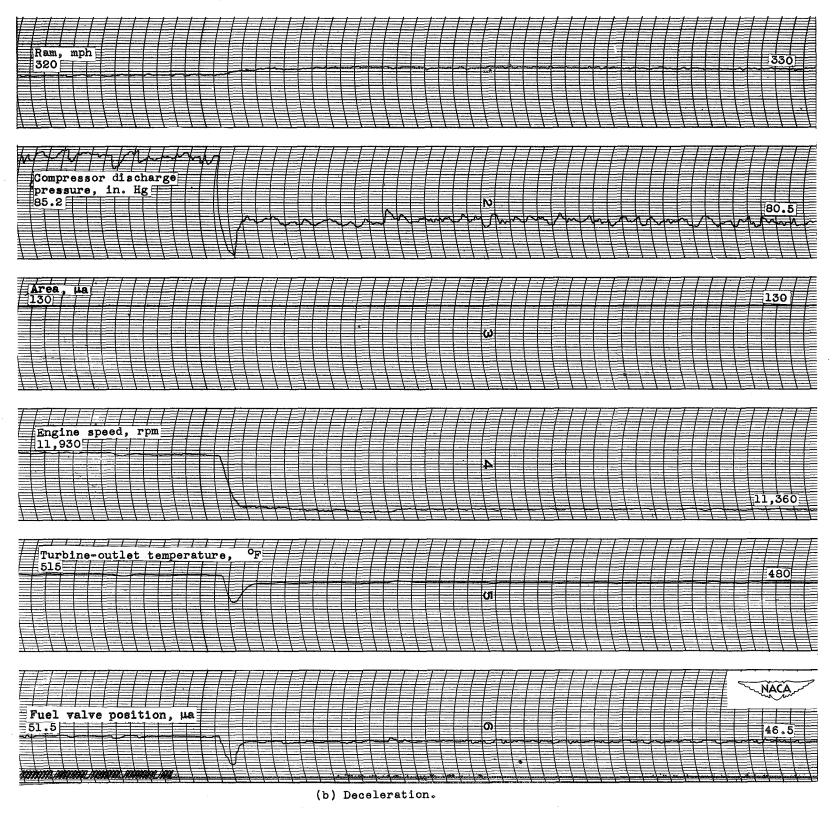


Figure 6. - Continued. Transient operation of automatically-controlled engine. Throttle position, 51° to 62°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.

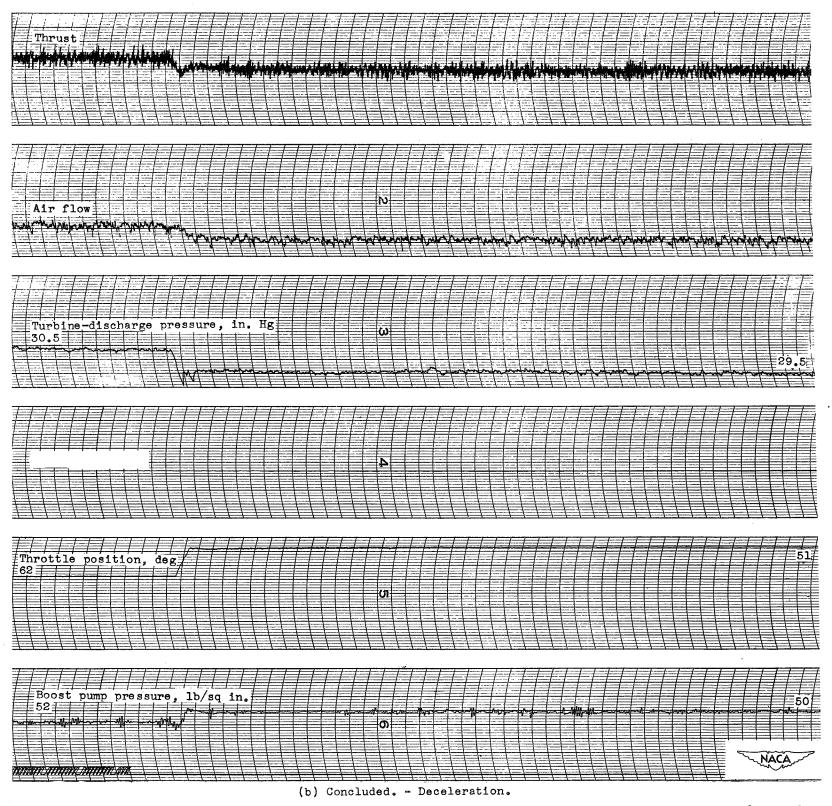


Figure 6. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 51° to 62°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.

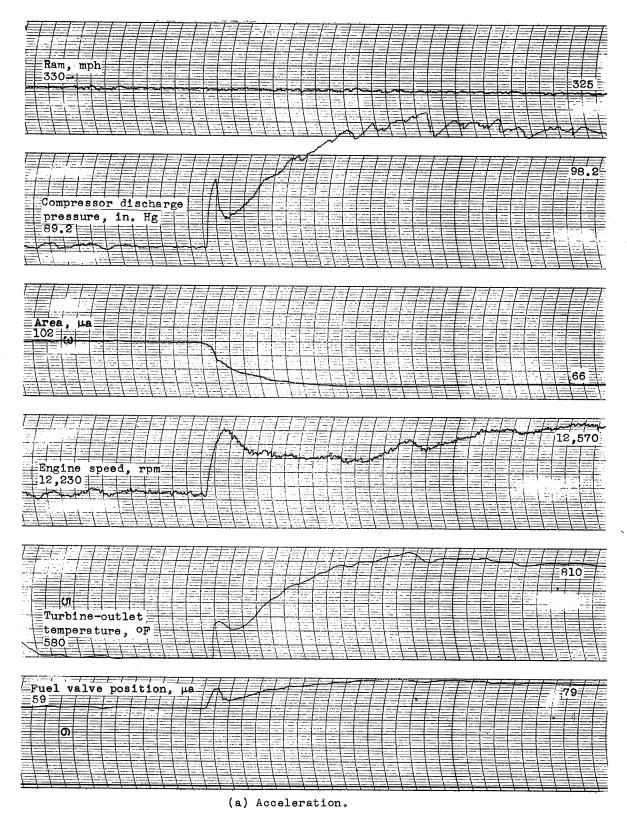


Figure 7. - Transient operation of automatically-controlled engine. Throttle position, 65° to 70.5°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.



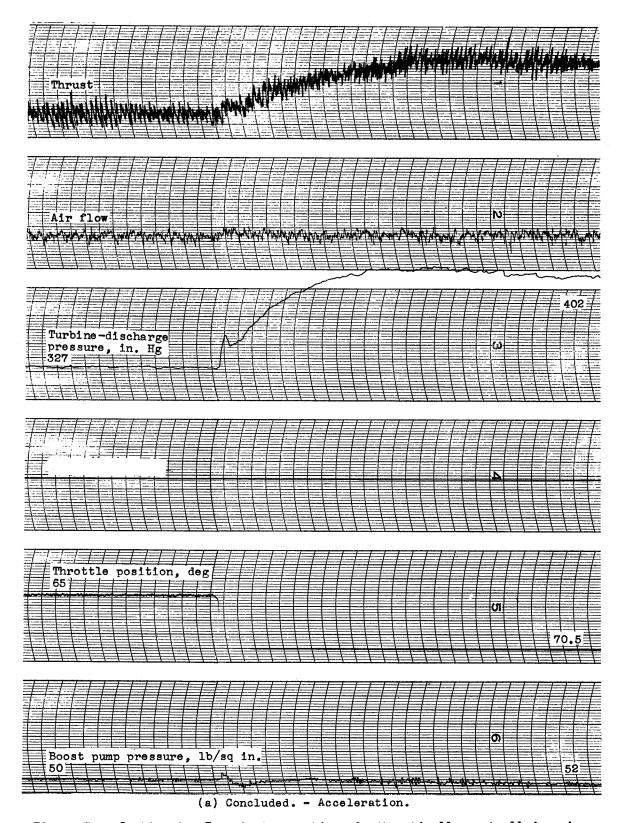


Figure 7. - Continued. Transient operation of automatically-controlled engine. Throttle position, 65° to 70.5°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.



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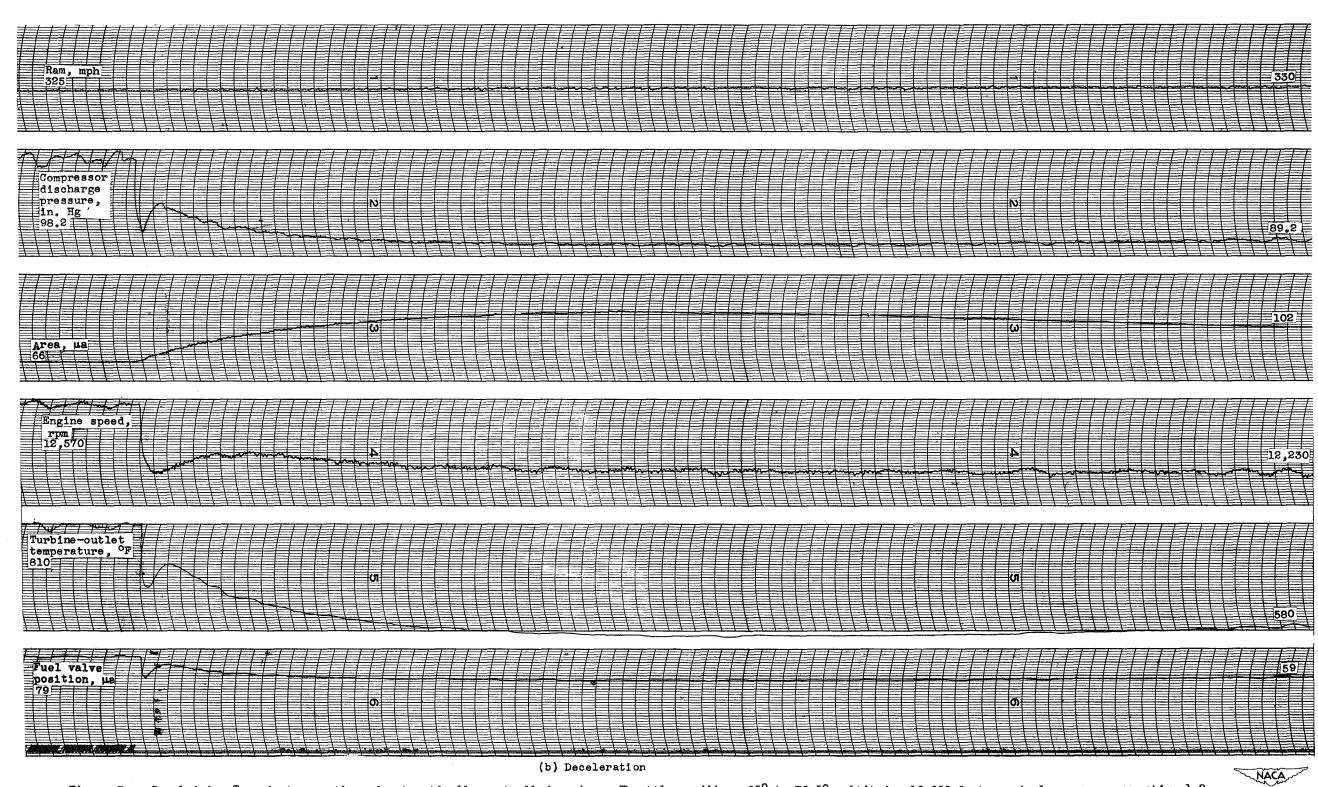


Figure 7. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 65° to 70.5°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.

Figure 8. - Transient operation of automatically-controlled engine. Throttle position, 65° to 84°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.

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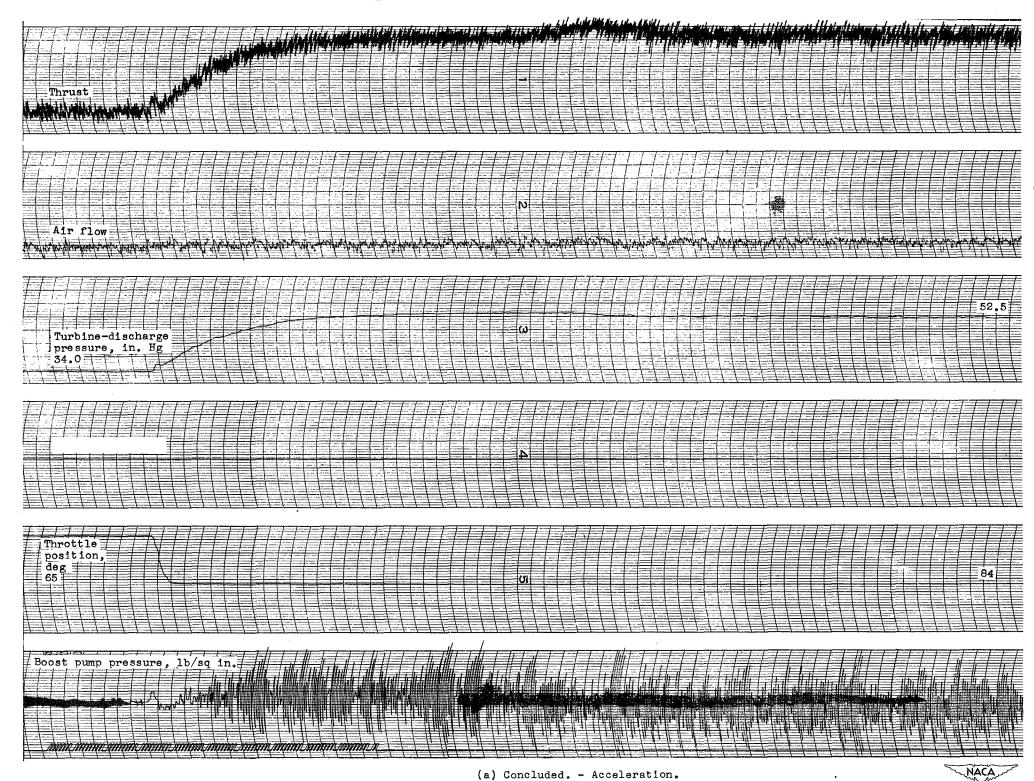


Figure 8. - Continued. Transient operation of automatically-controlled engine. Throttle position, 65° to 84°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.

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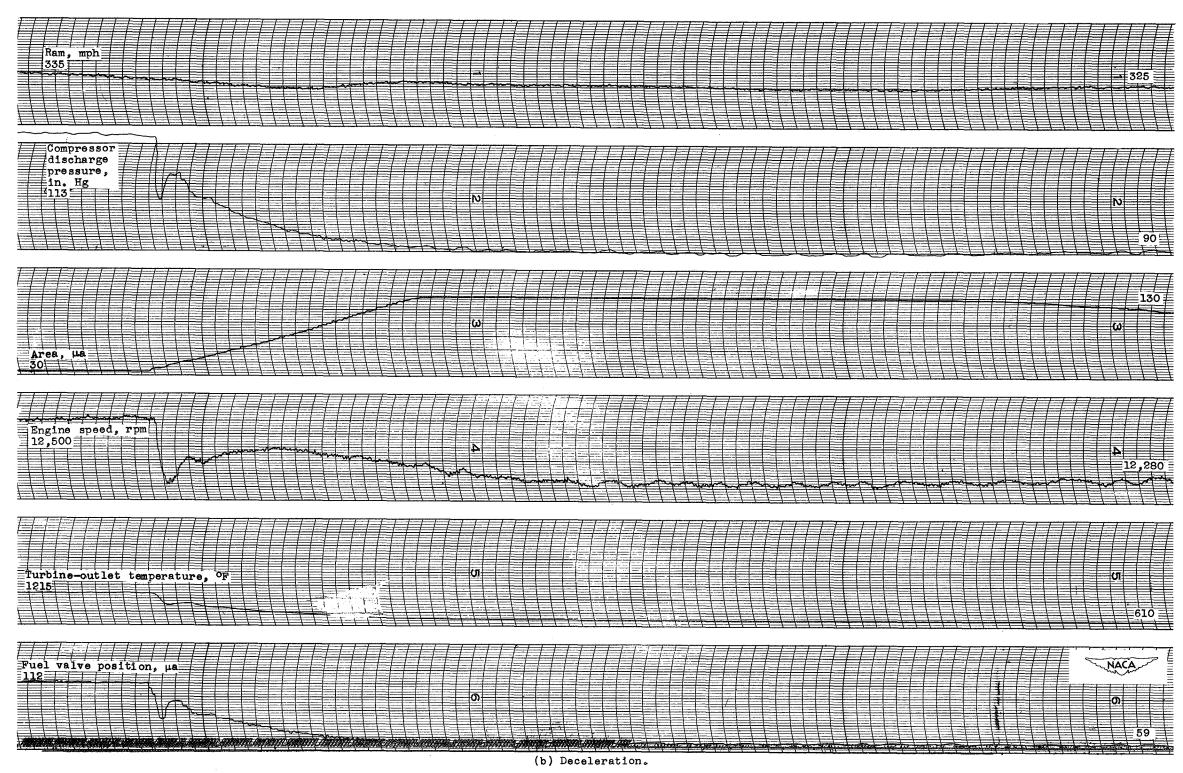


Figure 8. - Continued. Transient operation of automatically-controlled engine. Throttle position, 65° to 84°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.

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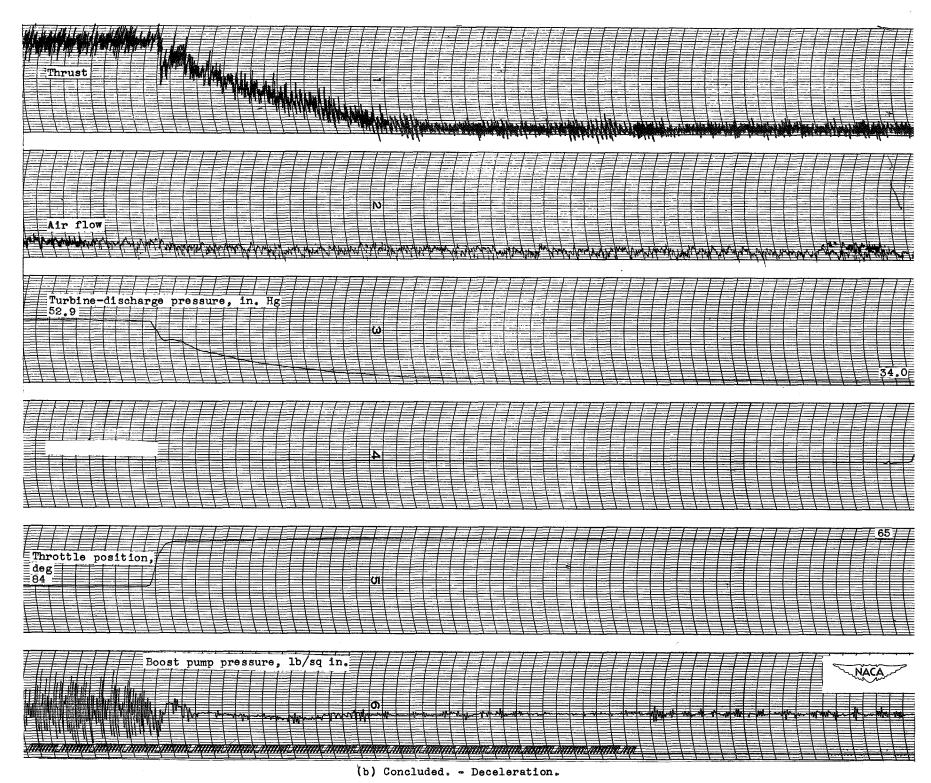


Figure 8. - Concluded. Transient operation of automatically-controlled engine. Throttle pogition, 65° to 84°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.

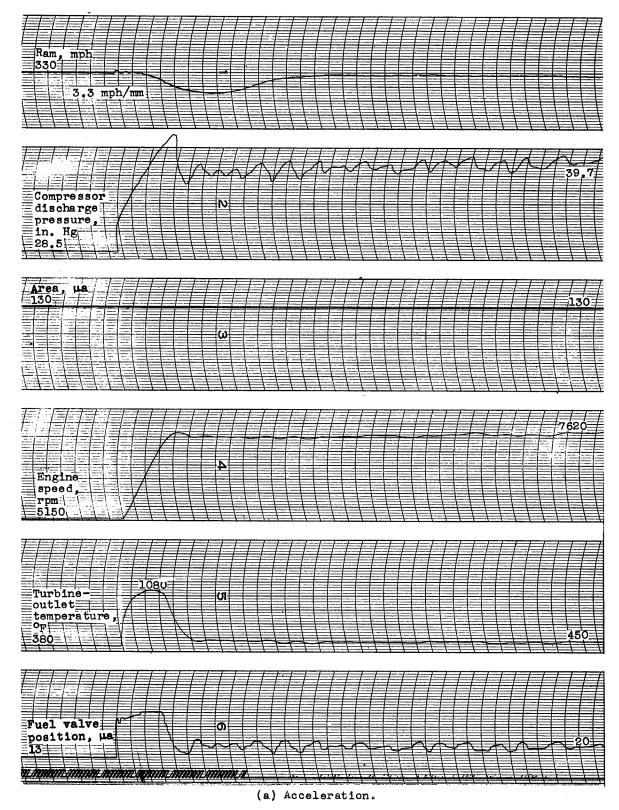


Figure 9. - Transient operation of automatically-controlled engine. Throttle position, 22° to 35°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.

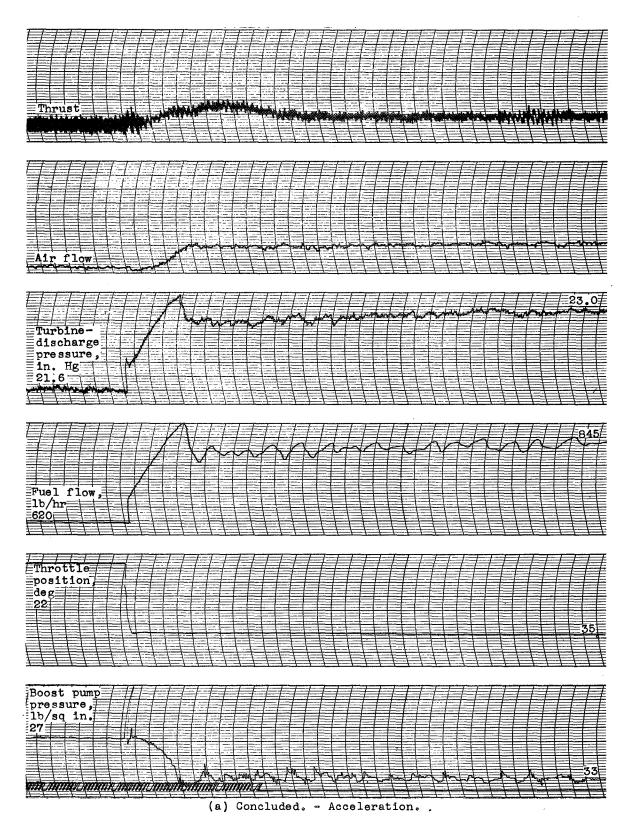


Figure 9. - Continued. Transient operation of automatically-controlled engine. Throttle position, 22° to 35°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.



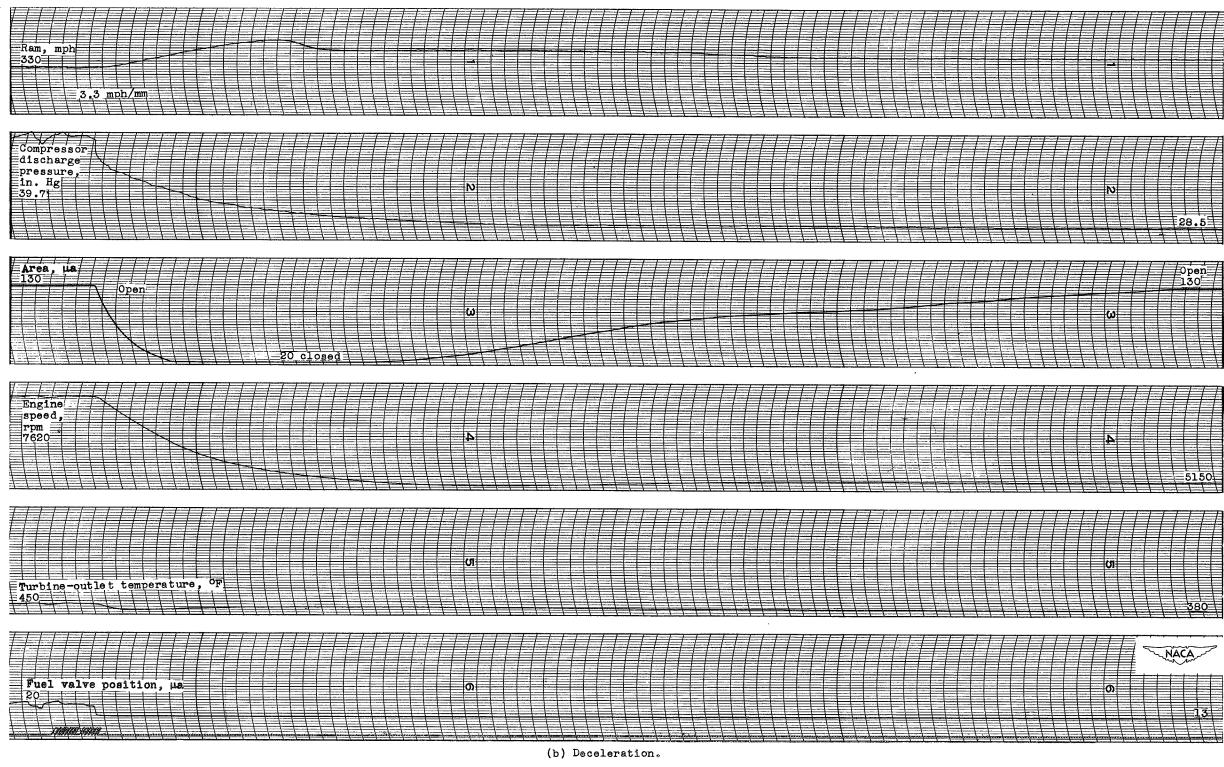


Figure 9. - Continued. Transient operation of automatically-controlled engine. Throttle position, 22° to 35°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2,

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Figure 9. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 22° to 35°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.

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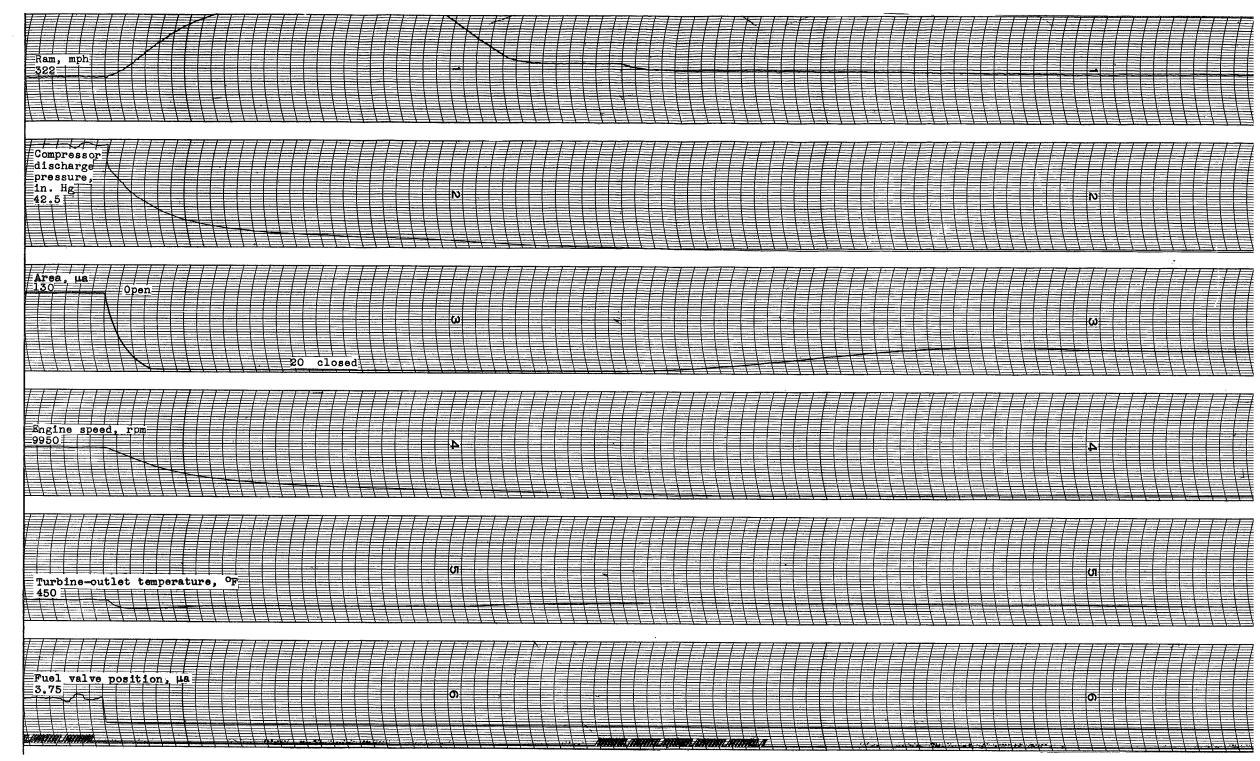
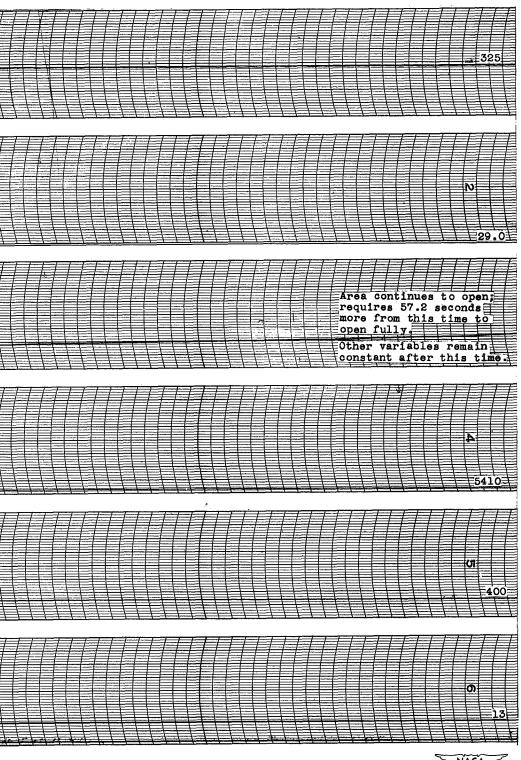


Figure 10. - Transient deceleration of automatically-controlled engine. Throttle position, 42° to 22°; altitude, 10,000 feet; nominal ram-pr

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ssure ratio, 1.2;.

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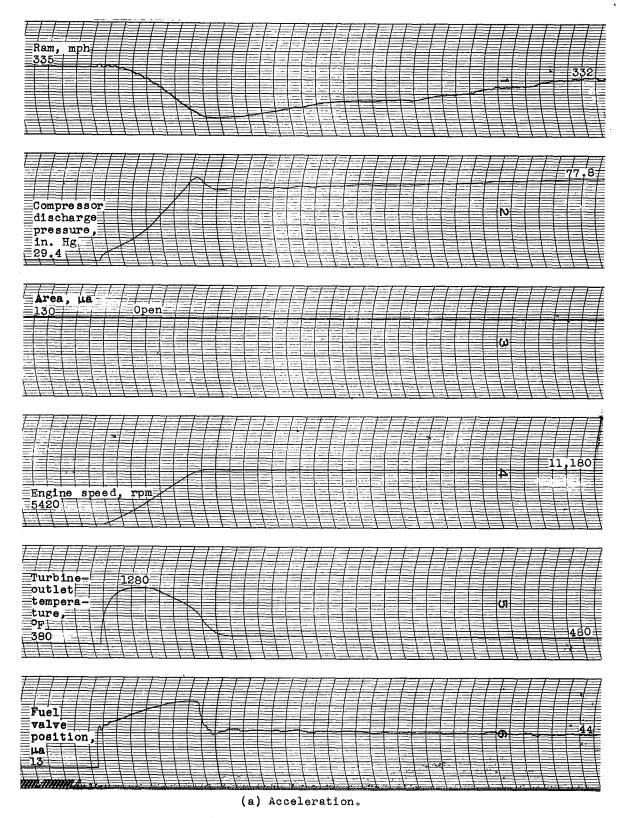
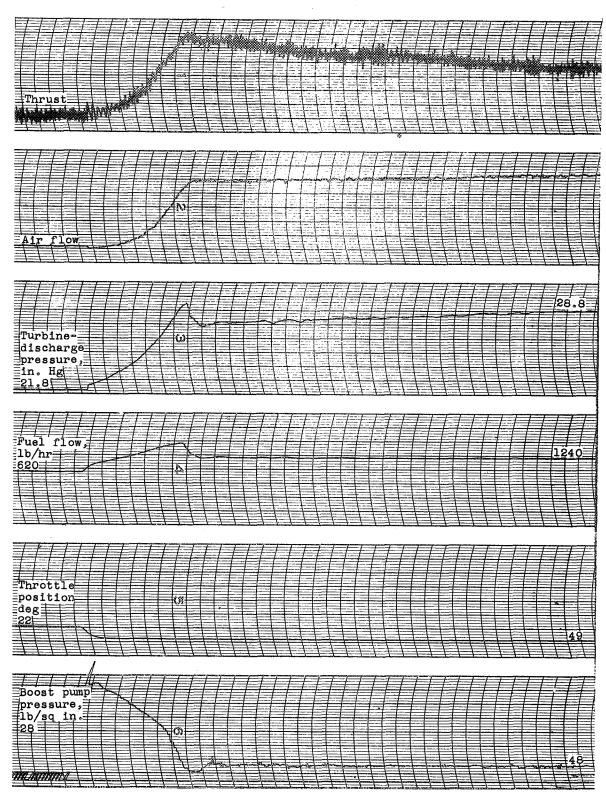


Figure 11. - Transient operation of automatically-controlled engine. Throttle position, 22° to 49°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.





(a) Concluded. - Acceleration.

Figure 11. - Continued. Transient operation of automatically-controlled engine. Throttle position, 22° to 49°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.



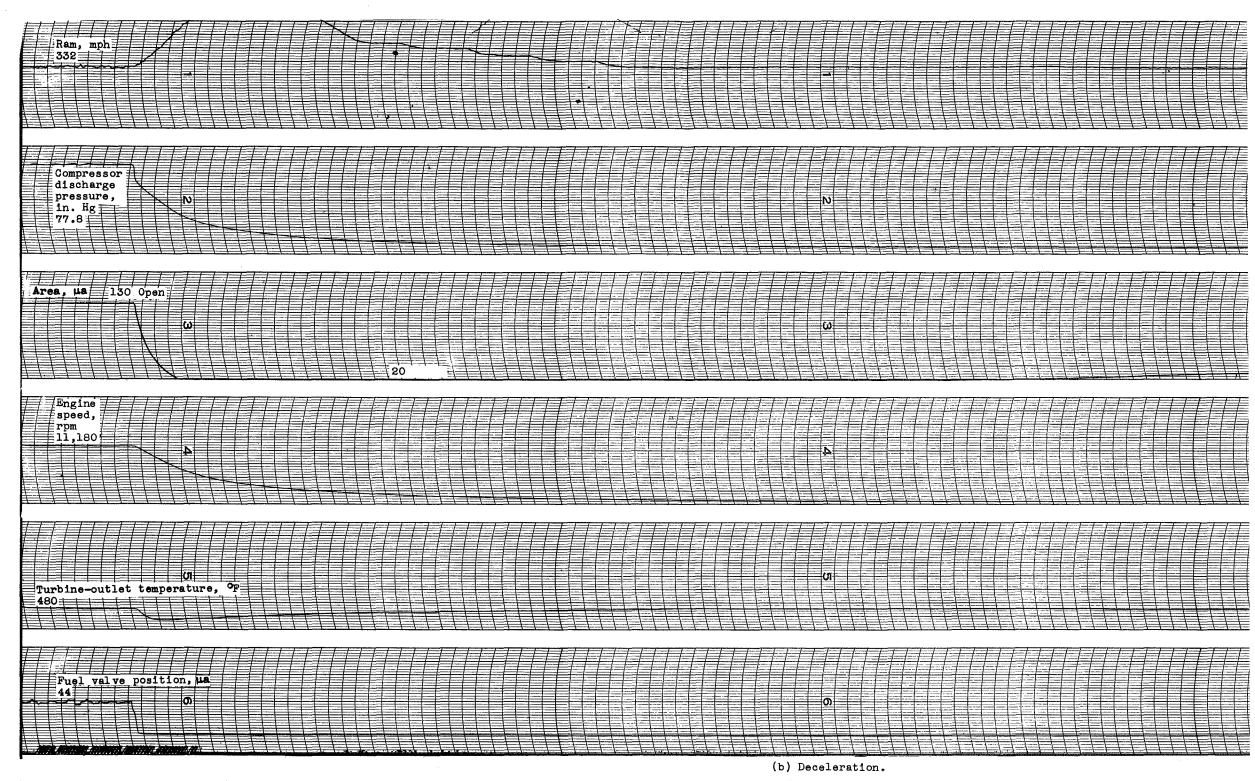


Figure 11. - Continued. Transient operation of automatically-controlled engine. Throttle position, 22° to 49°; altitude, 10,000 feet; nomin CONFIDENTIAL

335 N Area continues to open; requires 75 seconds more from this time to open fully. ω Other variables remain constant after this time. UI 480 Ŏ 13

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il ram-pressure ratio, 1.2.

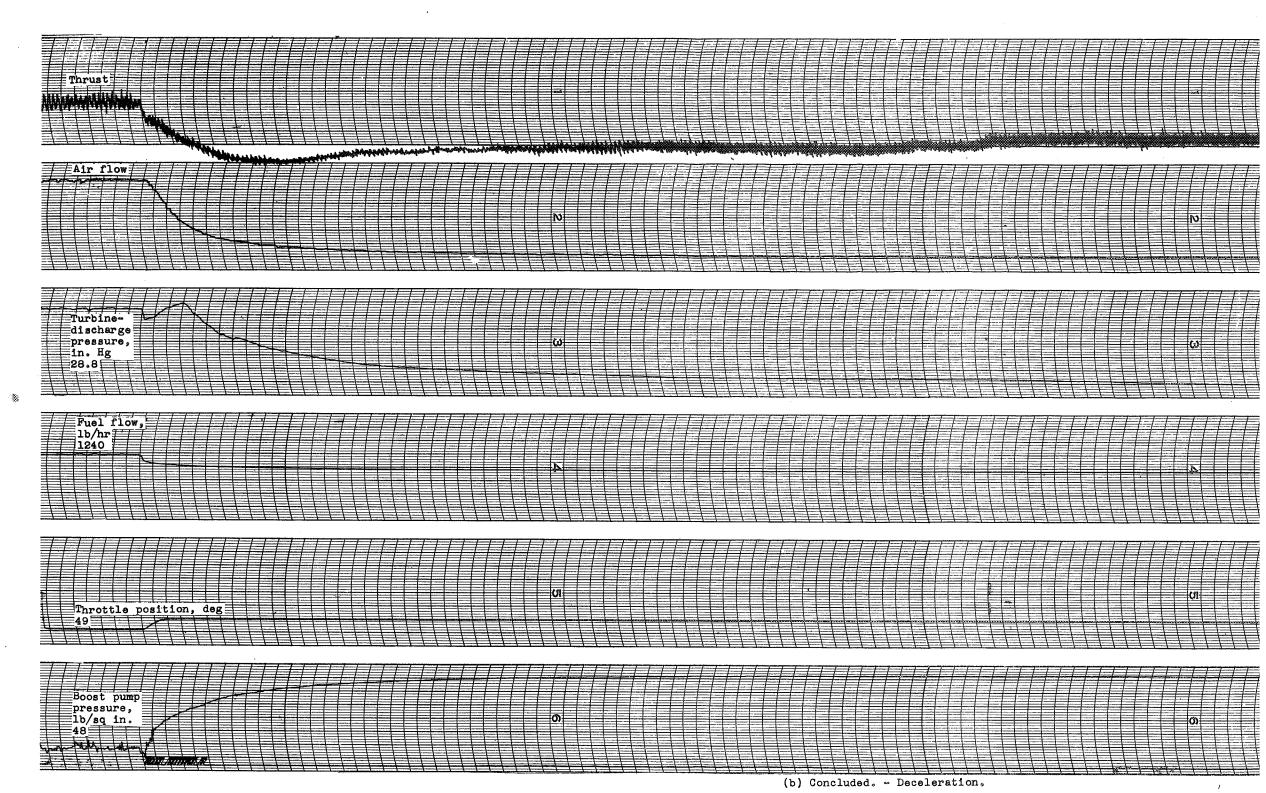
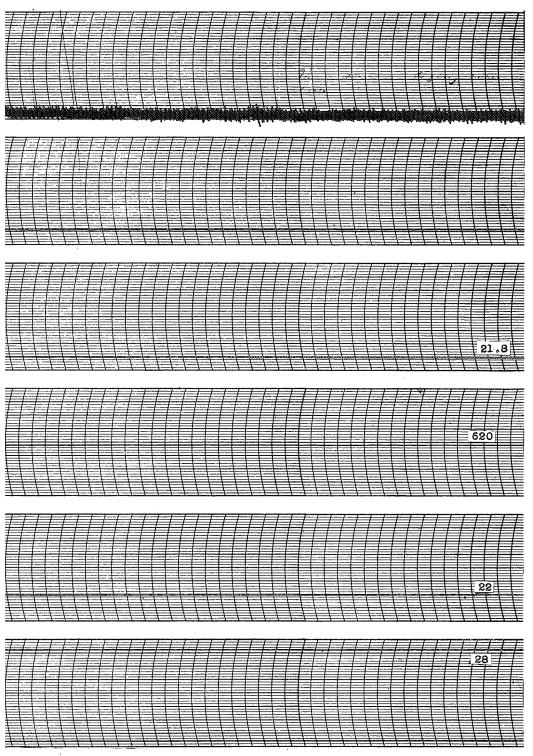


Figure 11. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 22° to 49°; altitude, 10,000 feet





nominal ram-pressure ratio, 1.2.

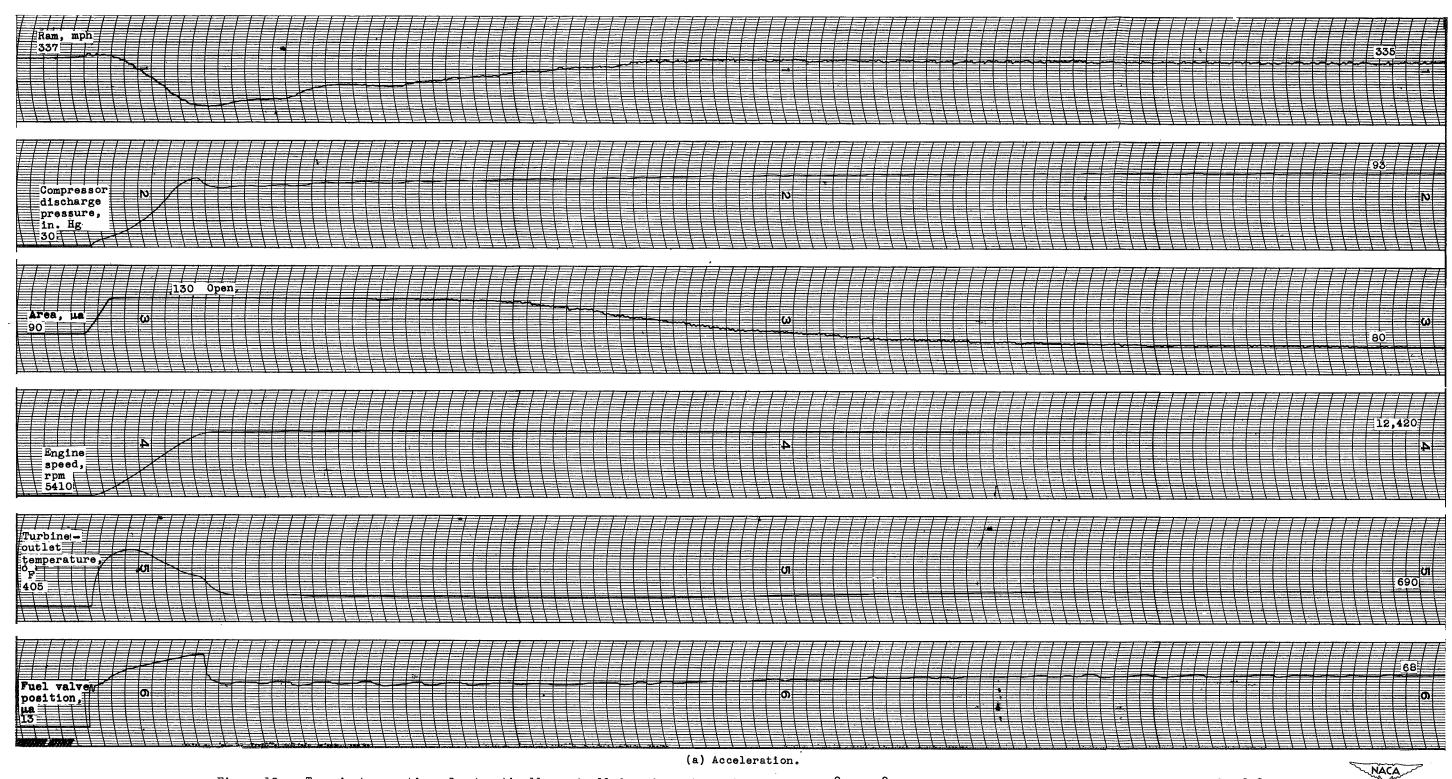


Figure 12. - Transient operation of automatically-controlled engine. Throttle position, 22° to 65°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.

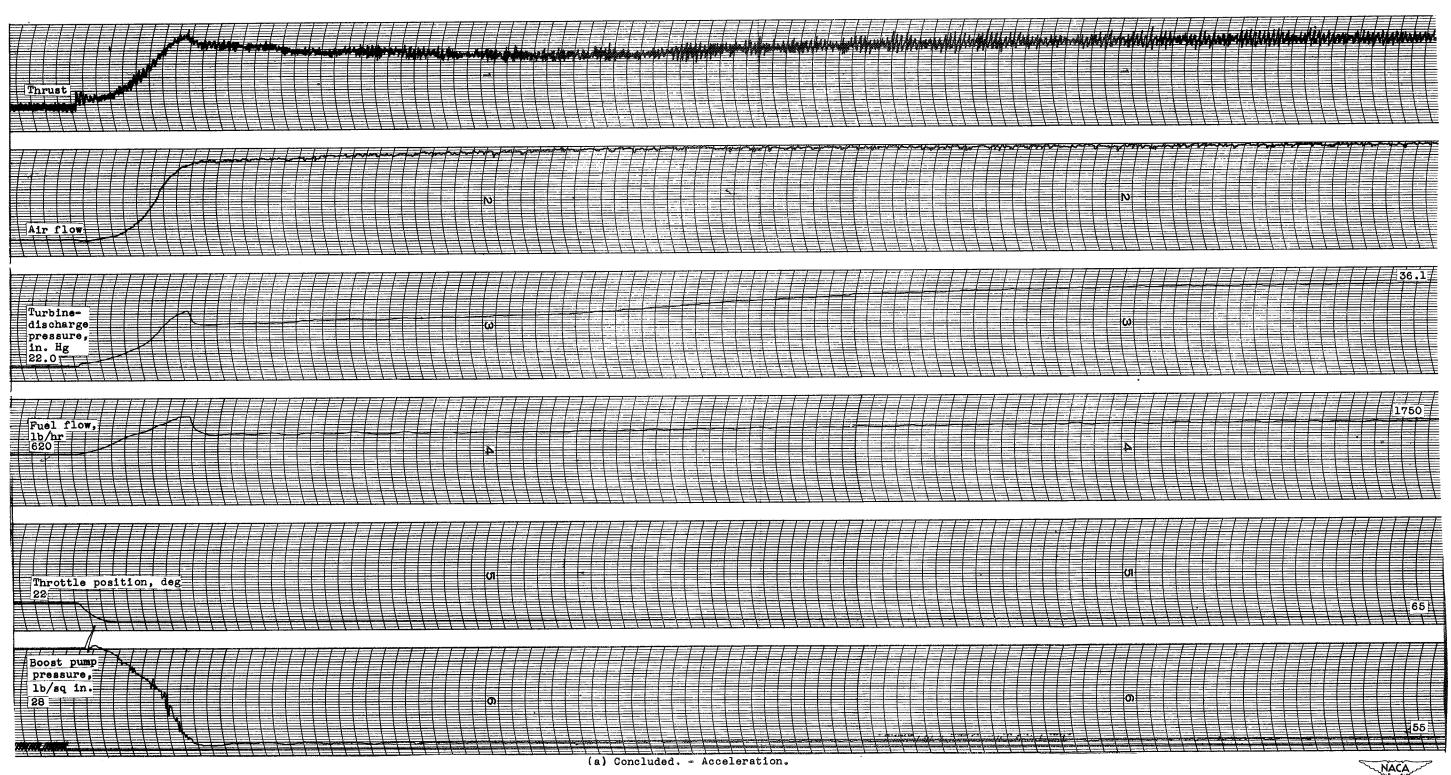


Figure 12. - Continued. Transient operation of automatically-controlled engine. Throttle position, 22° to 65°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.

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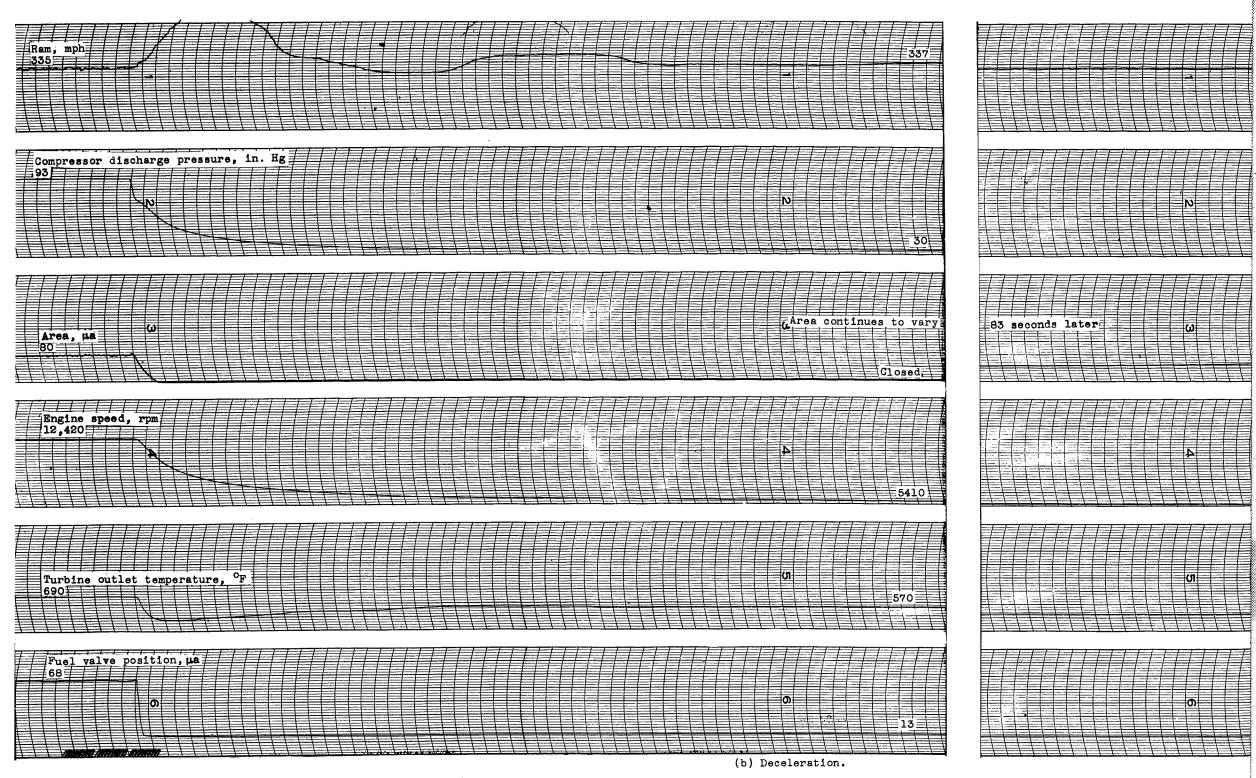
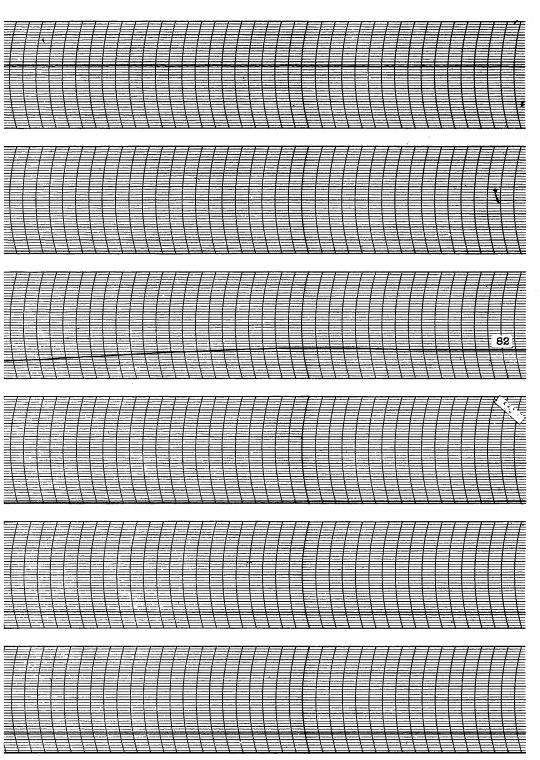


Figure 12. - Continued. Transient operation of automatically-controlled engine. Throttle position, 22° to 65°; altitude, 10,000 feet; nominal ram-preconfidential.



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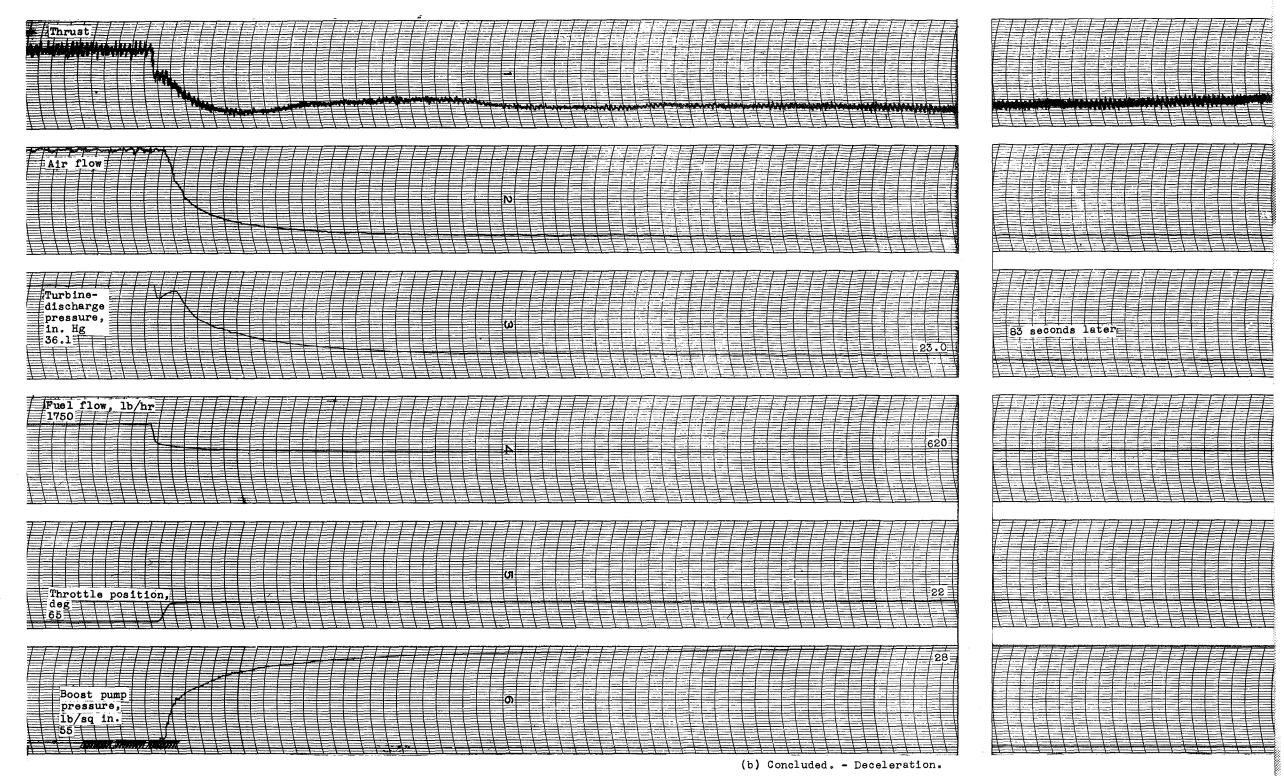
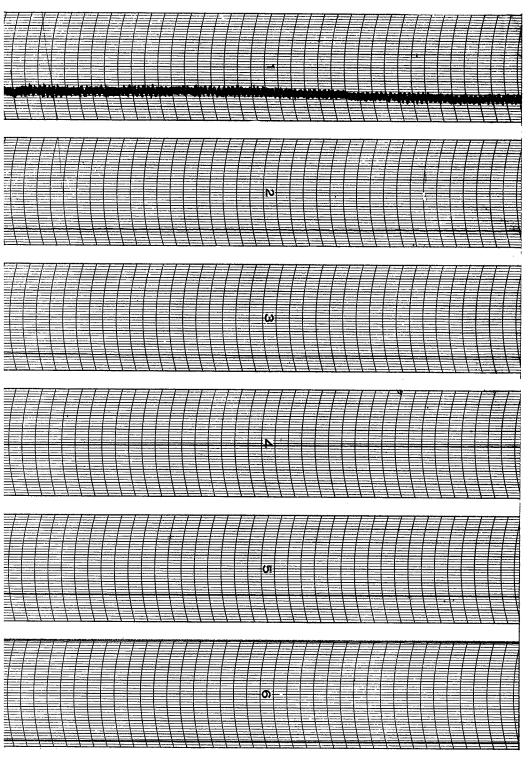
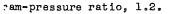


Figure 12. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 22° to 65°; altitude, 10,000 feet; nominal

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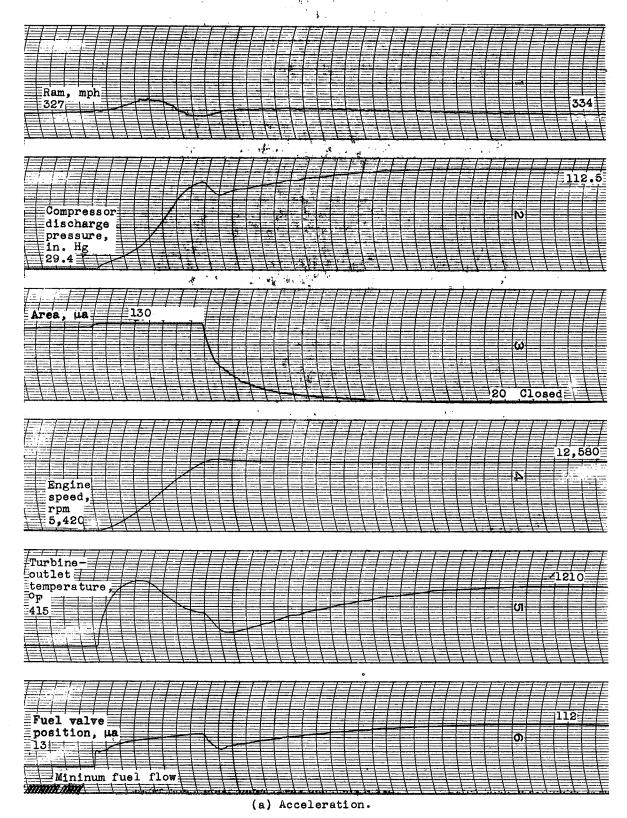
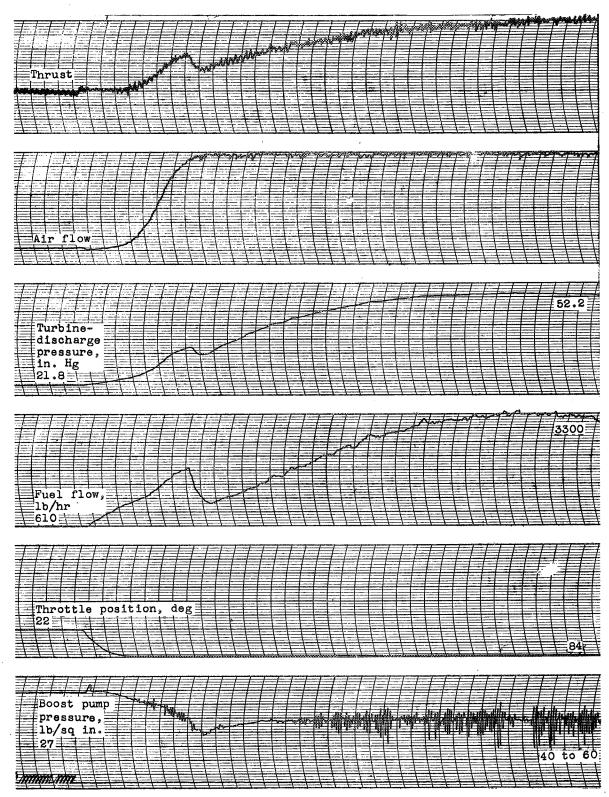


Figure 13. - Transient operation of automatically-controlled engine. Throttle position, 22° to 84°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.





(a) Concluded. - Acceleration.

Figure 13. - Continued. Transient operation of automatically-controlled engine. Throttle position, 22° to 84°; altitude, 10,000 feet; nominal ram-pressure ratio, 1.2.



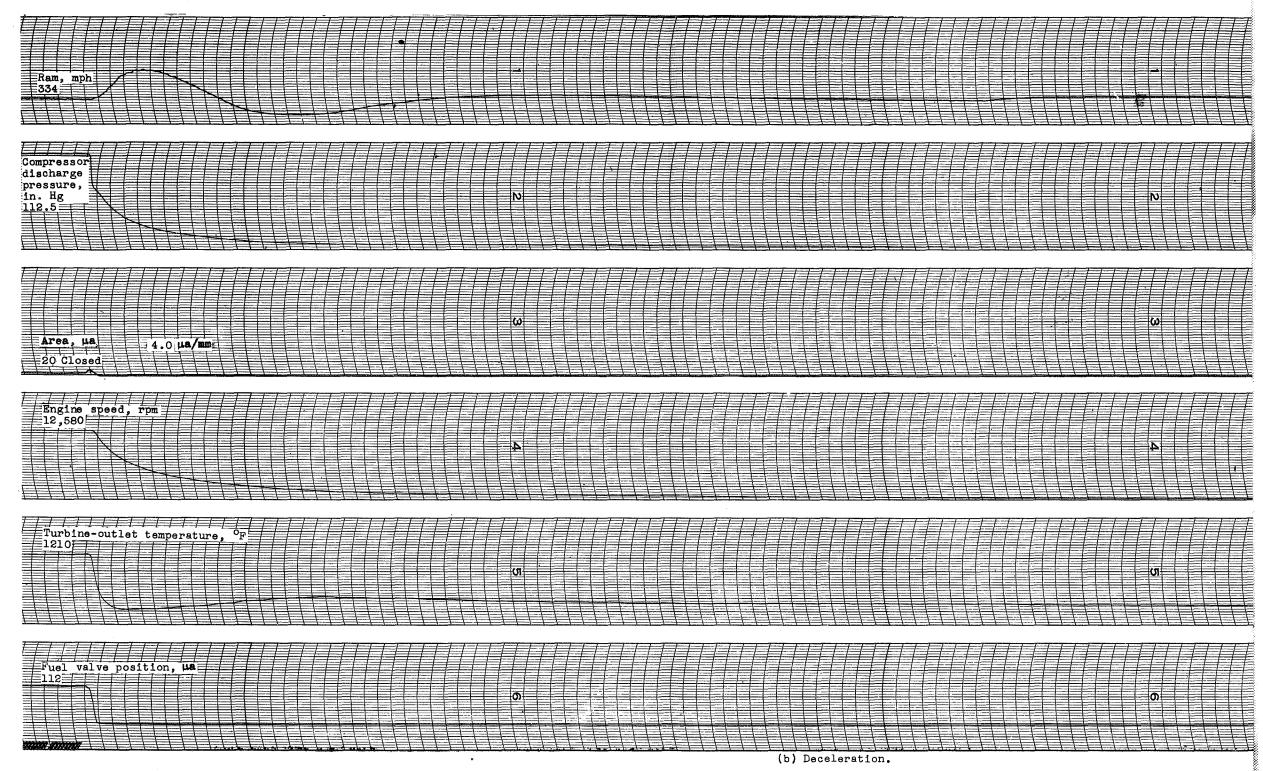
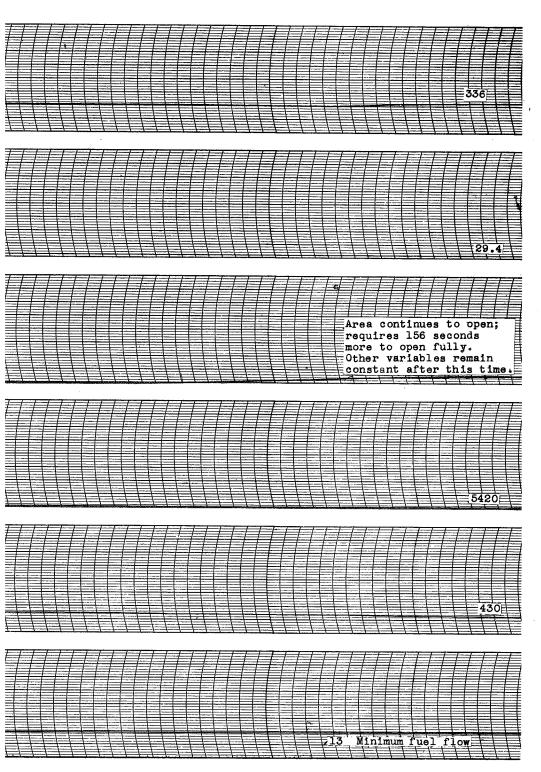


Figure 13. - Continued. Transient operation of automatically-controlled engine. Throttle position, 22° to 84°; altitude, 10,000 feet; nominal r



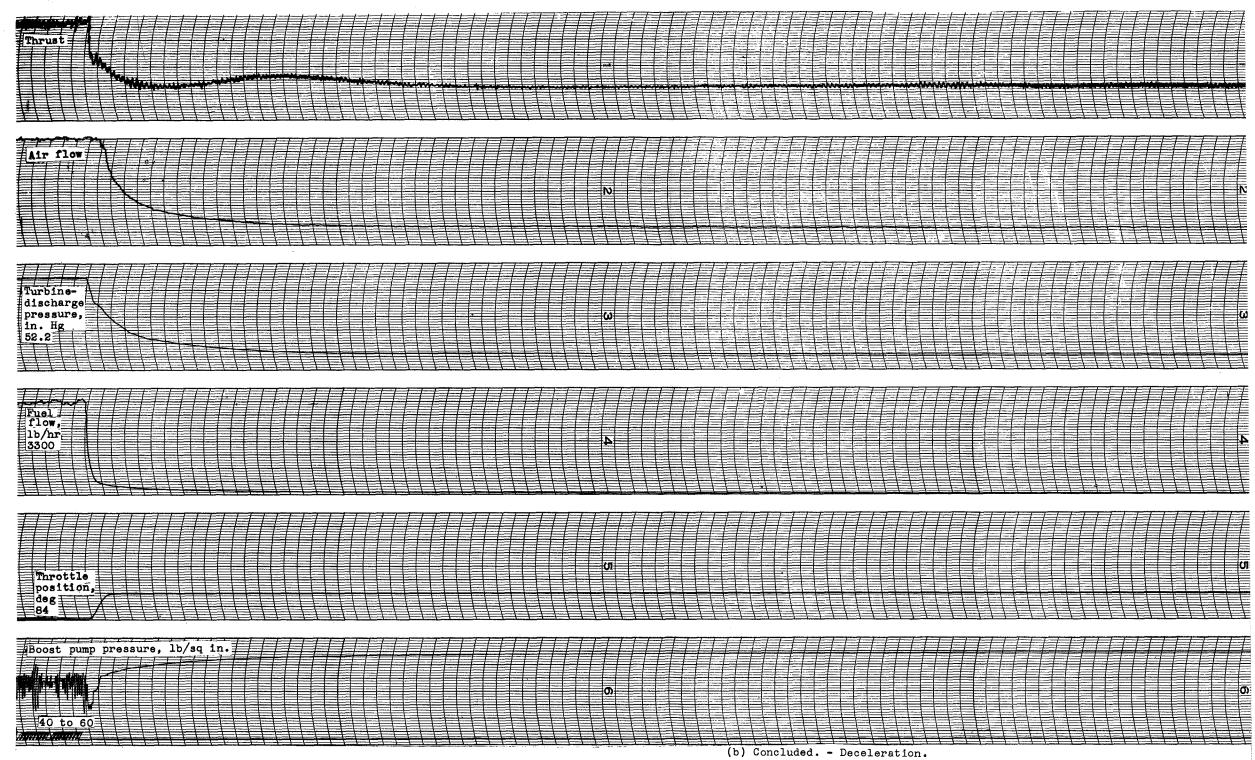
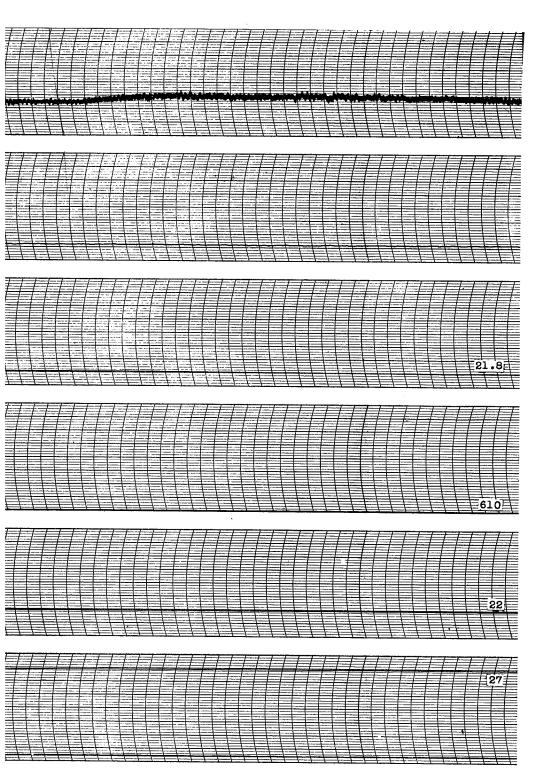


Figure 13. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 22° to 84°; altitude, 10,000 feet; nomina

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ram-pressure ratio, 1.2.

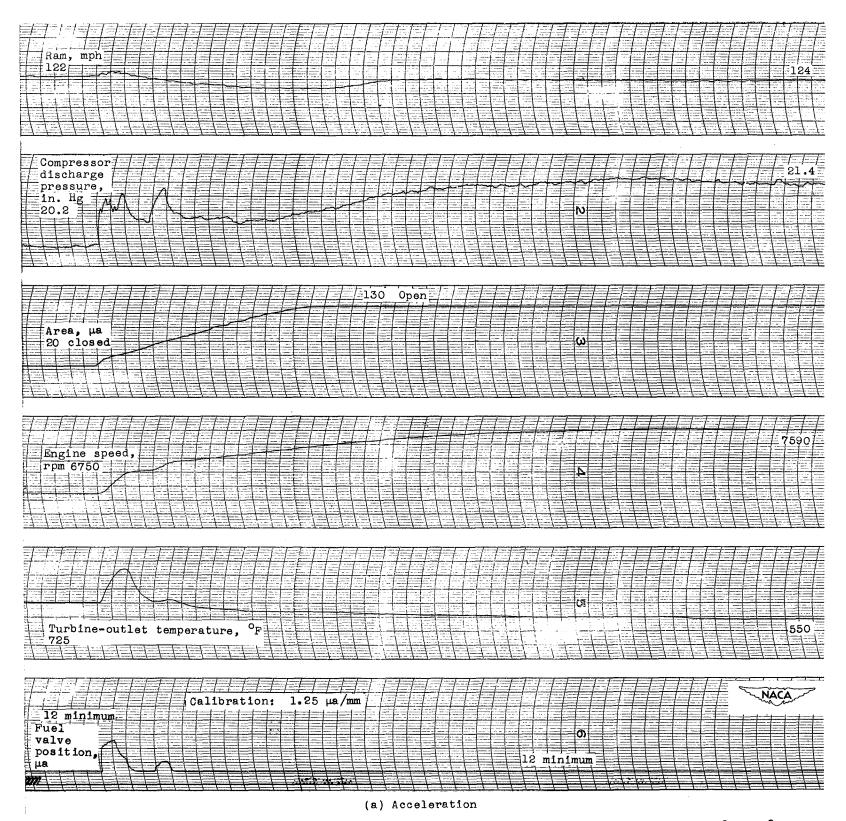


Figure 14. - Transient operation of automatically-controlled engine. Throttle position, 22° to 35°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.05.

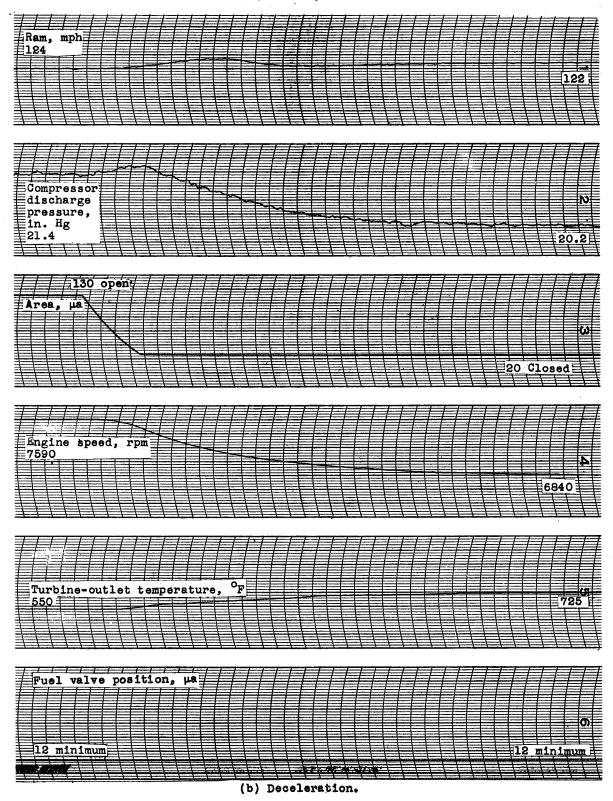


Figure 14. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 22° to 35°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.05.



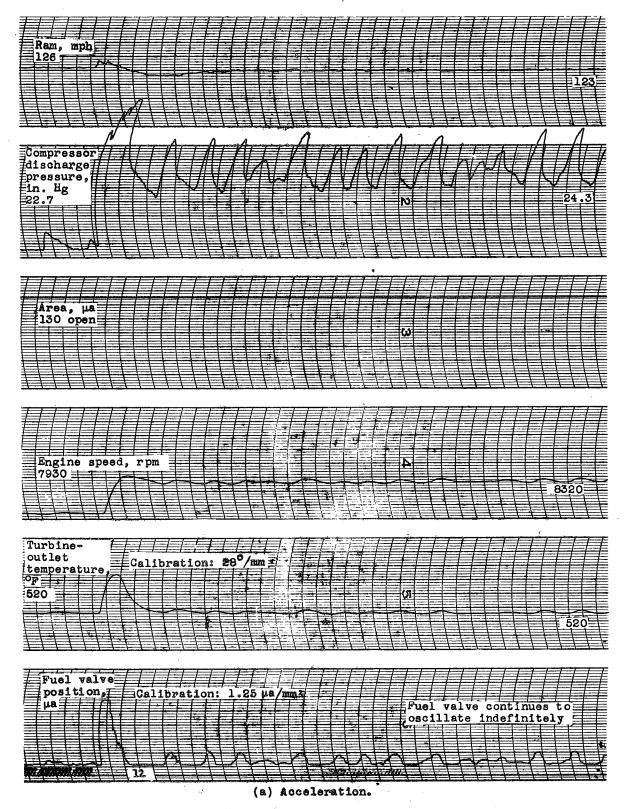


Figure 15. - Transient operation of automatically-controlled engine. Throttle position, 36.5° to 37.5°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.05.

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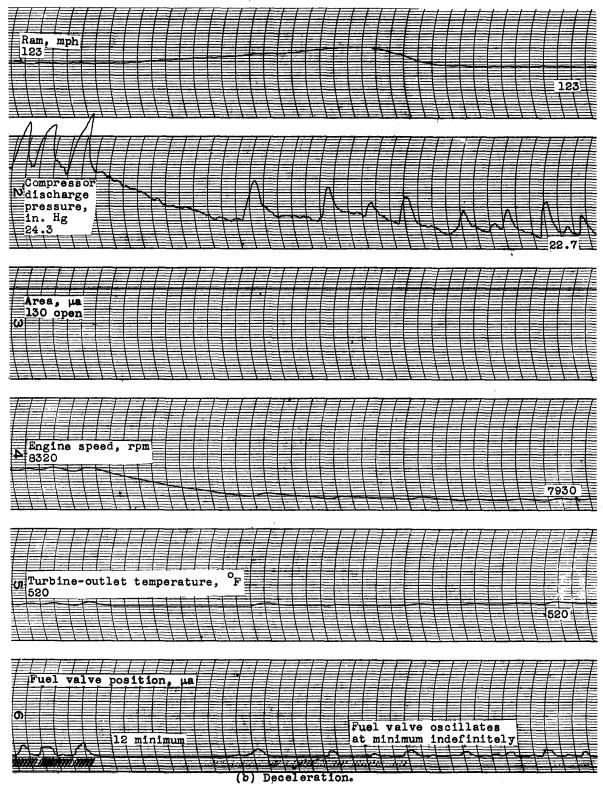


Figure 15. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 36.5° to 37.5°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.05.

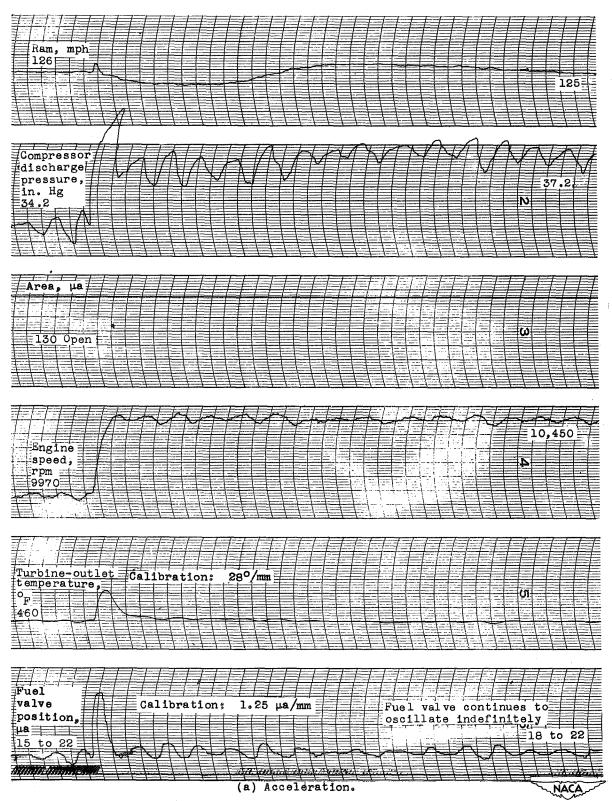


Figure 16. - Transient operation of automatically-controlled engine. Throttle position, 41.5° to 43°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.05.

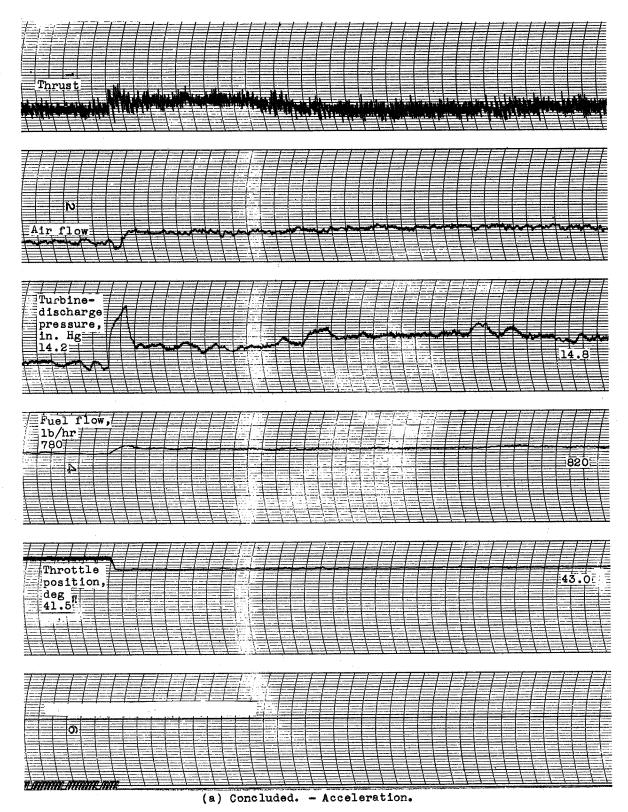


Figure 16. - Continued. Transient operation of automatically-controlled engine. Throttle position, 41.5° to 43°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.05.

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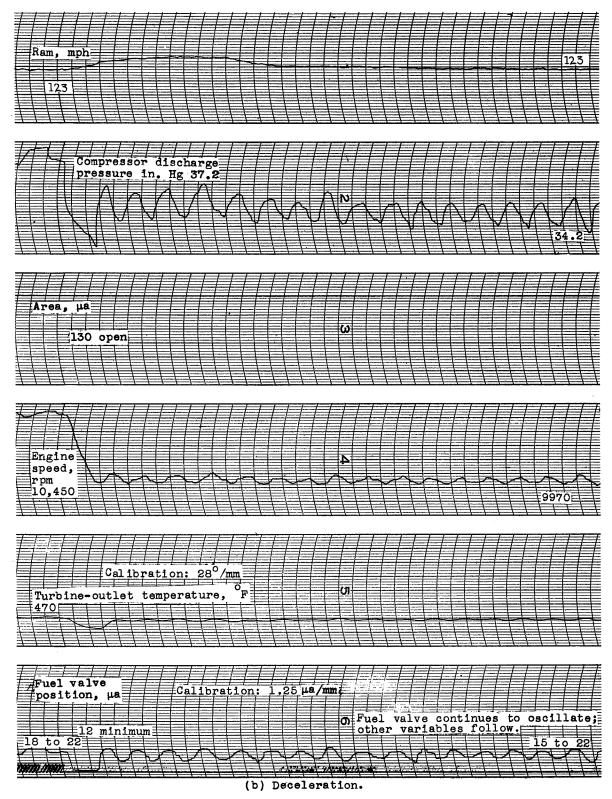
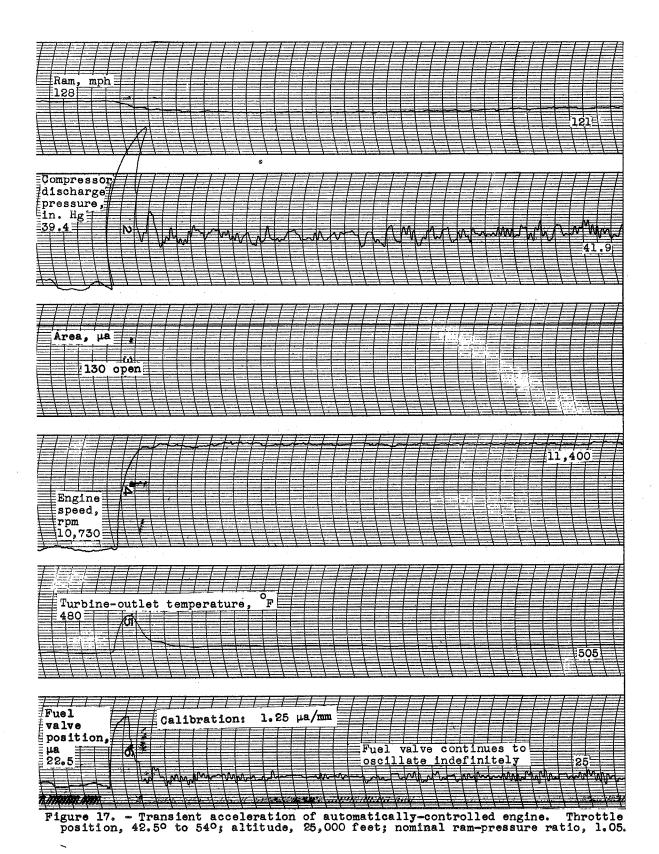


Figure 16. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 41.5° to 43°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.05.







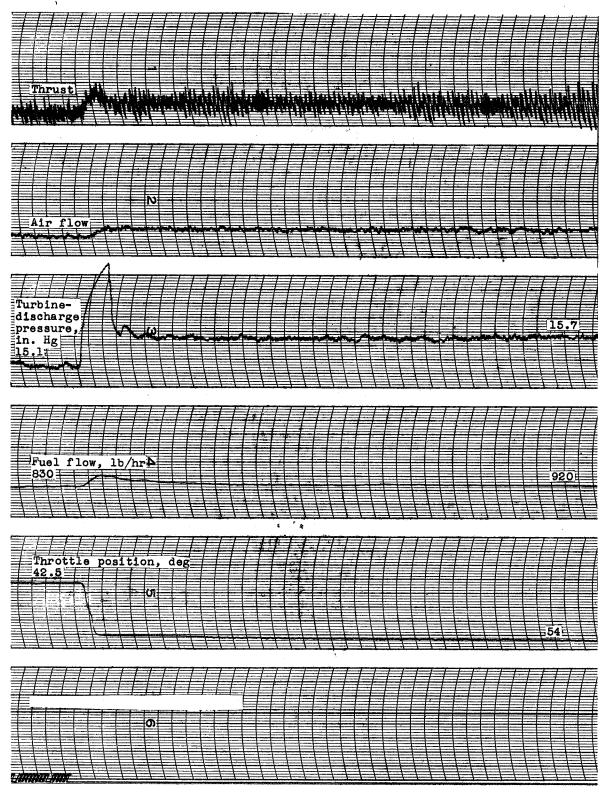


Figure 17. - Concluded. Transient acceleration of automatically-controlled engine. Throttle position, 42.5° to 54°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.05.





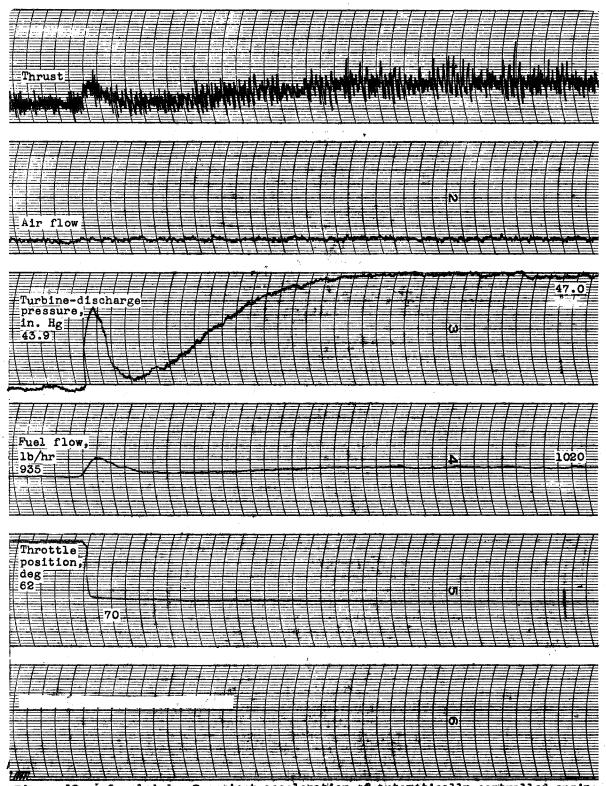


Figure 18. - Concluded. Transient acceleration of automatically-controlled engine. Throttle position, 62° to 70°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.05.



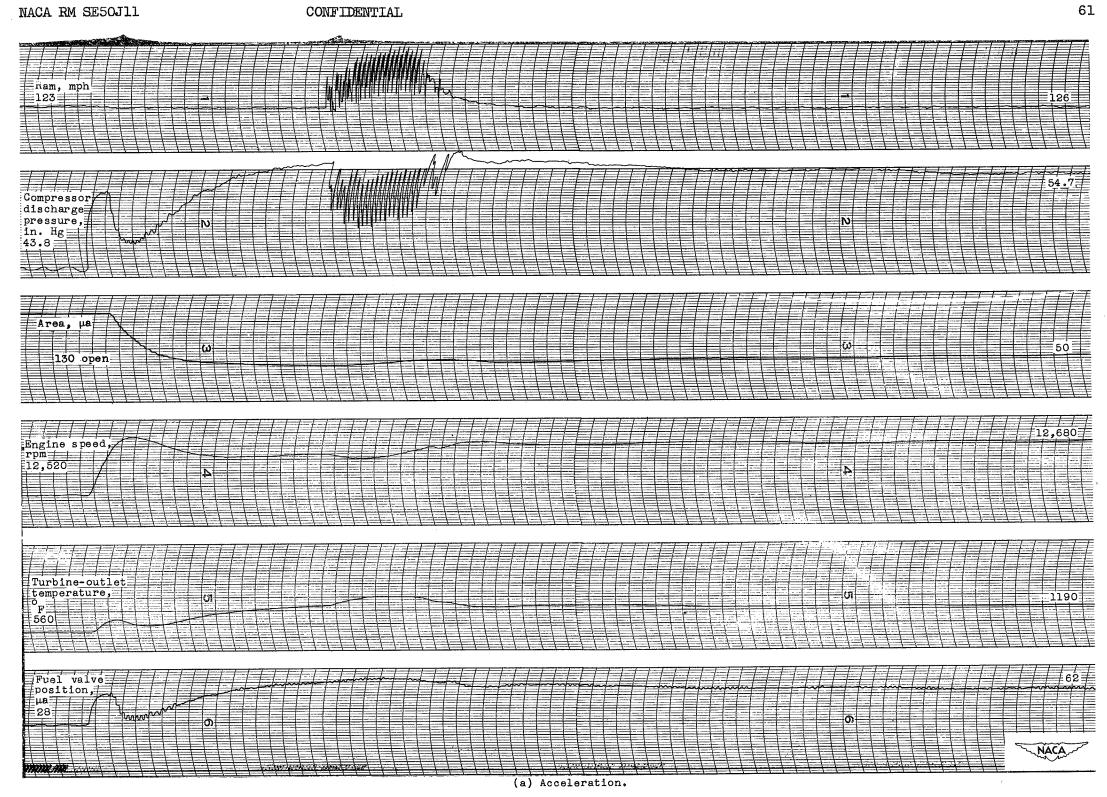


Figure 19. - Transient operation of automatically-controlled engine. Throttle position, 62° to 83°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.05.

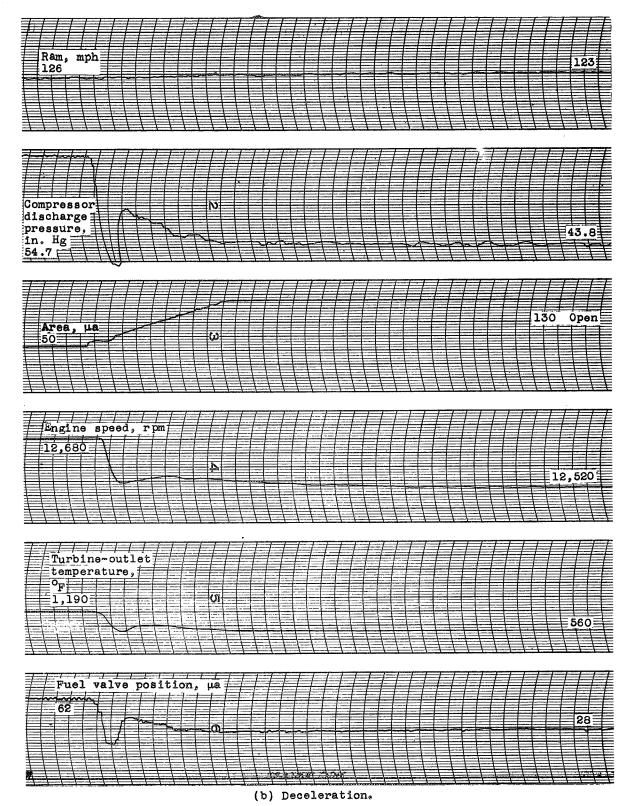
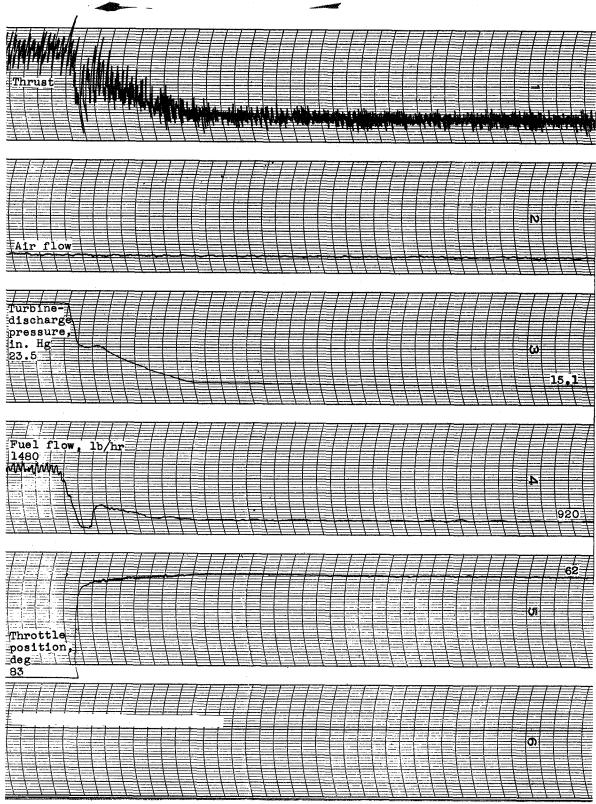


Figure 19. - Continued. Transient operation of automatically-controlled engine. Throttle position, 62° to 83°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.05.





(b) Concluded. - Deceleration

Figure 19. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 62° to 83°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.05.



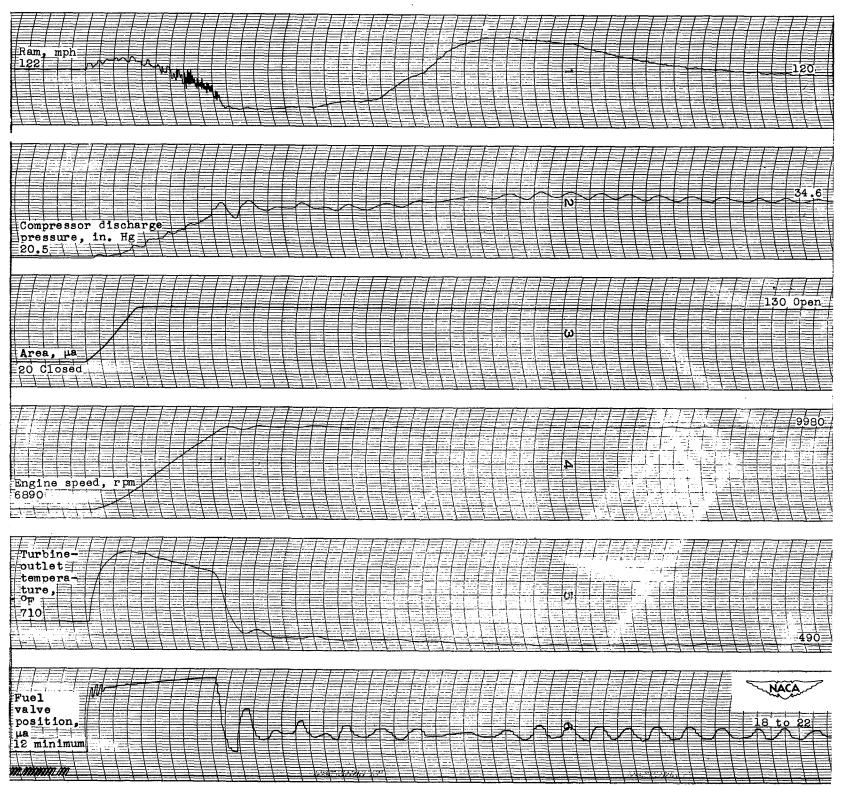


Figure 20. - Transient acceleration of automatically-controlled engine. Throttle position, 22° to 42°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.05.

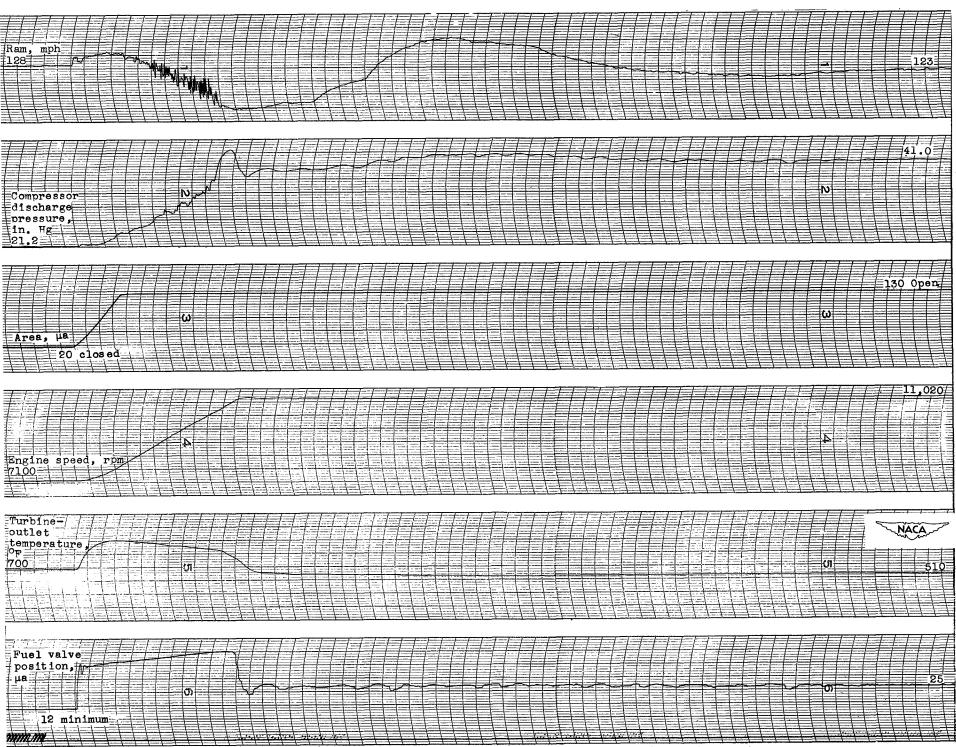
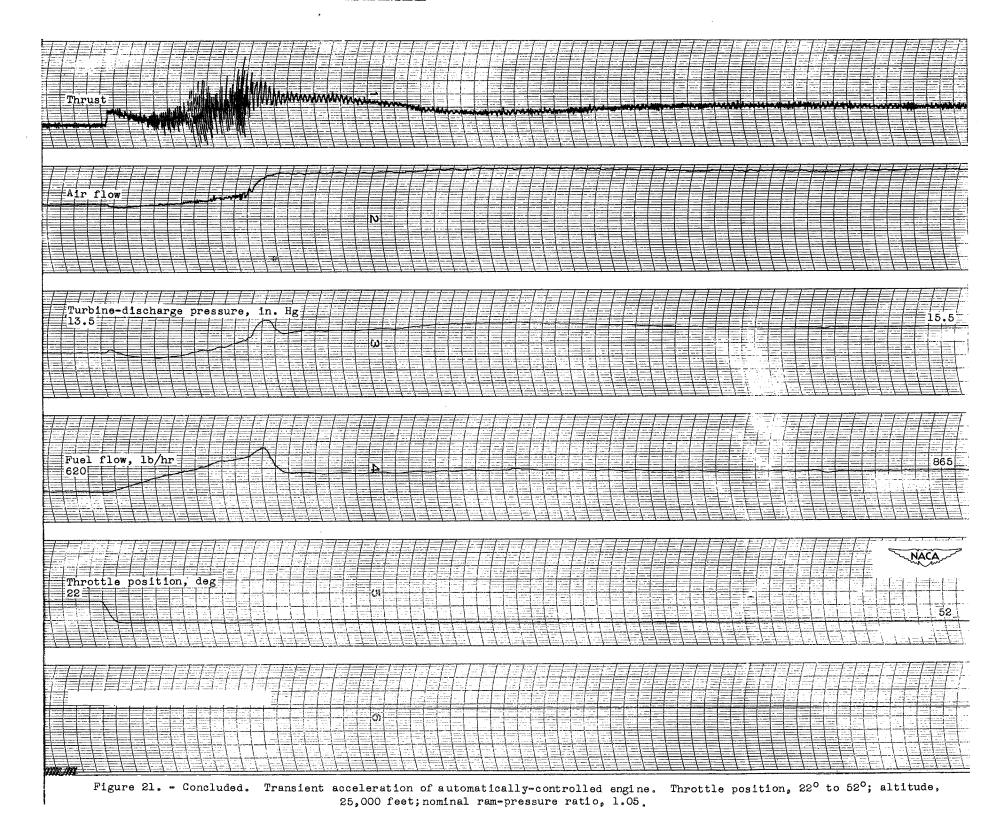


Figure 21. - Transient acceleration of automatically-controlled engine. Throttle position, 22° to 52°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.05.



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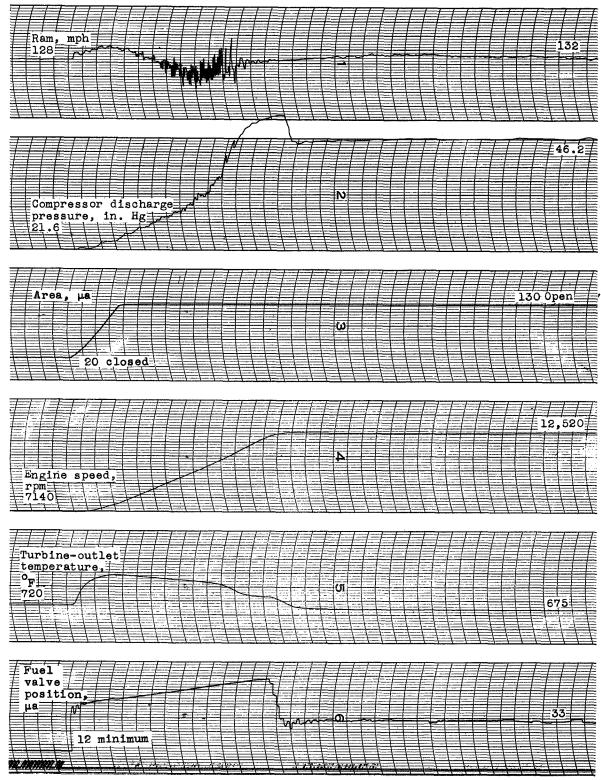


Figure 22. Transient acceleration of automatically-controlled engine. Throttle position. 22.50 to 700; altitude, 25,000 feet; nominal ram-pressure ratio, 1.05.



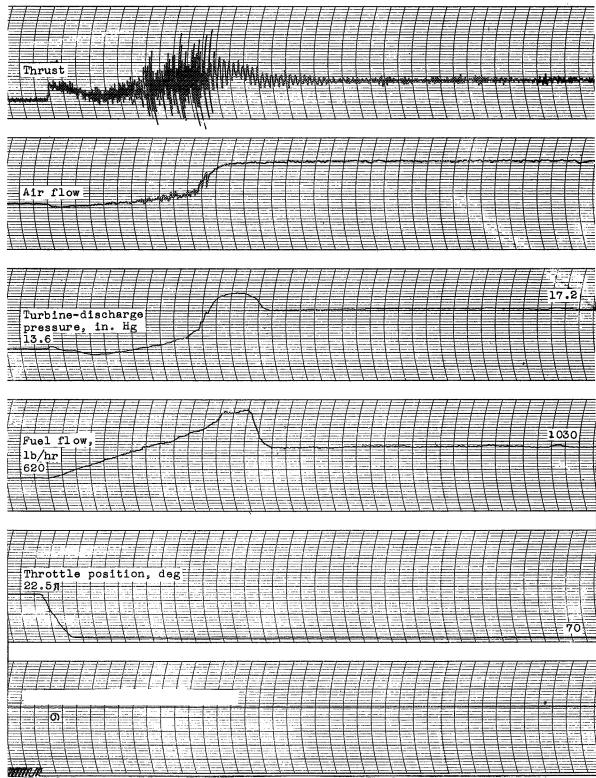


Figure 22. - Concluded. Transient acceleration of automatically-controlled engine. Throttle position, 22.50 to 700; altitude, 25,000 feet; nominal ram-pressure ratio, 1.05.



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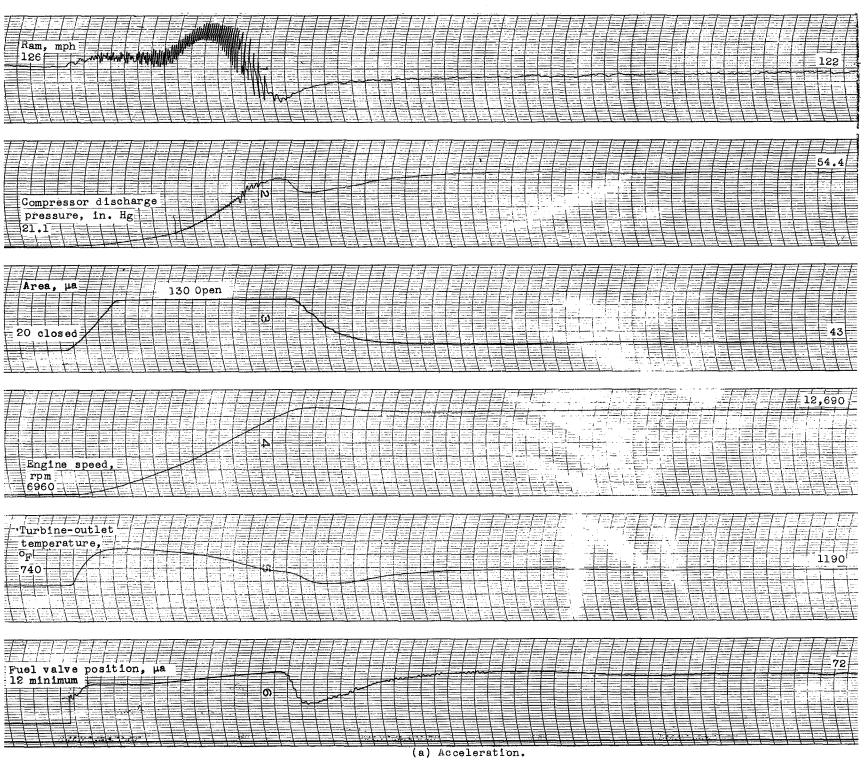


Figure 23. - Transient operation of automatically-controlled engine. Throttle position, 22° to 85°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.05.

Turbine-discharge Fuel flow, lb/hr 85 (a) Concluded. - Acceleration.

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Figure 23. - Continued. Transient operation of automatically-controlled engine. Throttle position, 22° to 85°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.05.

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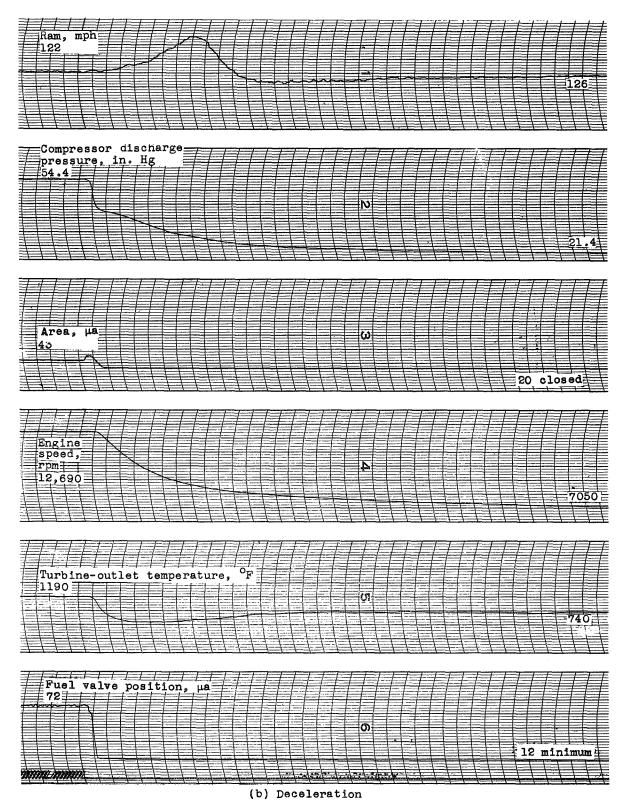


Figure 23. - Continued. Transient operation of automatically-controlled engine. Throttle position, 22° to 85°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.05



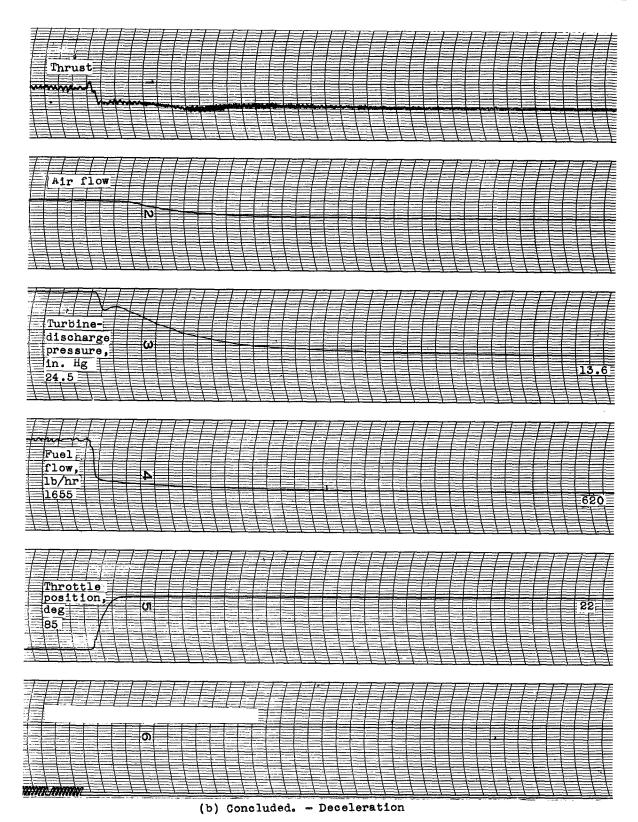
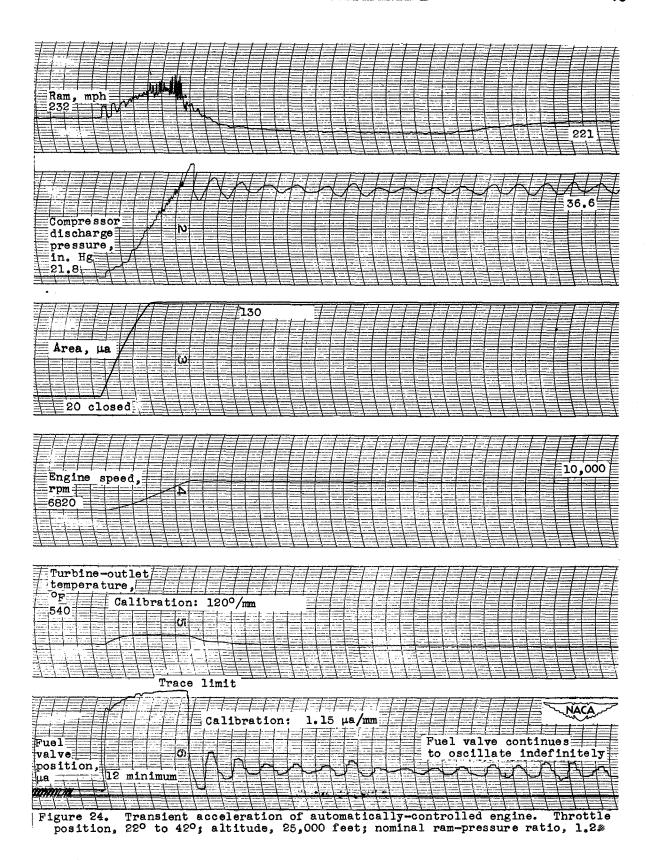


Figure 23. - Concluded. Transient operation of automatically-controlled engine.
Throttle position 22 to 85; altitude, 25,000 feet; nominal ram-pressure ratio, 1.05.





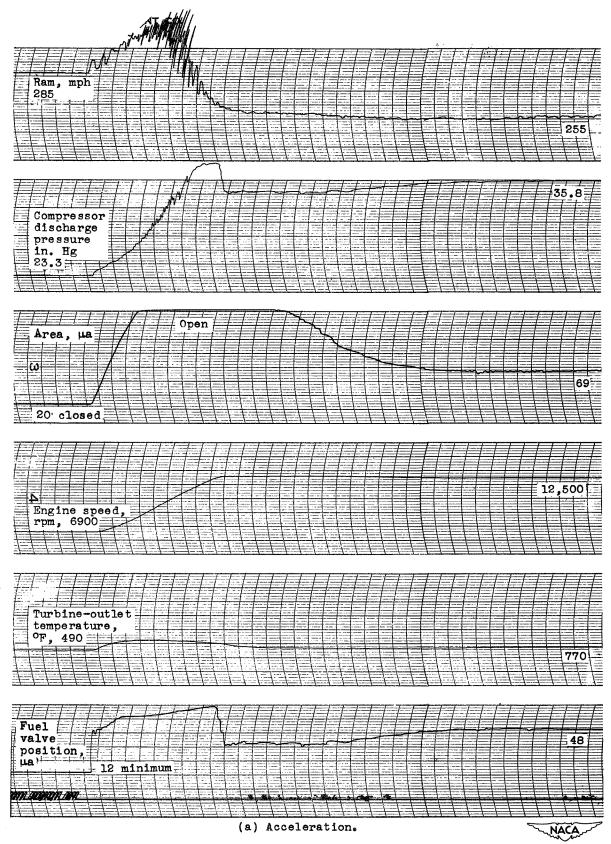


Figure 25. - Transient operation of automatically-controlled engine. Throttle position, 260 to 720; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2

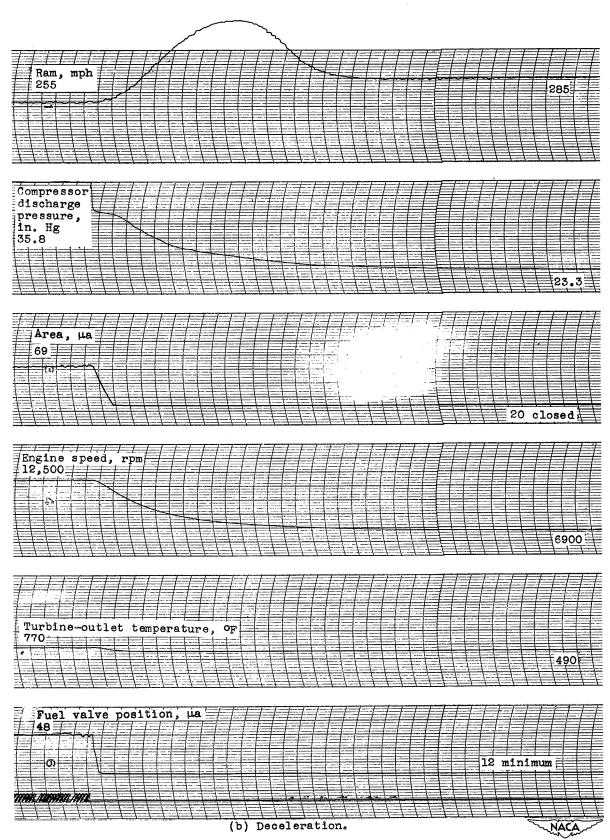


Figure 25. - Concluded. Transient operation of automatically-controlled engine.
Throttle position, 72° to 26°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.

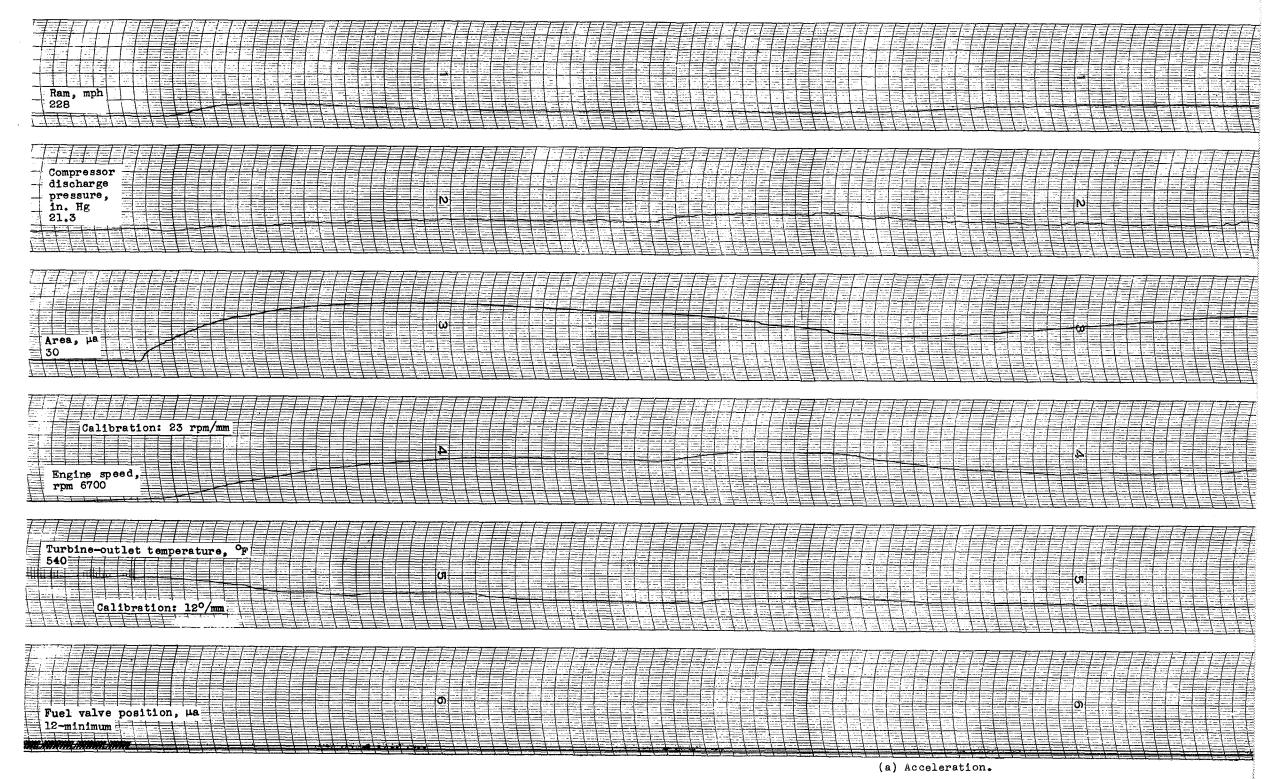
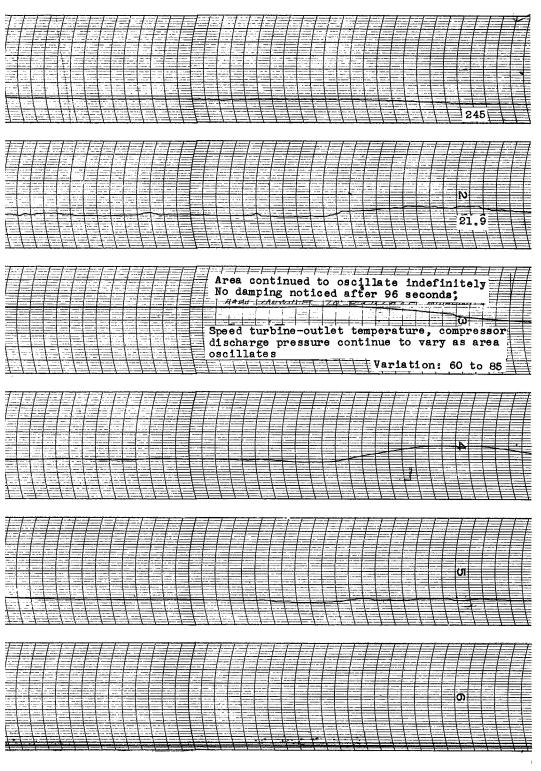


Figure 26. Transient operation of automatically-controlled engine. Throttle position, 31.50 to 33.50; altitude, 25,000 feet; no

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inal ram-pressure ratio, 1.2.

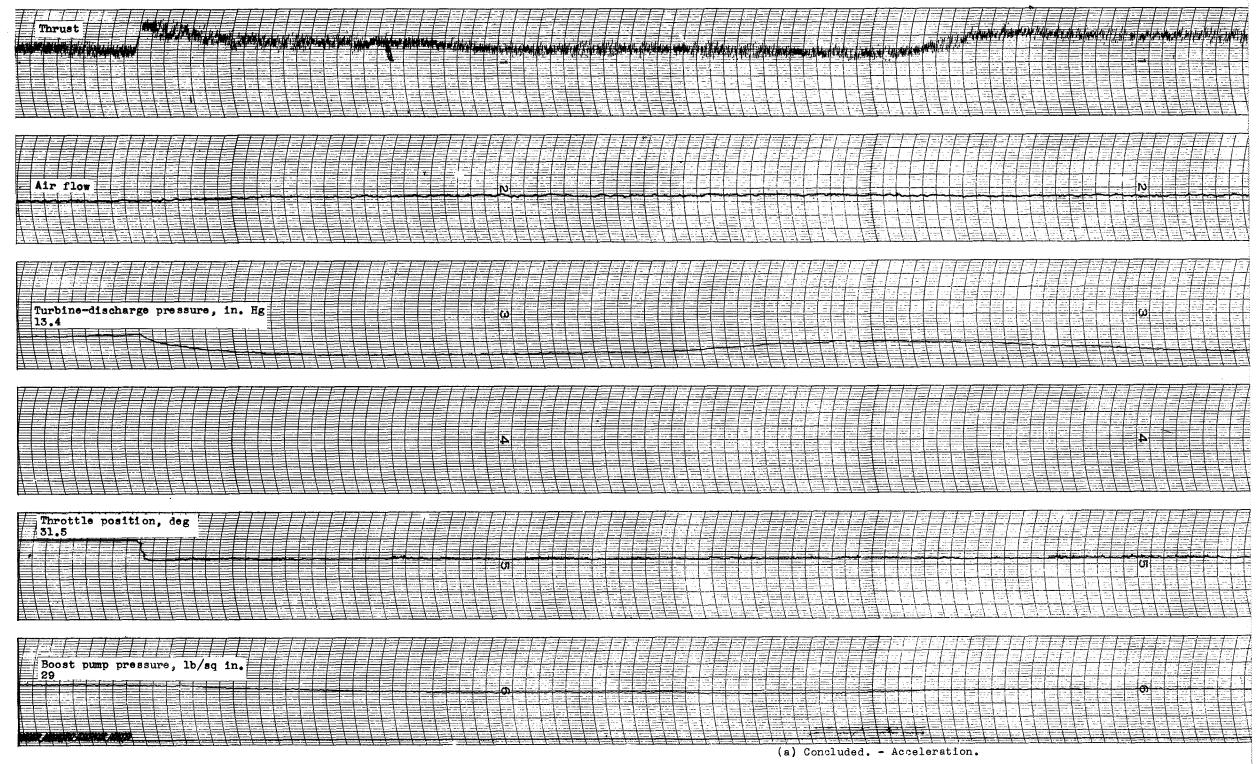
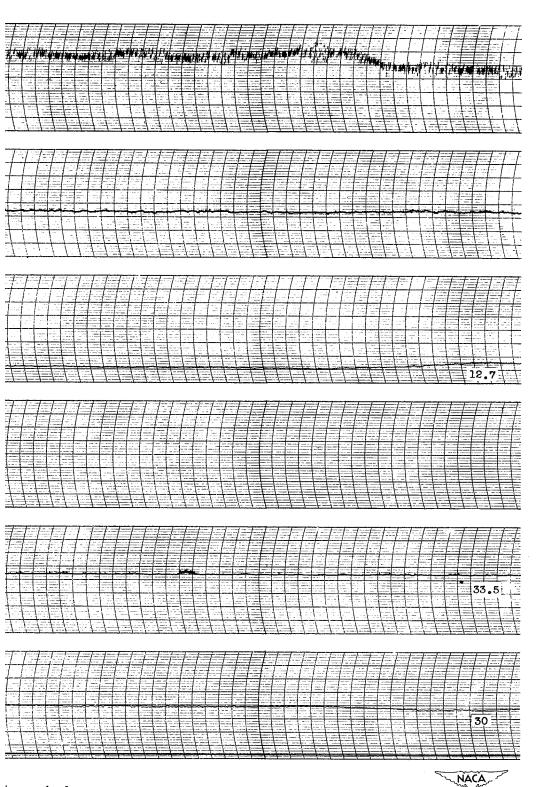


Figure 26. - Continued. - Transient operation of automatically-controlled engine. Throttle position, 31.5° to 33.5°; altitude, 25,000 fe





t; nominal ram-pressure ratio, 1.2.

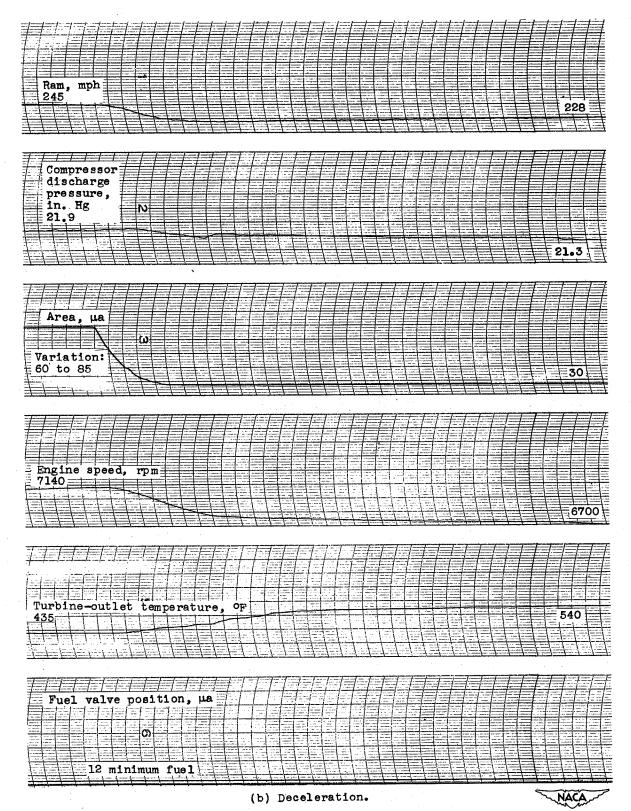


Figure 26. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 31.5° to 33.5°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.

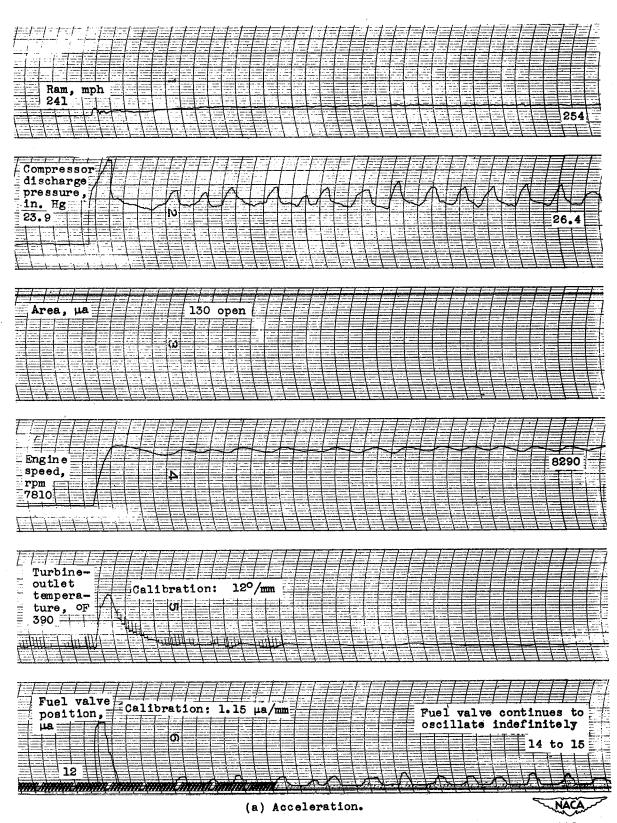


Figure 27. - Transient operation of automatically-controlled engine. Throttle position, 34° to 37°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.

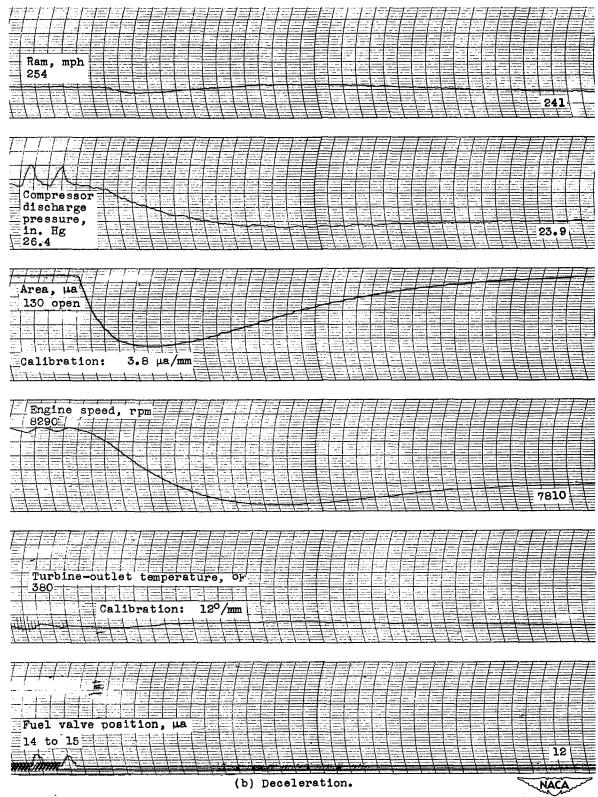
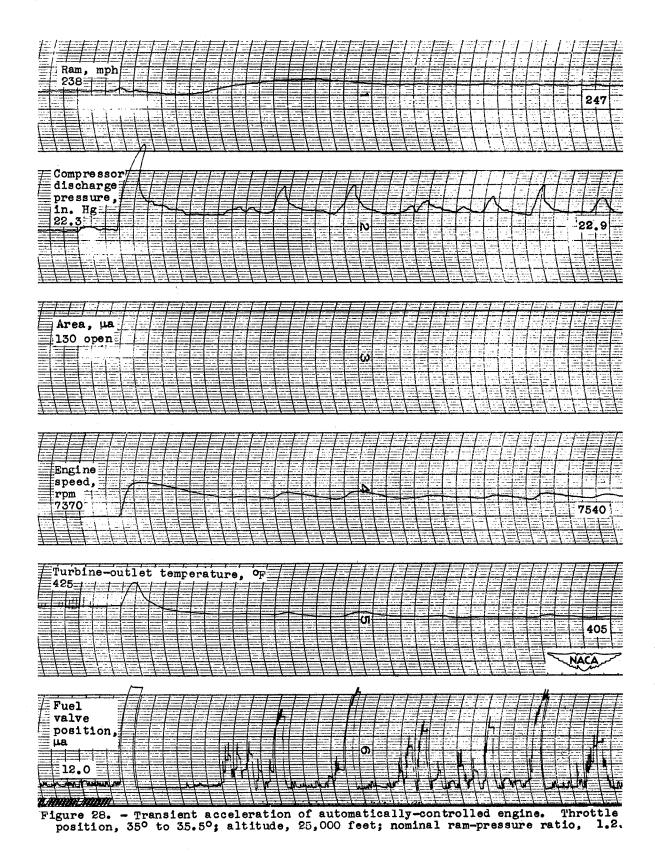


Figure 27. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 340 to 370; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.



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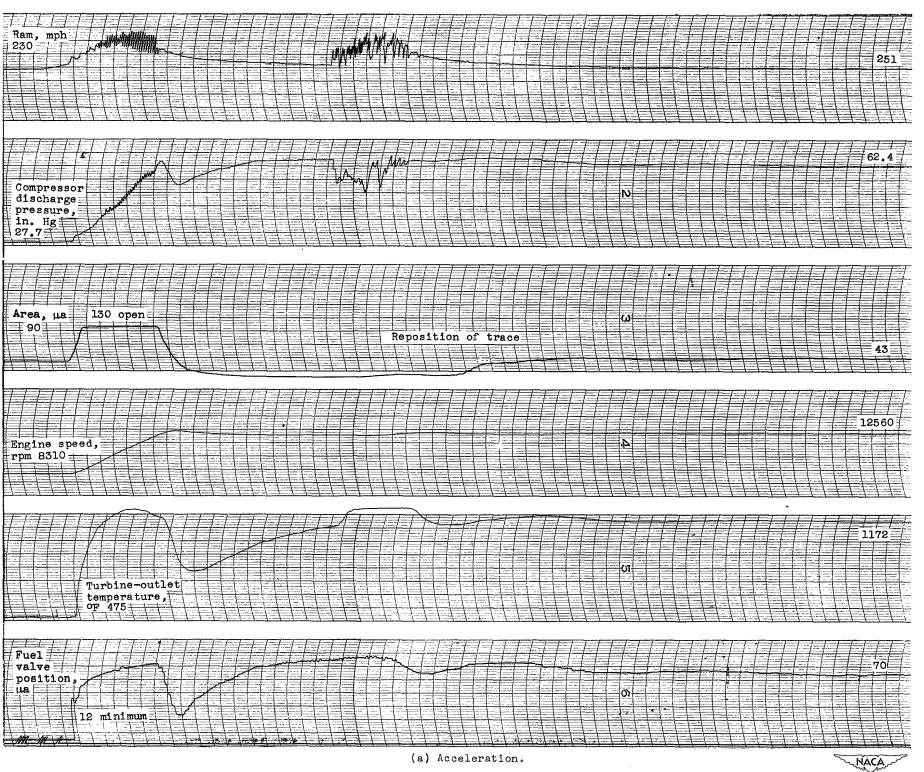


Figure 29. - Transient operation of automatically-controlled engine. Throttle position, 35° to 85°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.

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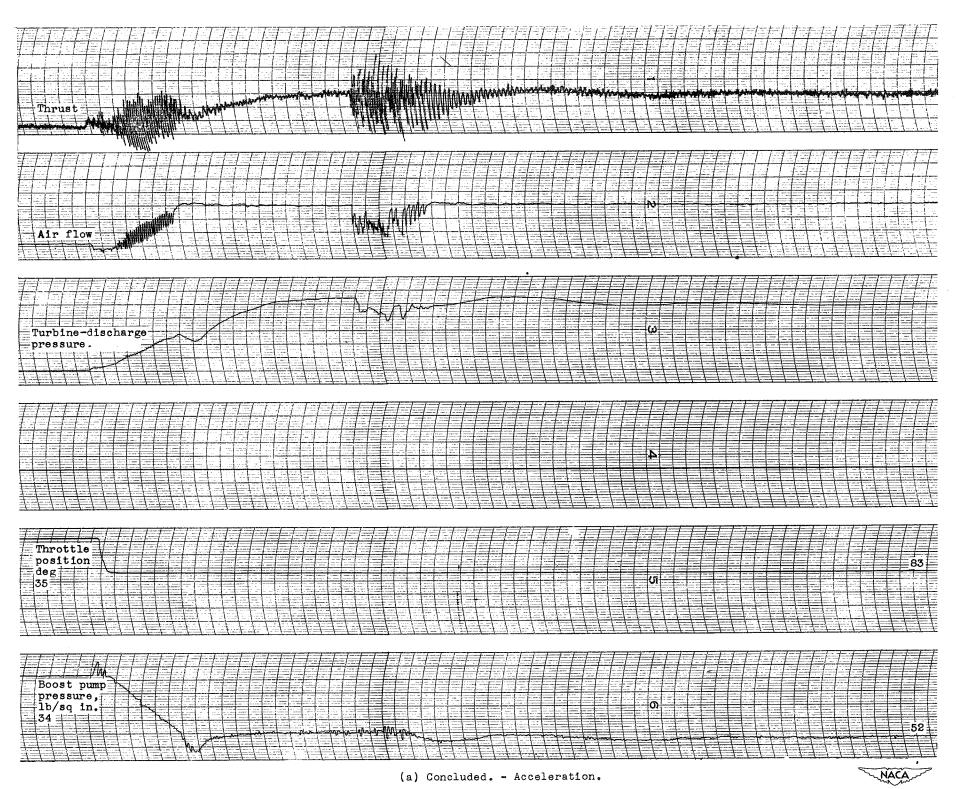


Figure 29. - Continued. Transient operation of automatically-controlled engine. Throttle position, 35° to 85°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.

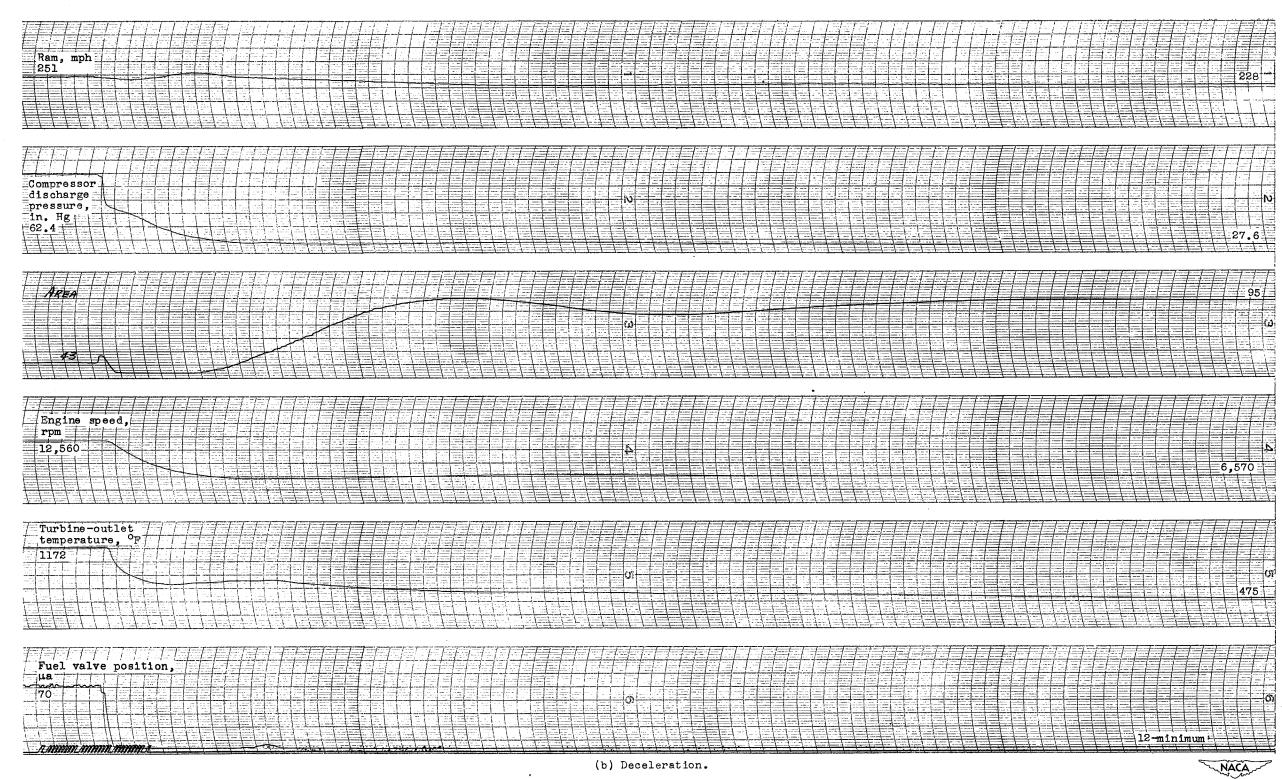


Figure 29. - Continued. Transient operation of automatically-controlled engine. Throttle position, 35° to 85°; altitude, 25,000 feet; nominal ram-pressure ratio 1.2.

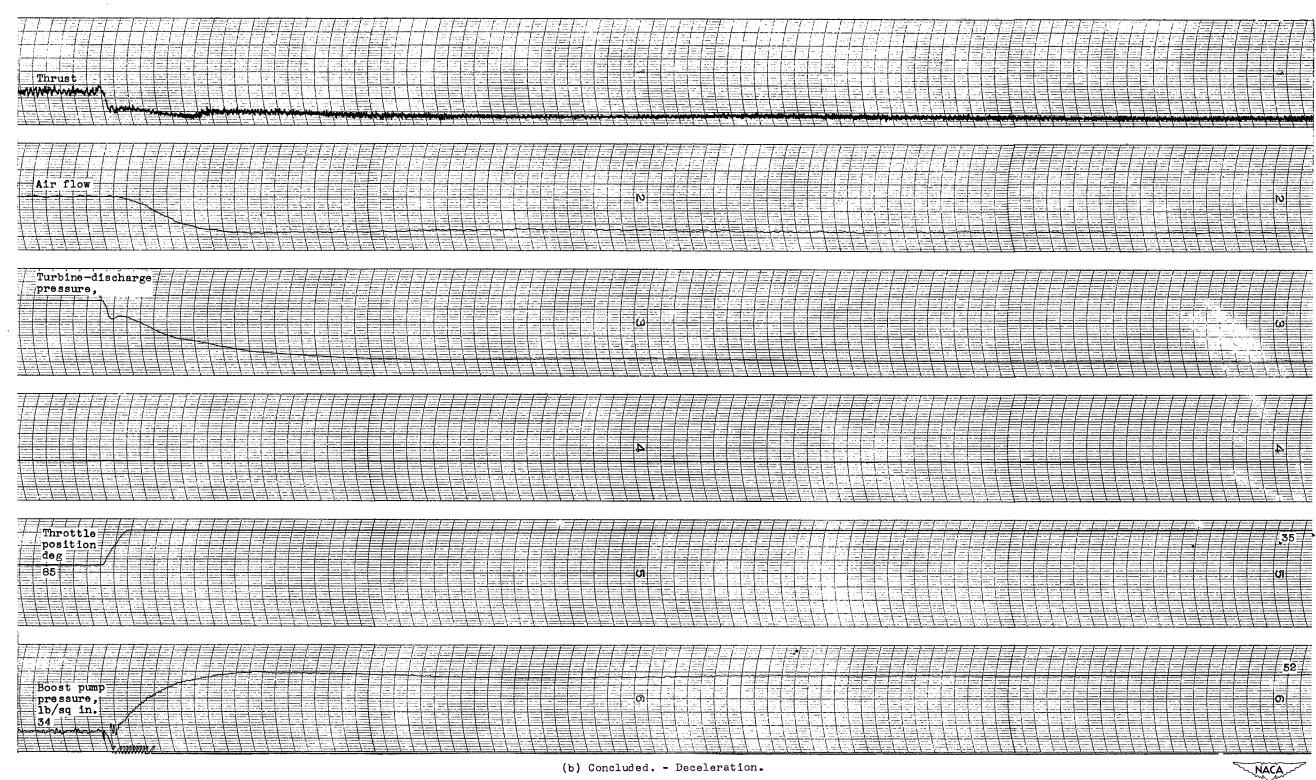


Figure 29. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 35° to 85°; altitude, 25,000 feet; nominal ram-pressure ratio 1.2.

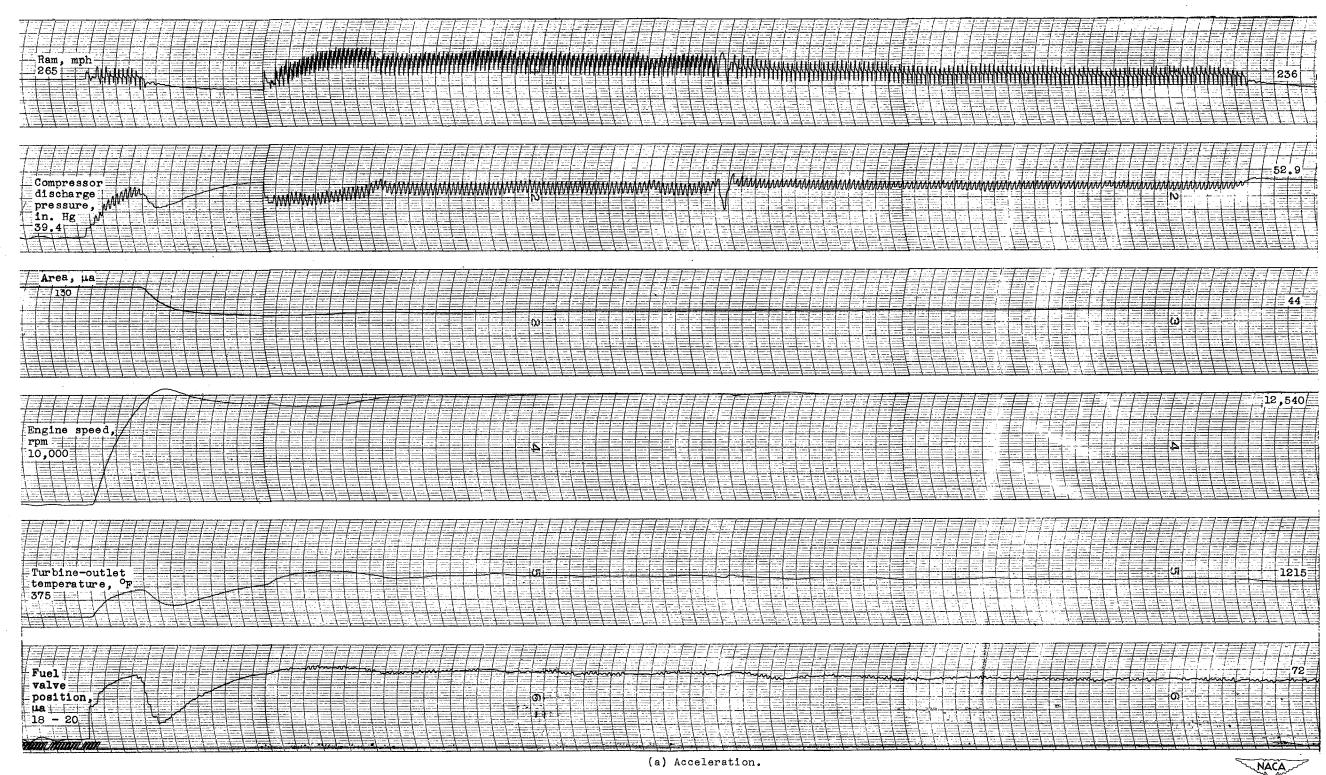


Figure 30. - Transient operation of automatically-controlled engine. Throttle position, 41° to 85°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.

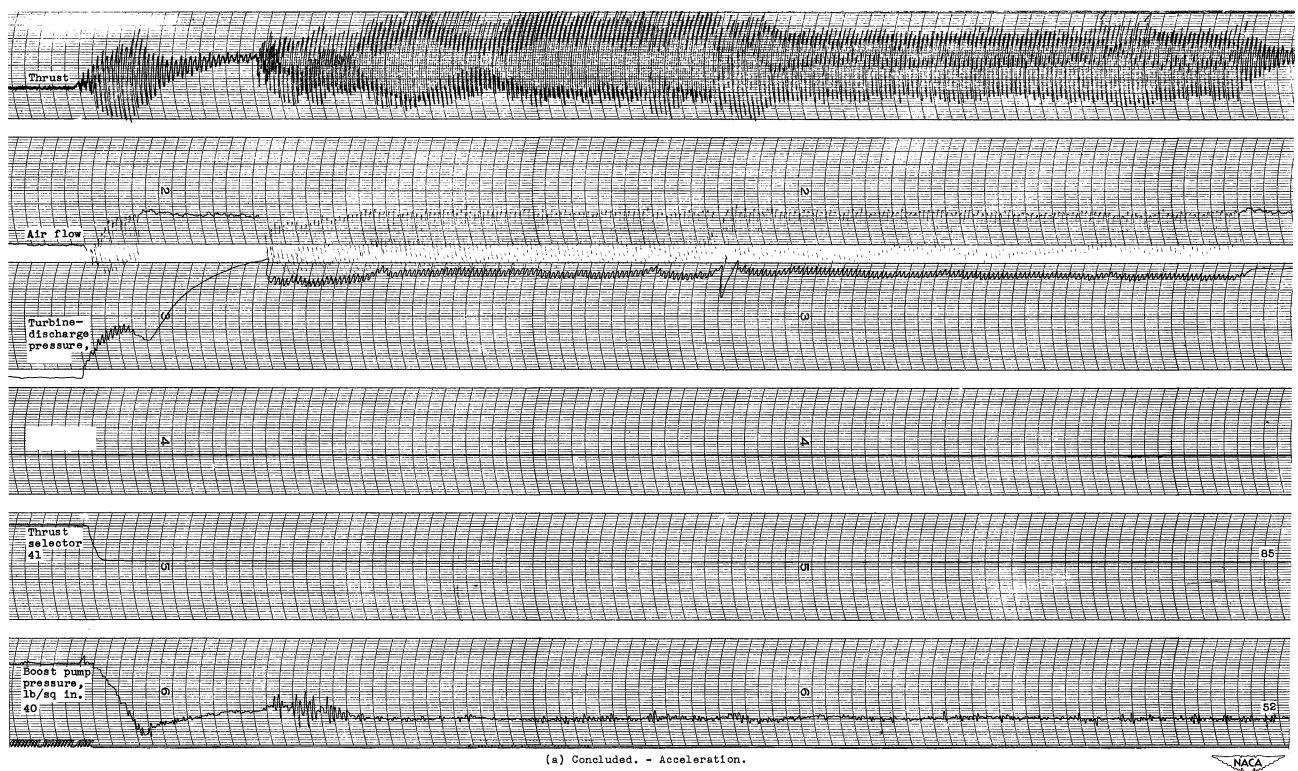


Figure 30. - Continued. Transient operation of automatically-controlled engine. Throttle position, 41° to 85°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.

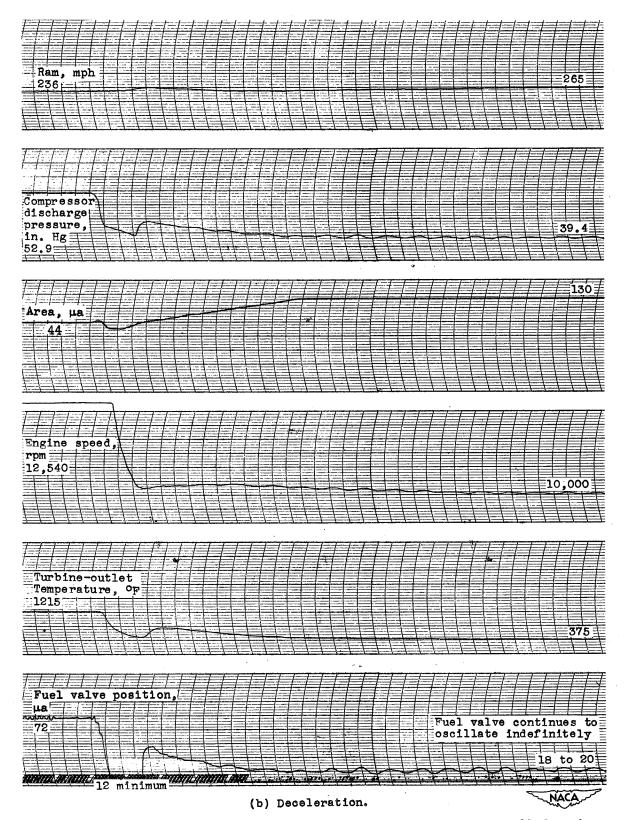


Figure 30. - Continued. Transient operation of automatically-controlled engine. Throttle position, 41° to 85°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.

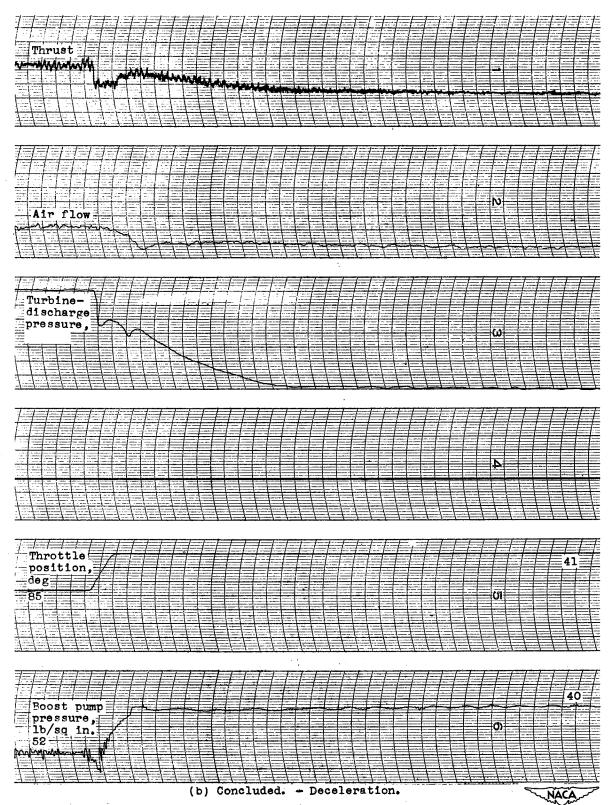


Figure 30. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 41° to 85°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.

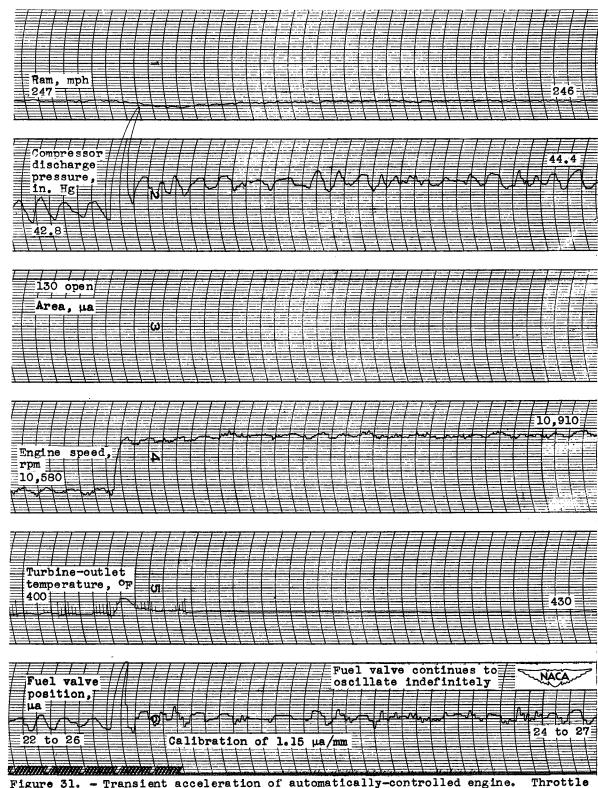


Figure 31. - Transient acceleration of automatically-controlled engine. Throttle position, 43.5 to 44; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.

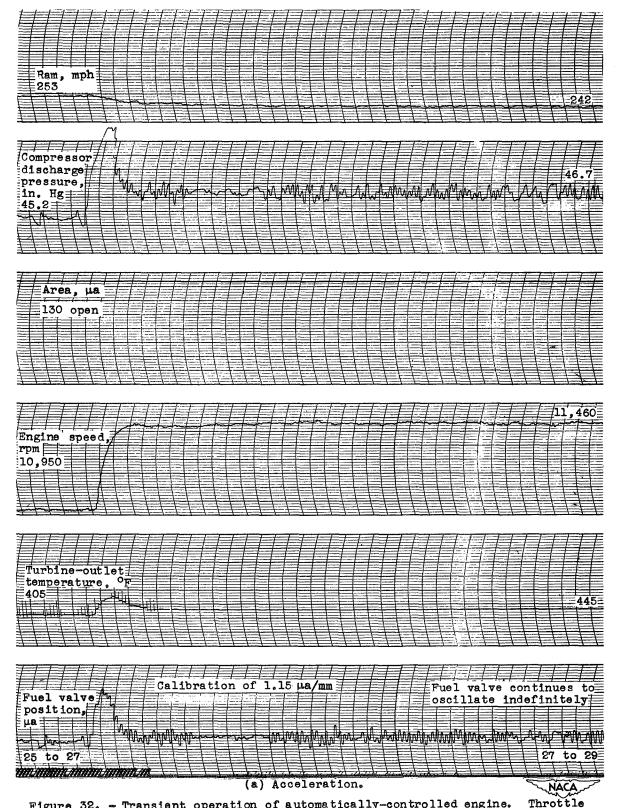


Figure 32. - Transient operation of automatically-controlled engine. Throttle position, 43.5° to 55°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2°

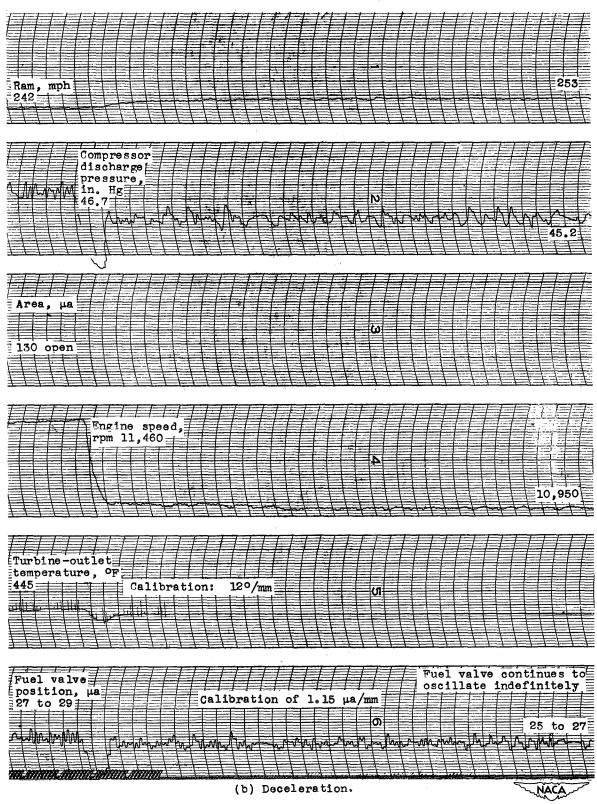
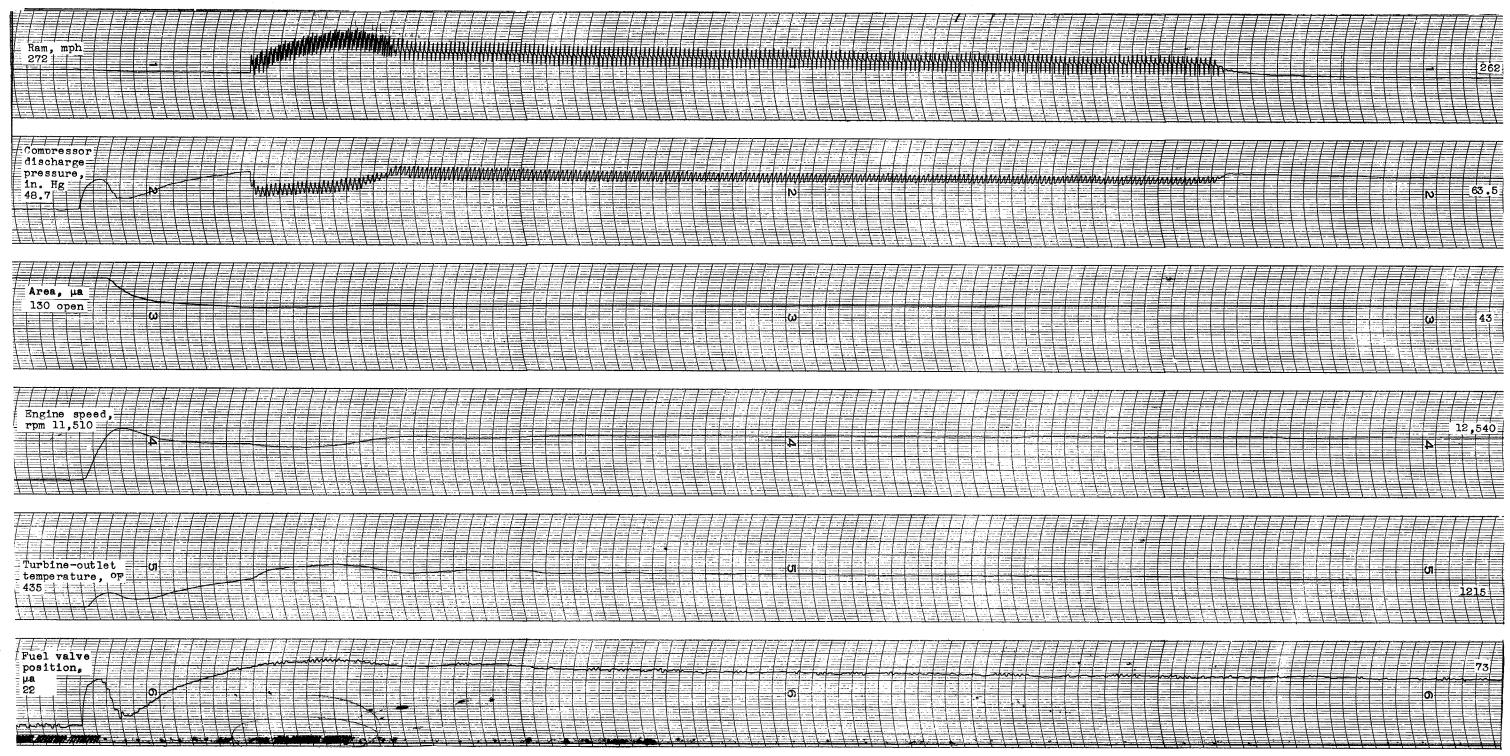


Figure 32. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 43.5° to 55°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.



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Figure 33. - Transient acceleration of automatically-controlled engine. Throttle position, 55° to 85°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.

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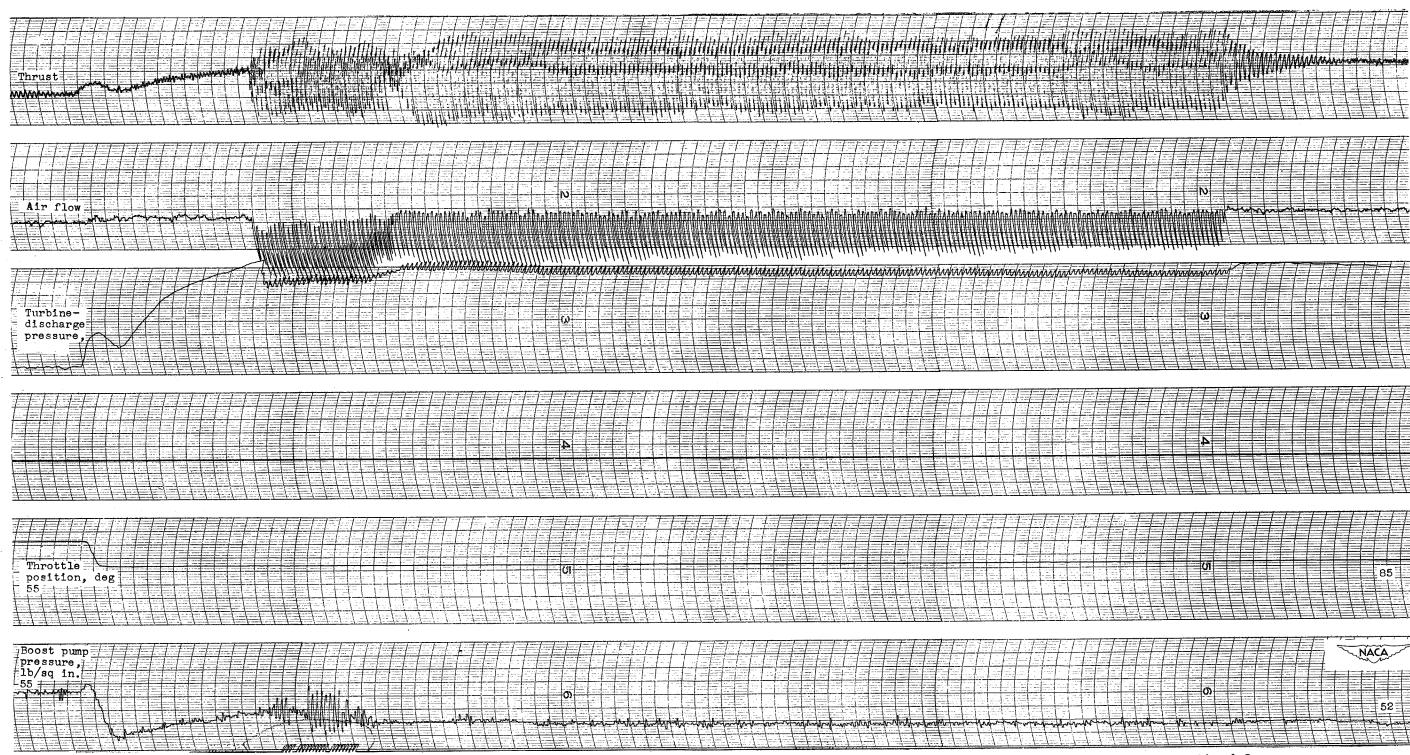


Figure 33. - Concluded. Transient acceleration of automatically-controlled engine. Throttle position, 55° to 85°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.

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Figure 34. - Transient acceleration of automatically-controlled engine. Throttle position, 70° to 85°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.

Turbinedischarge pressure, Throttle position, deg NACA Boost pump pressure, 1b/sq in. 

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Figure 34. - Concluded. Transient acceleration of automatically-controlled engine. Throttle position, 70° to 85°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2 acceleration.

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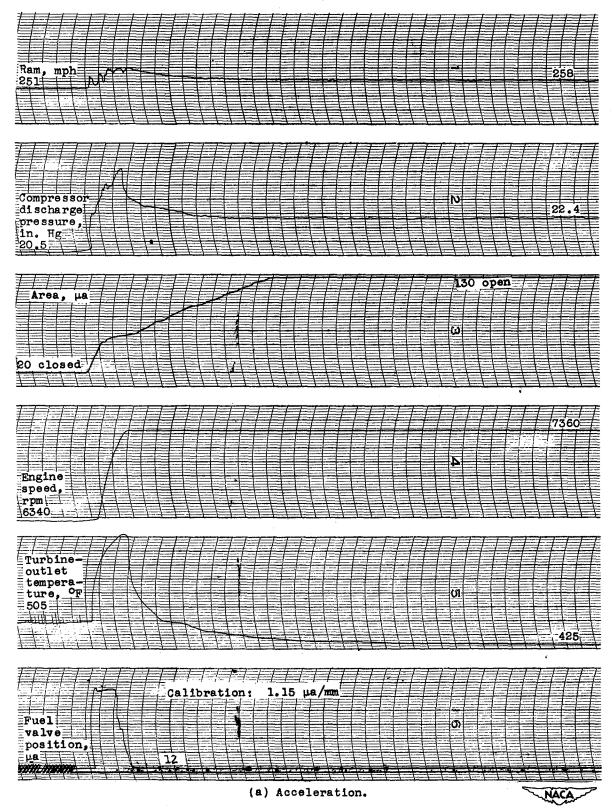


Figure 35. - Transient operation of automatically-controlled engine. Throttle position, 20.5° to 35°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.

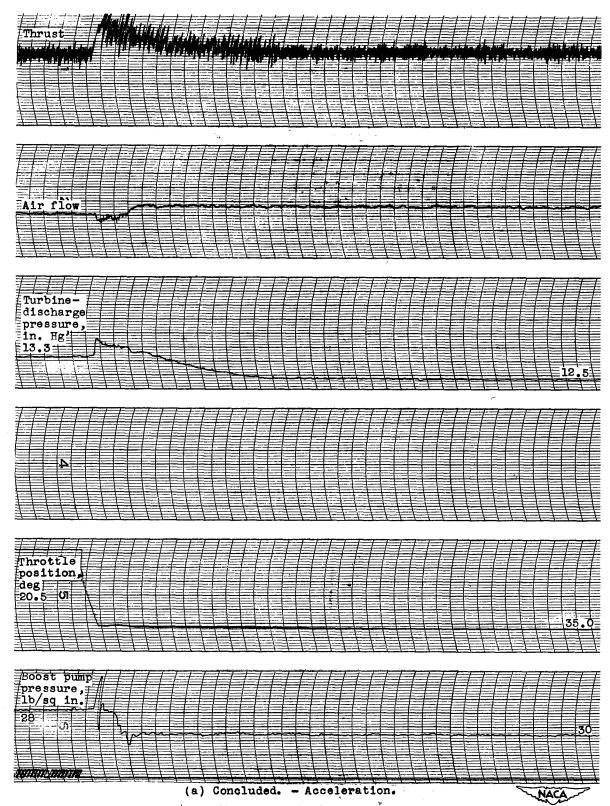


Figure 35. - Continued. Transient operation of automatically-controlled engine. Throttle position, 20.5° to 35°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.

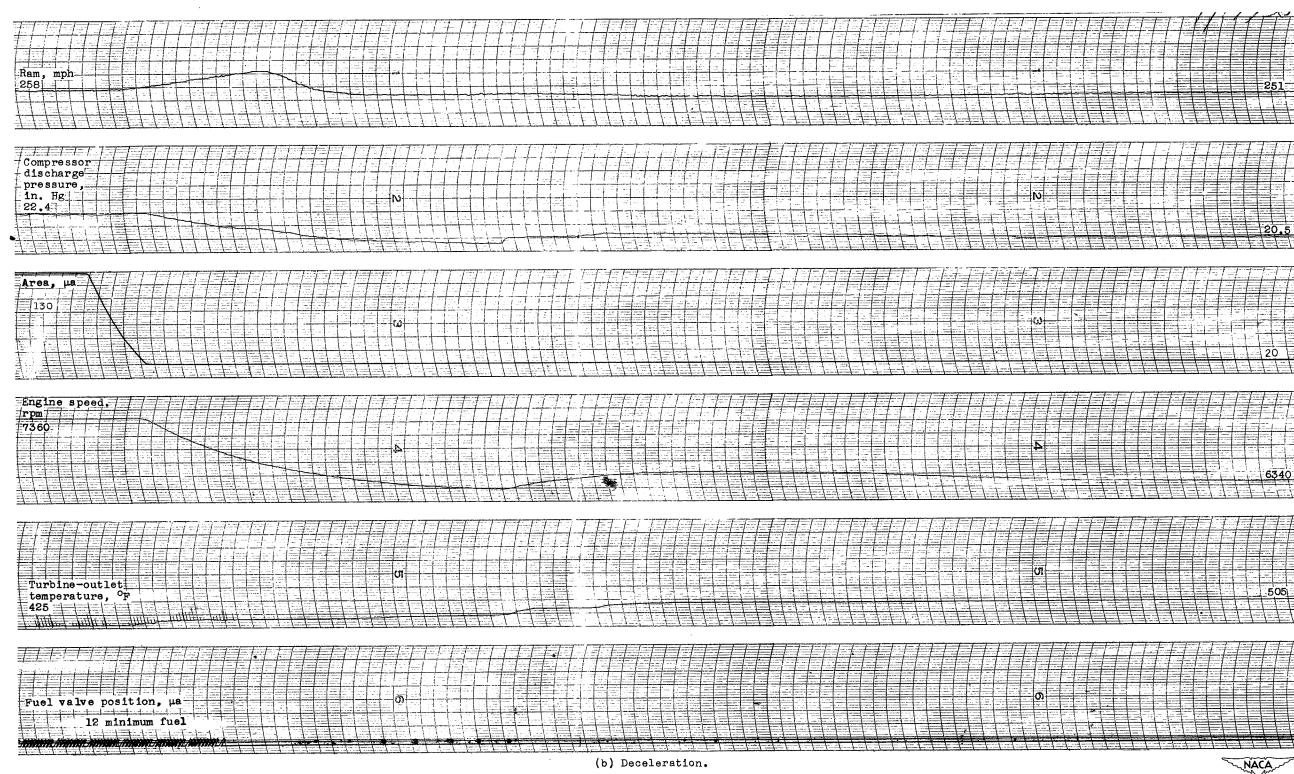


Figure 35. - Continued. Transient operation of automatically-controlled engine. Throttle position, 20.50 to 350; altitude, 25,000feet; nominal ram-pressure ratio, 1.2.

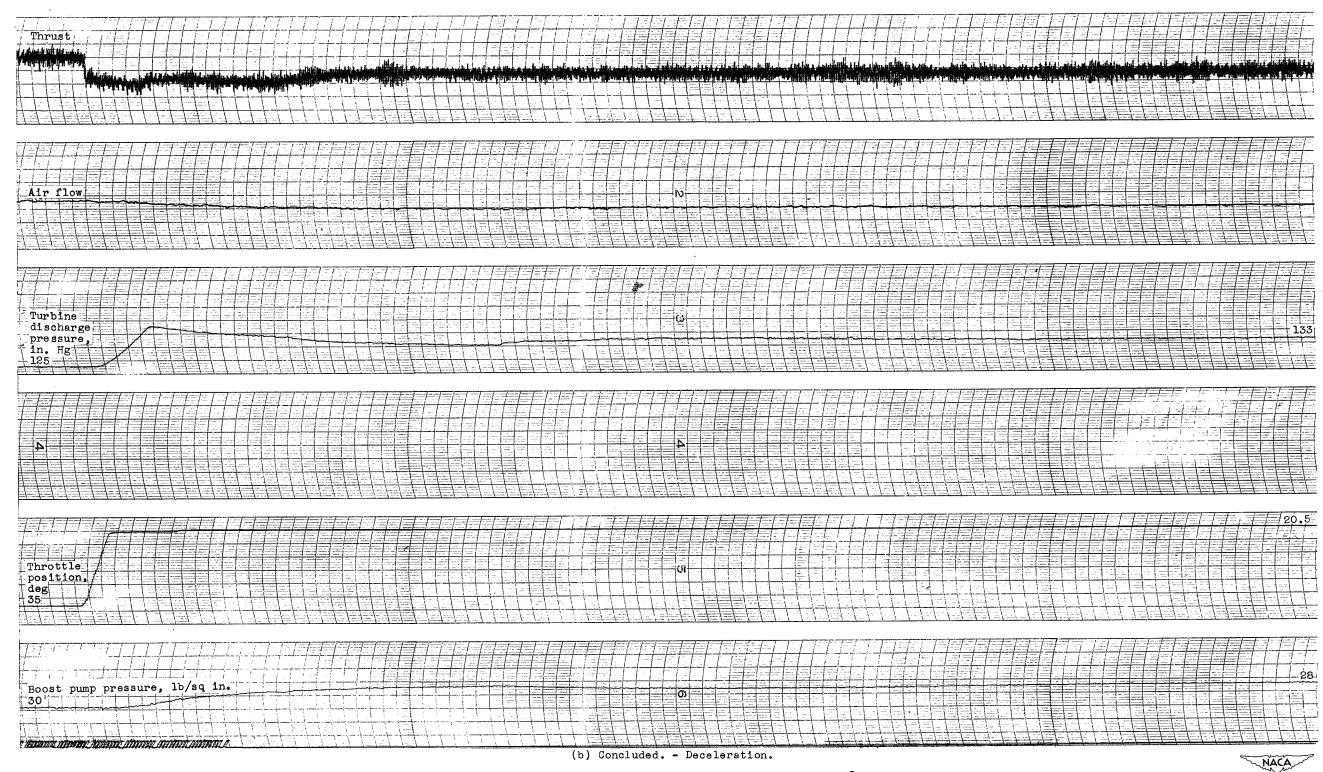
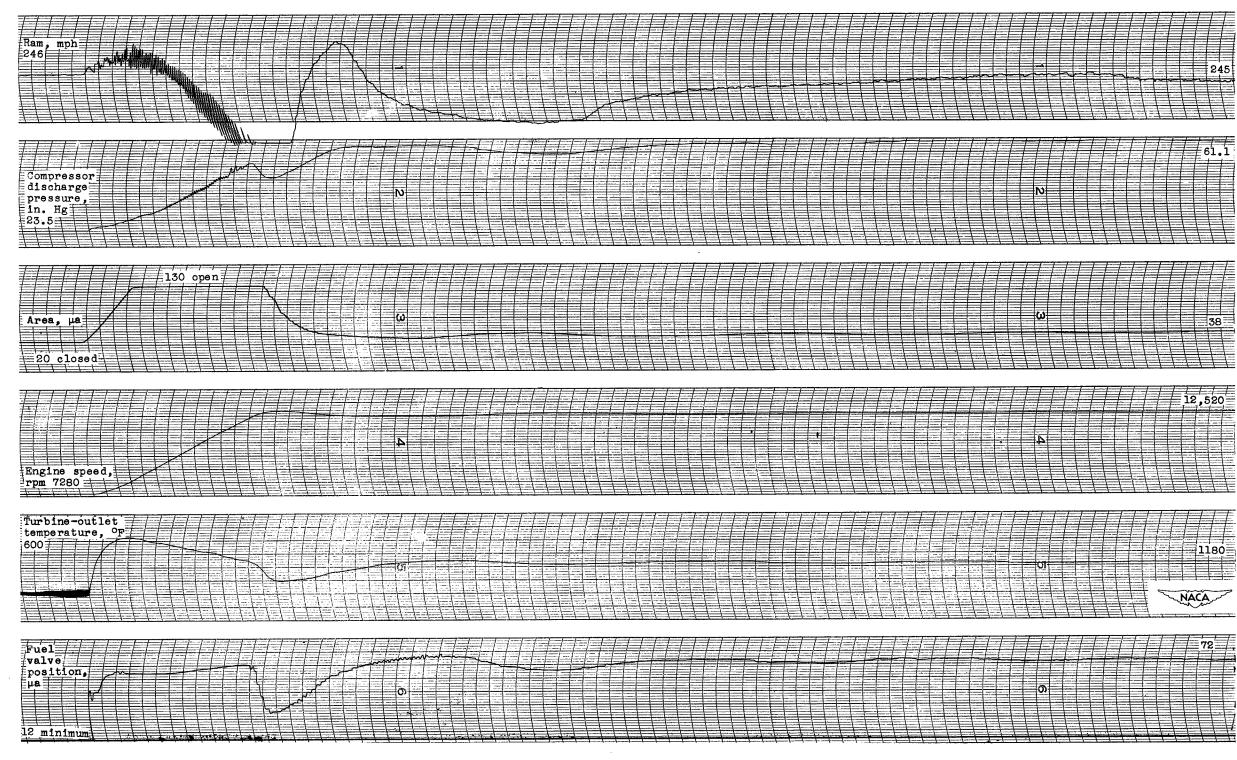
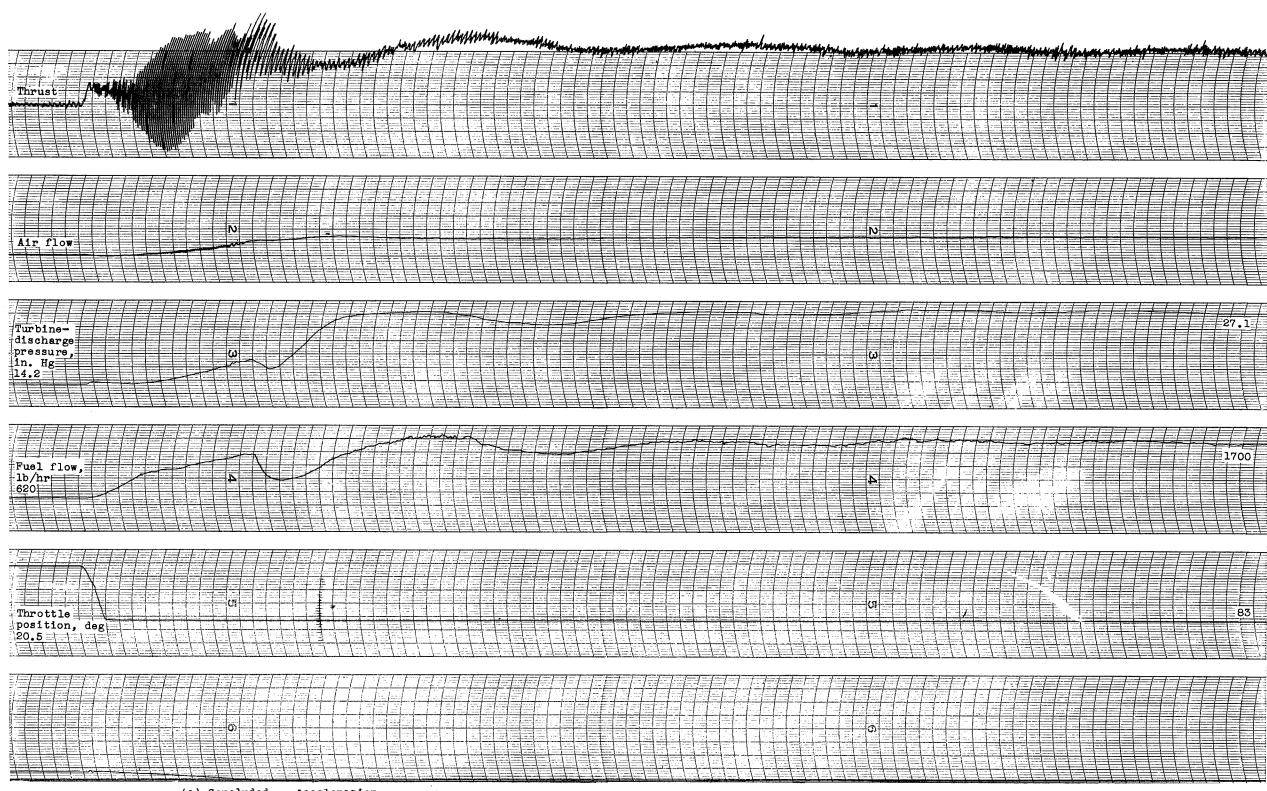


Figure 35. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 20.50 to 350; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.

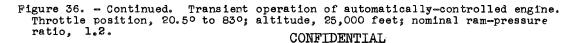


(a) Acceleration.

Figure 36. - Transient operation of automatically-controlled engine. Throttle position, 20.5° to 83°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.



(a) Concluded. - Acceleration.





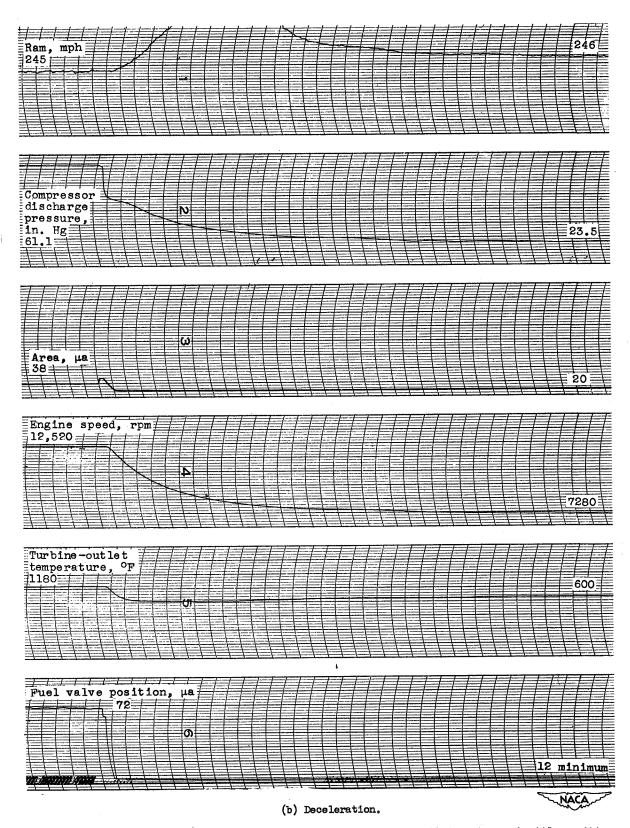


Figure 36. - Continued. Transient operation of automatically-controlled engine. Throttle position, 20.5° to 83°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.

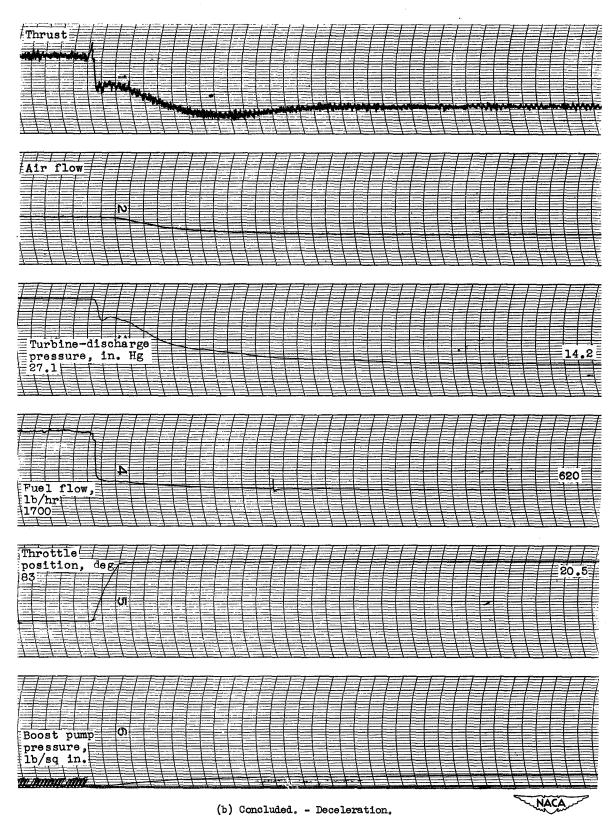


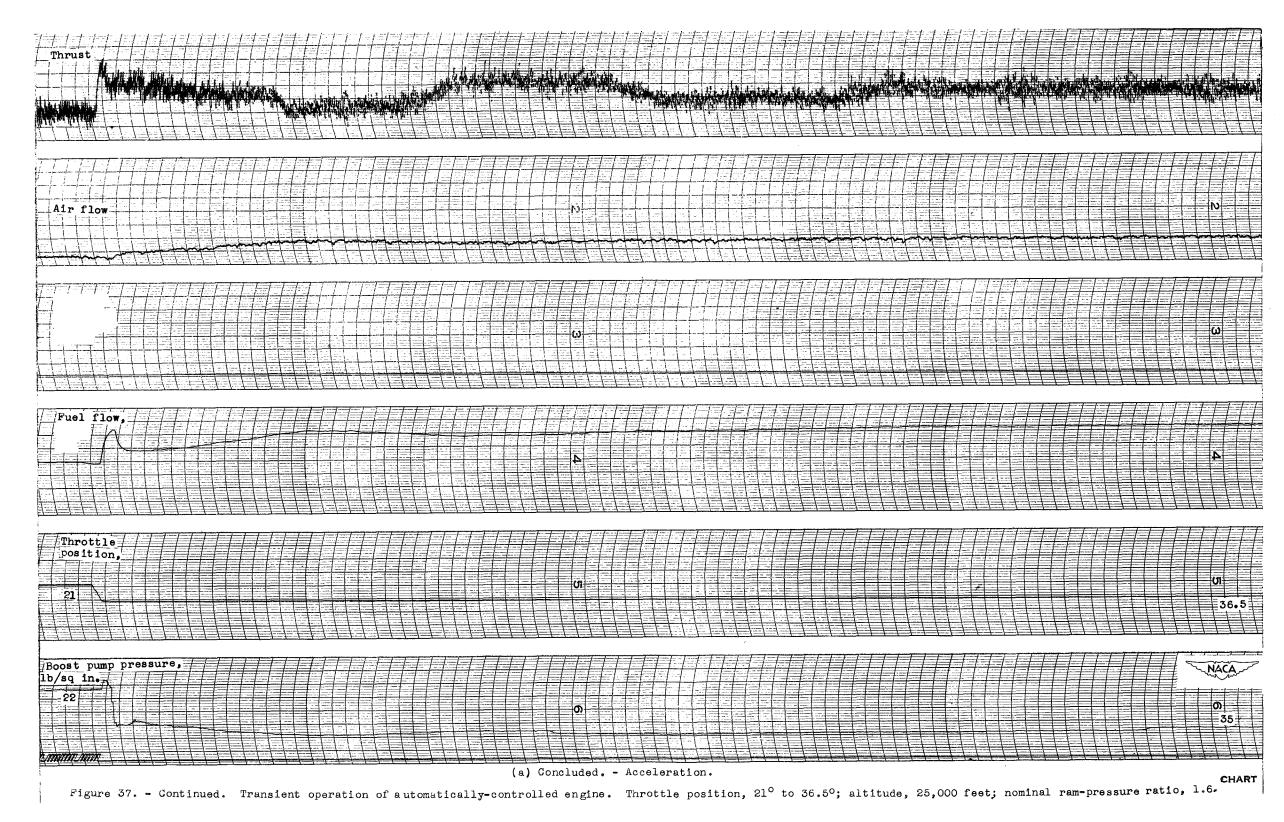
Figure 36. - Concluded. Transient operation of automatically-controlled engine. Throttle position,  $20.5^{\circ}$  to  $83^{\circ}$ ; altitude, 25,000 feet; nominal ram-pressure ratio, 1.2.

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Figure 37. - Transient operation of automatically-controlled engine. Throttle position, 21° to 36.5°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.6.

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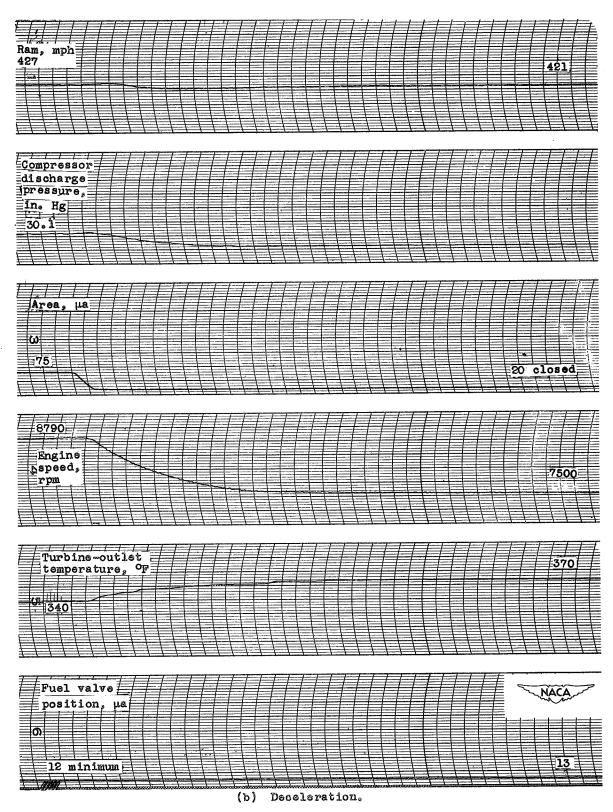


Figure 37. - Continued. Transient operation of automatically-controlled engine. Throttle position, 21° to 36.5°; altitude, 25,000-feet; nominal ram-pressure ratio, 1.6.

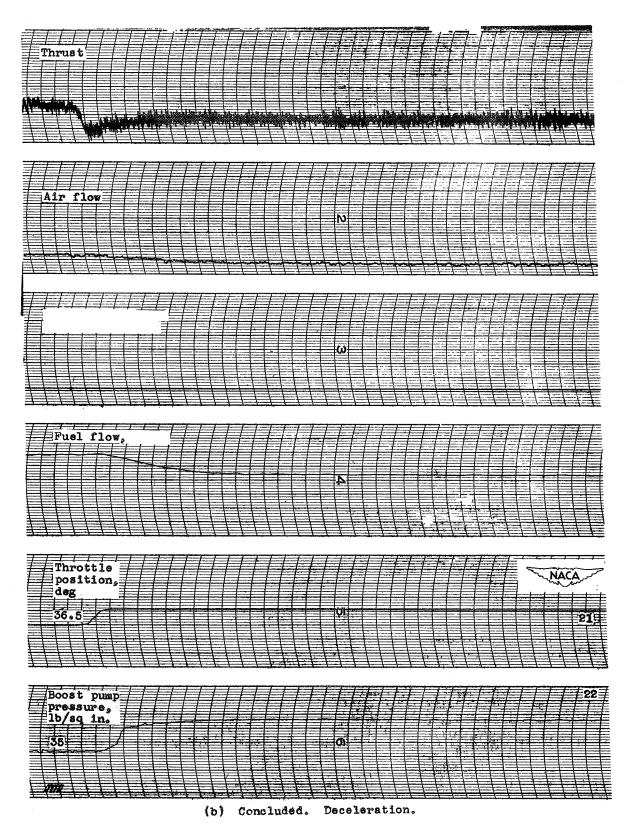


Figure 37. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 21° to 36.5°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.6.

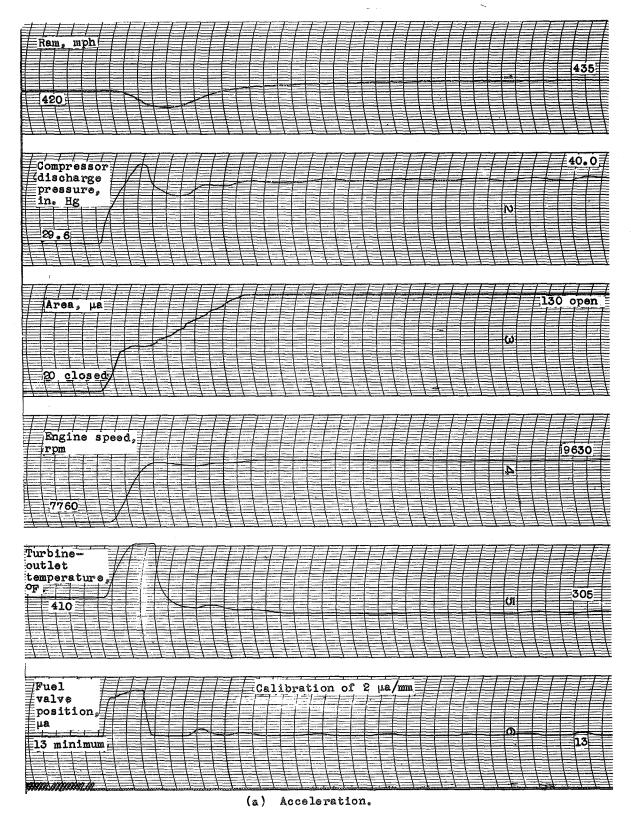


Figure 38. - Transient operation of automatically-controlled engine. Throttle position, 330 to 410; altitude, 25,000 feet; nominal ram-pressure ratio, 1.6.



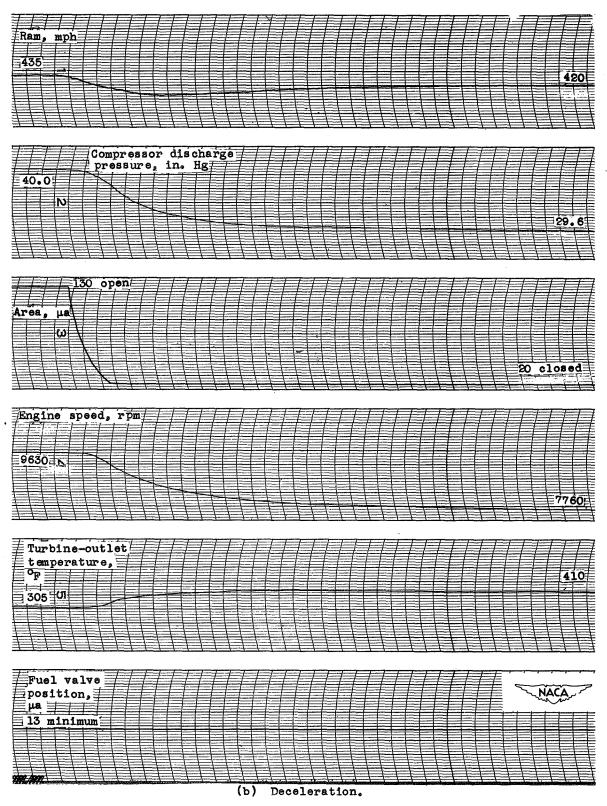


Figure 38. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 33 to 41°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.6.

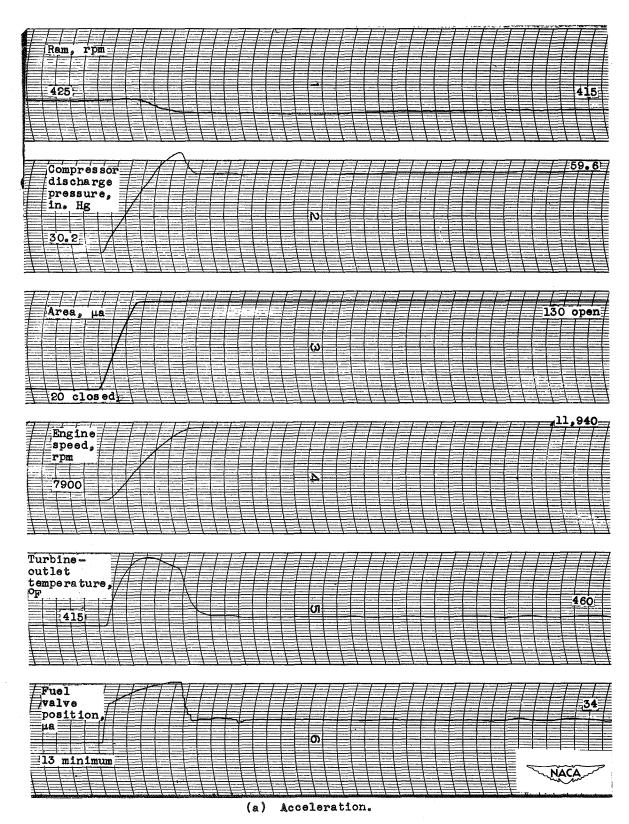


Figure 39. - Transient operation of automatically-controlled engine. Throttle position 33° to 62°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.6.

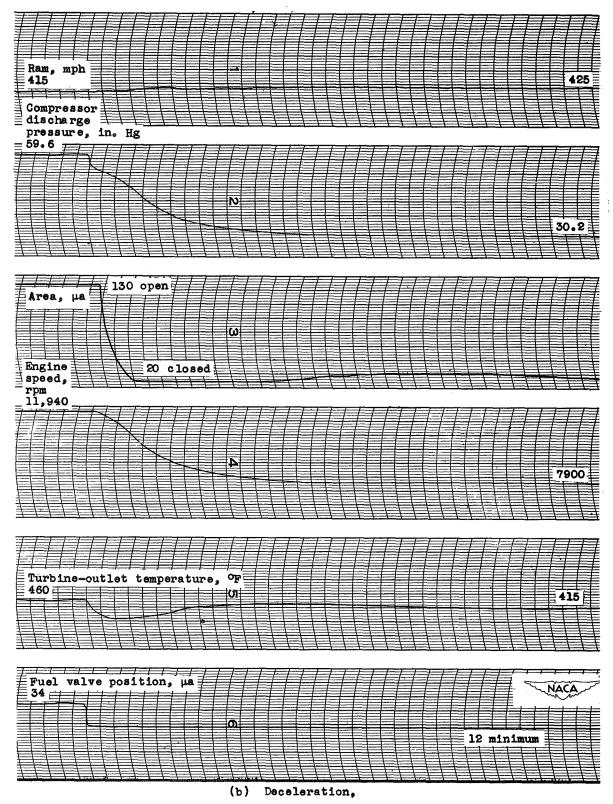


Figure 39. - Continued. Transient operation of automatically-controlled engine. Throttle position, 33° to 62°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.6.

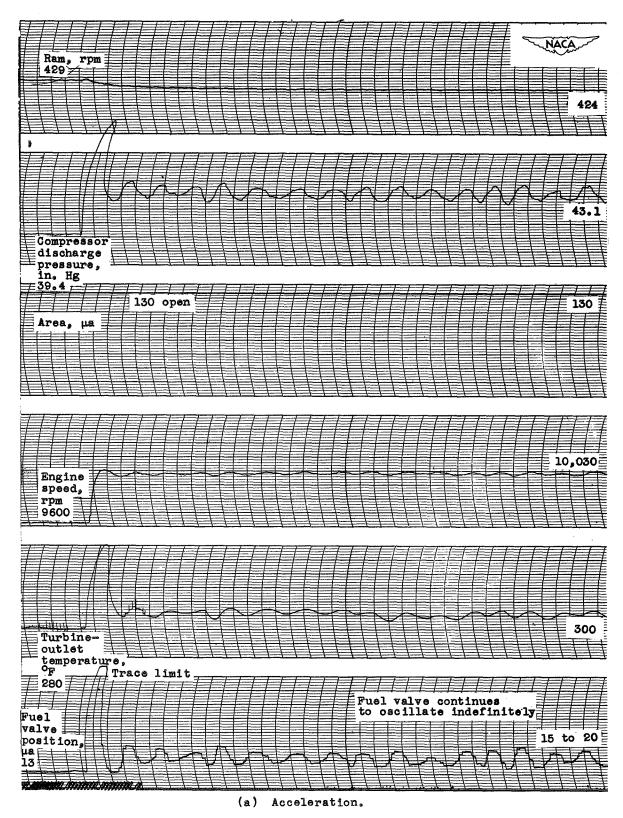


Figure 40. - Transient operation of automatically-controlled engine. Throttle position, 40° to 41°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.6.

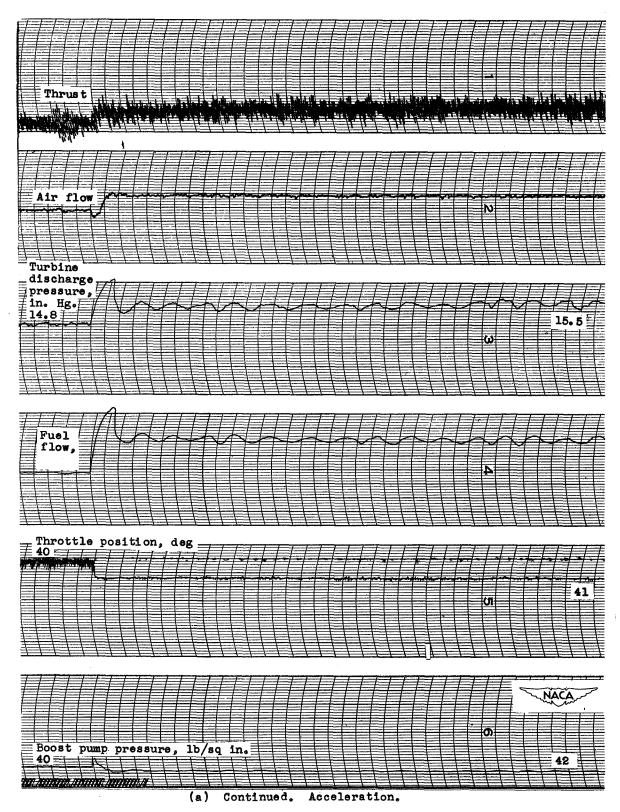


Figure 40. - Continued. Transient operation of automatically-controlled engine. Throttle position, 40° to 41°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.6.

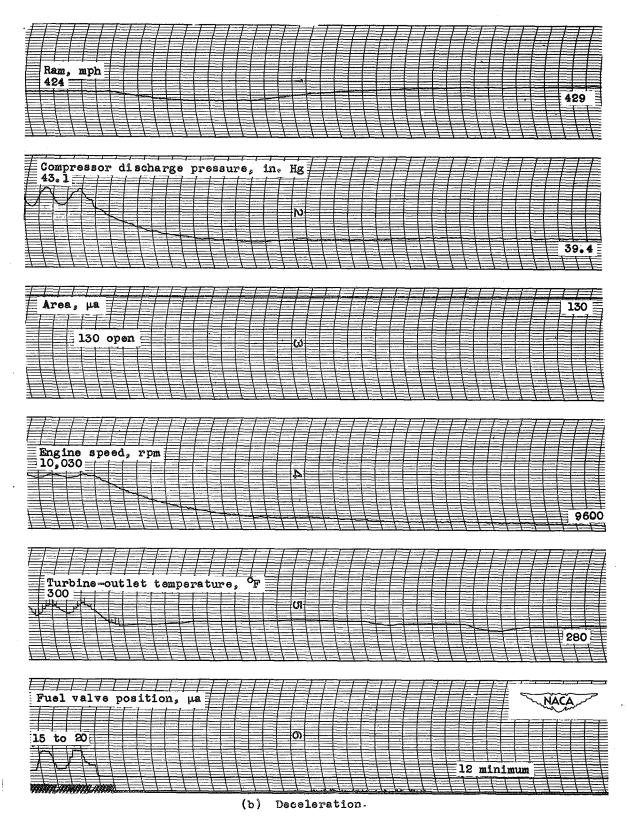


Figure 40. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 40° to 41°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.6.

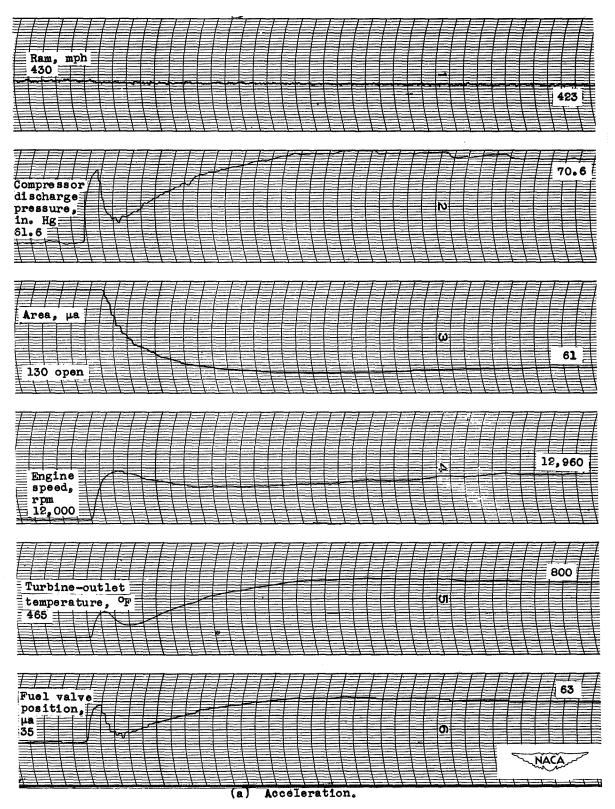


Figure 41. - Transient operation of automatically-controlled engine. Throttle position 62° to 71°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.6.



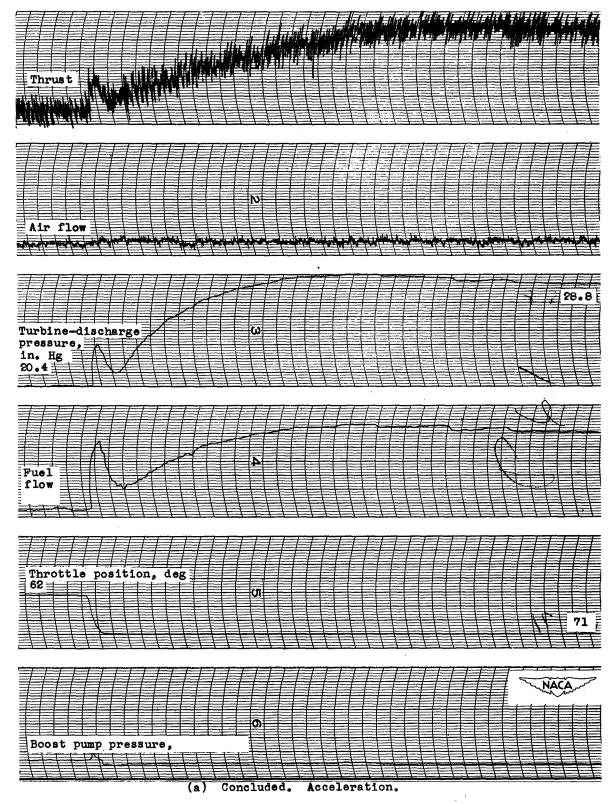


Figure 41. - Continued. Transient operation of automatically-controlled engine. Throttle position 62° to 71°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.6.

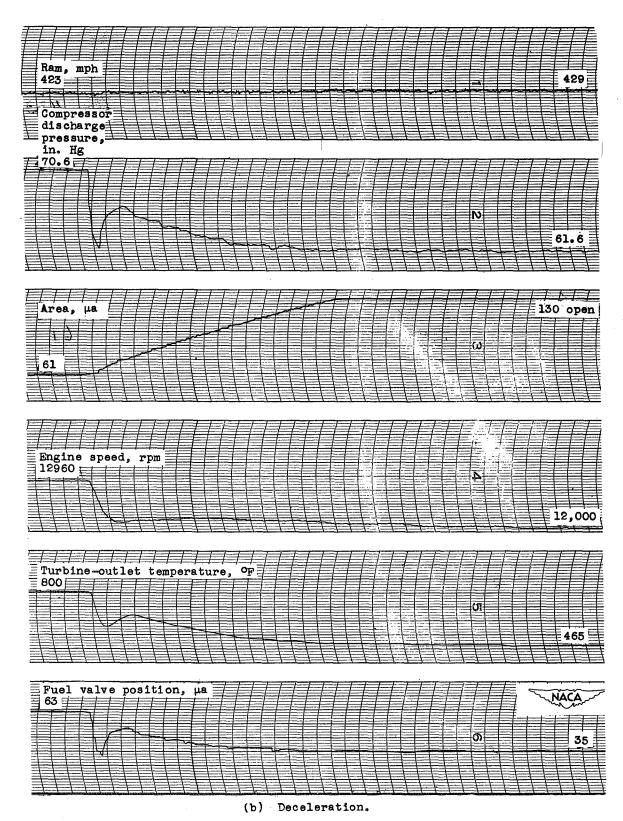
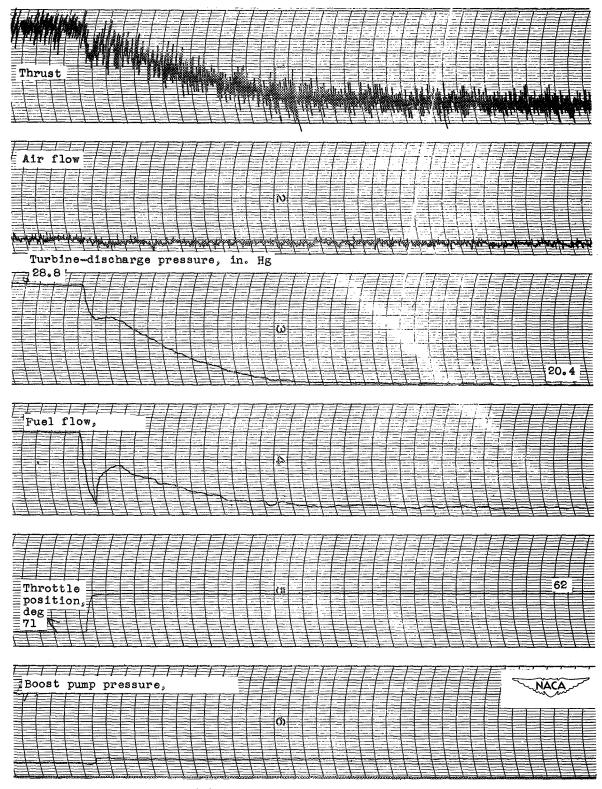


Figure 41. - Continued. Transient operation of automatically-controlled engine. Throttle position, 62° to 71°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.6.



(b) Concluded. Deceleration.

Figure 41. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 62° to 71°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.6.

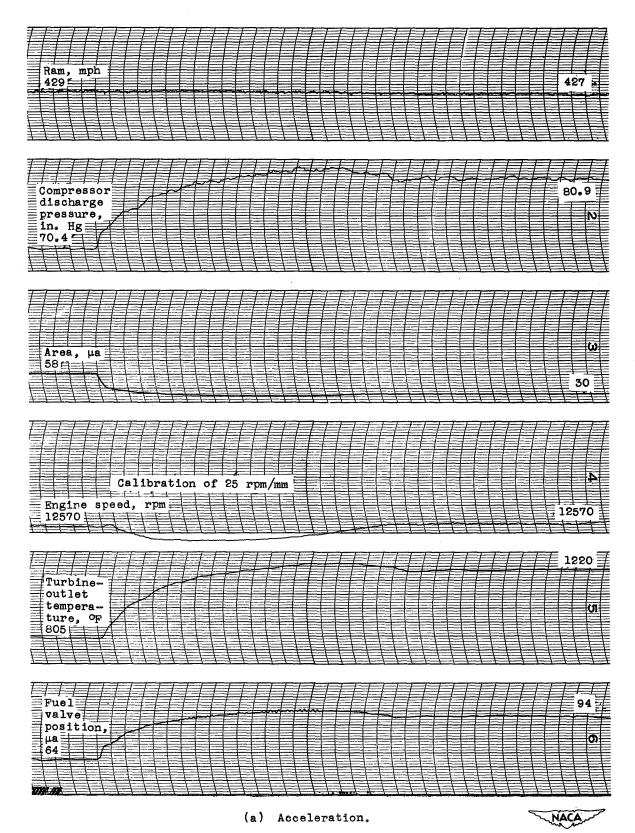
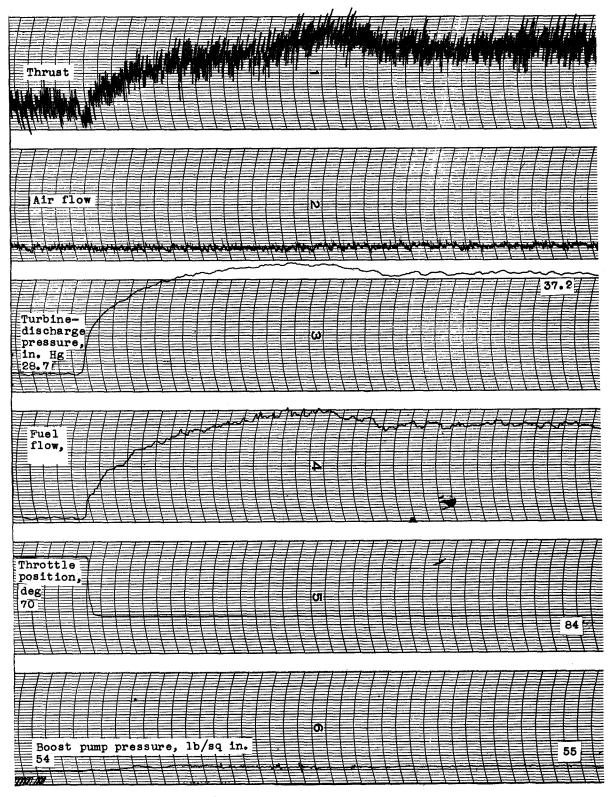


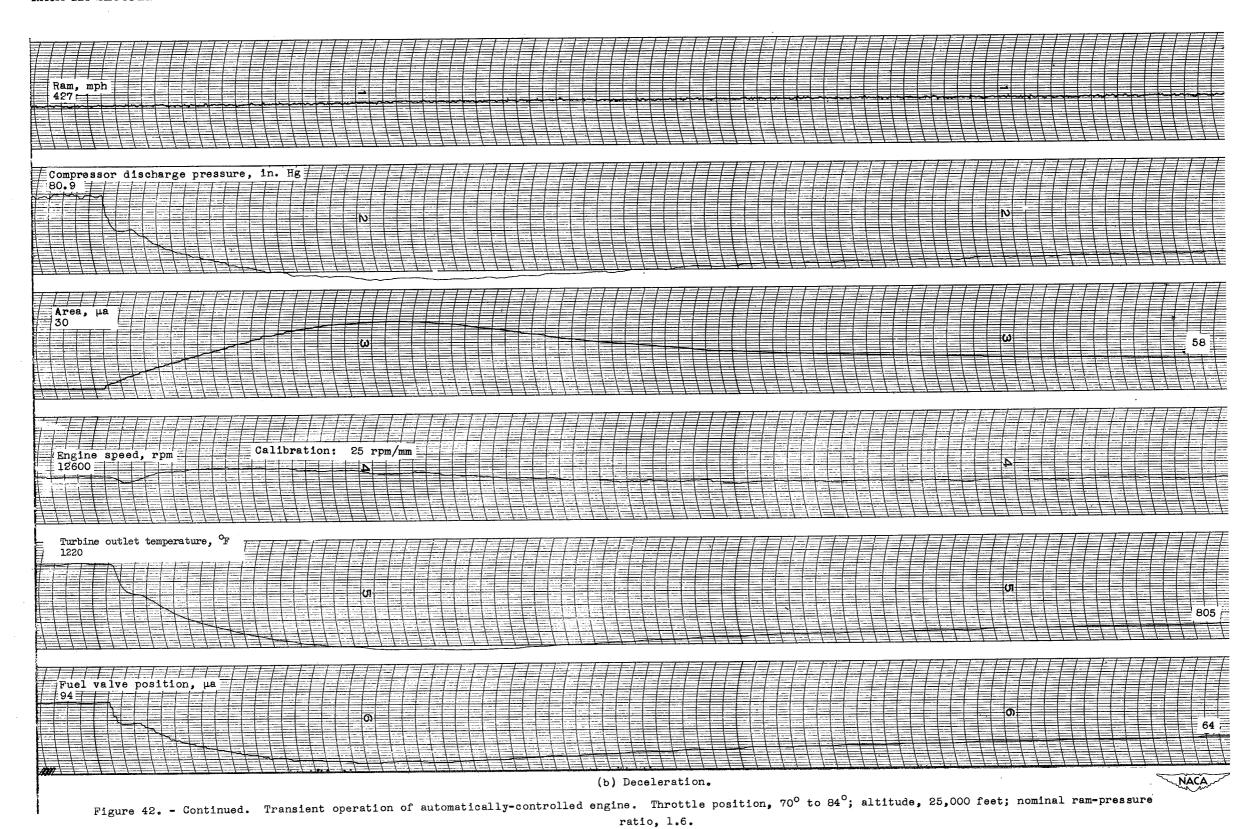
Figure 42. - Transient operation of automatically-controlled engine. Throttle position, 70° to 84°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.6.



(a) Concluded. Acceleration.

Figure 42. - Continued. Transient operation of automatically-controlled engine. Throttle position, 70° to 84°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.6.





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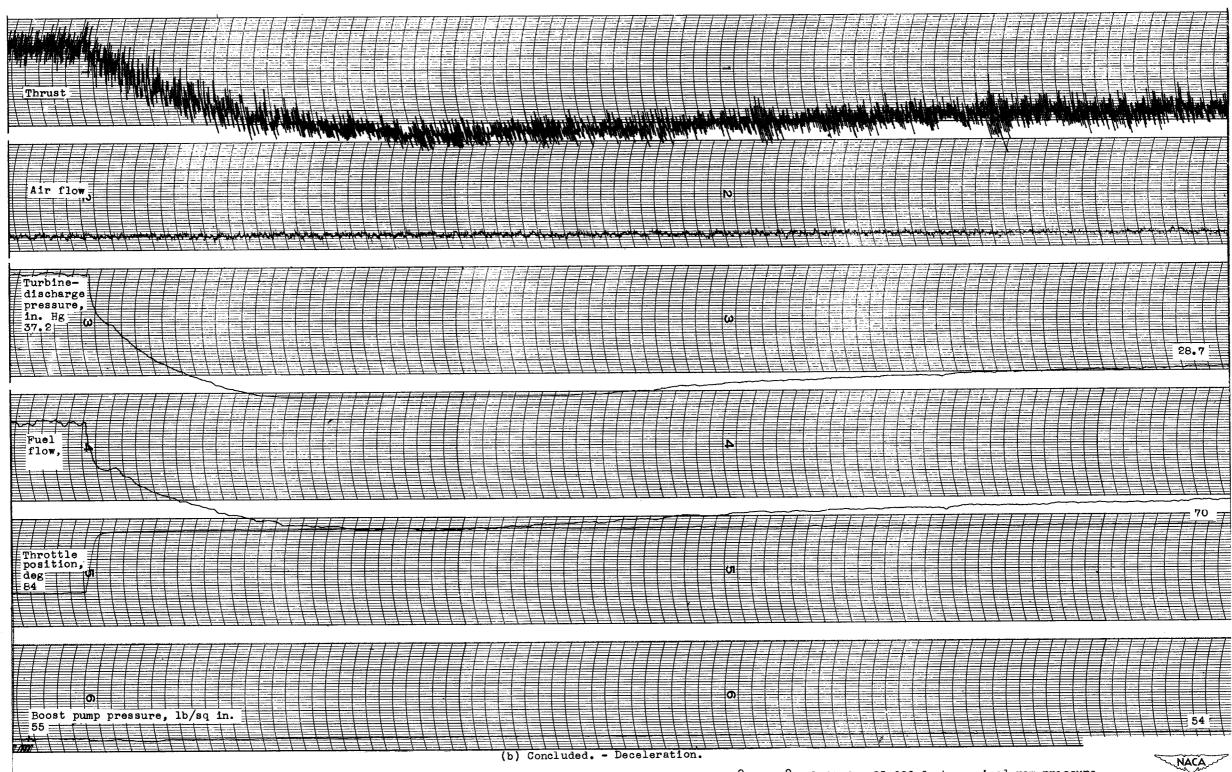


Figure 42. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 70° to 84°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.6.

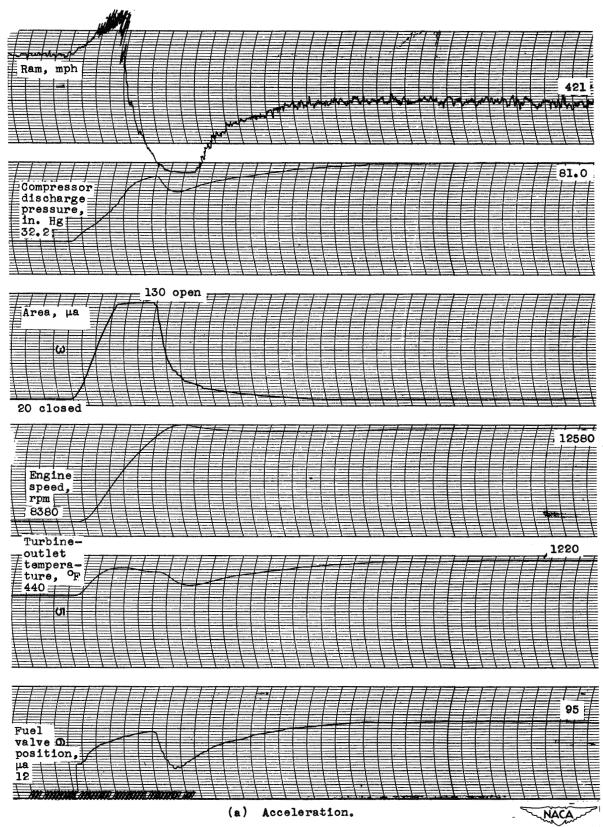


Figure 43. - Transient operation of automatically-controlled engine. Throttle position, 34 to 84; altitude, 25,000 feet; nominal ram-pressure ratio, 1.6.

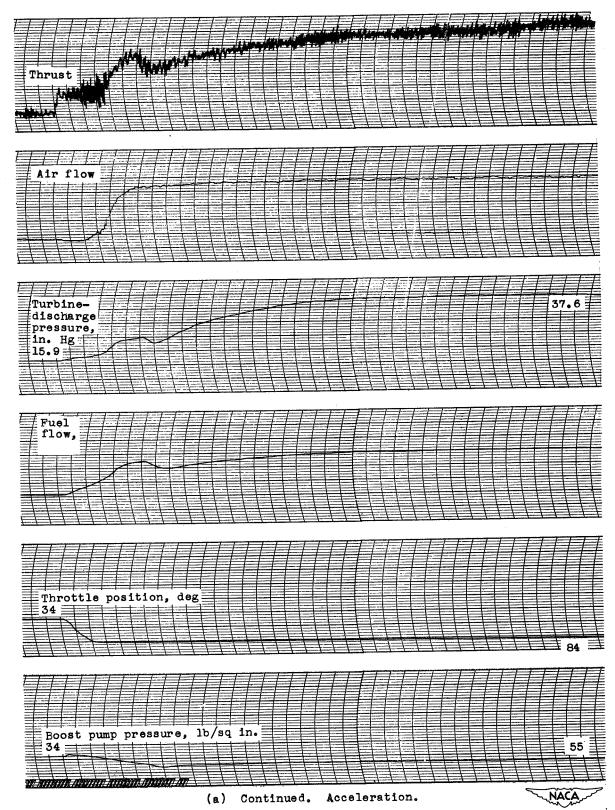


Figure 43. - Continued. Transient operation of automatically-controlled engine. Throttle position, 34 to 84; altitude, 25,000 feet; nominal ram-pressure ratio, 1.6.

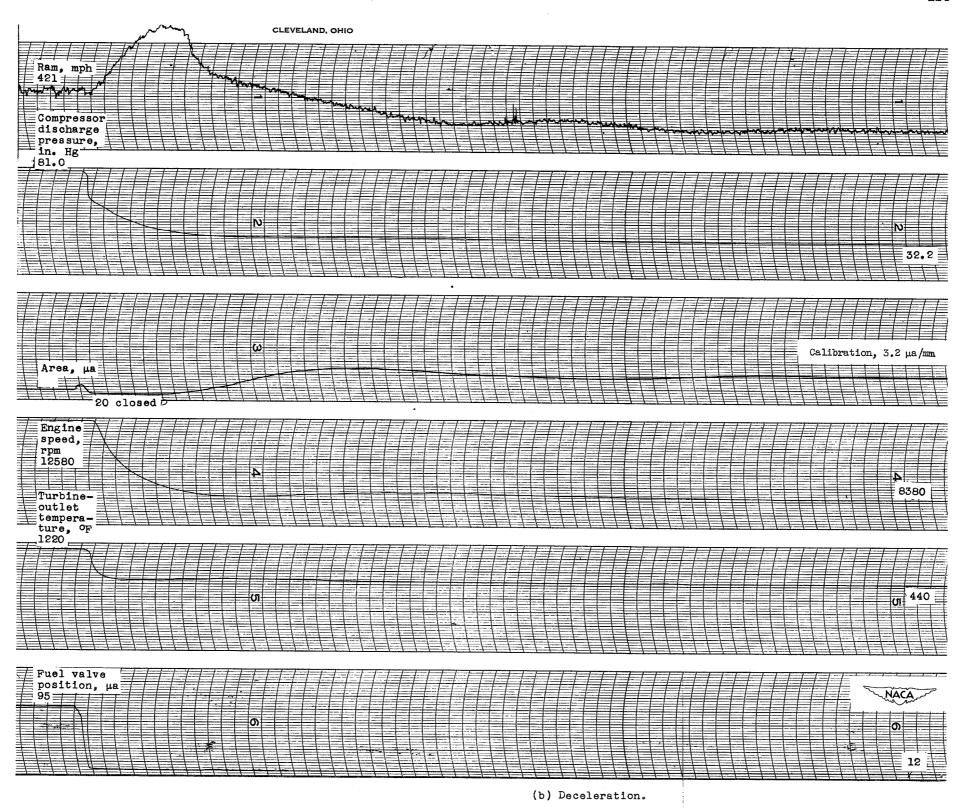


Figure 43. - Continued. Transient operation of automatically-controlled engine. Throttle position, 34° to 84°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.6.

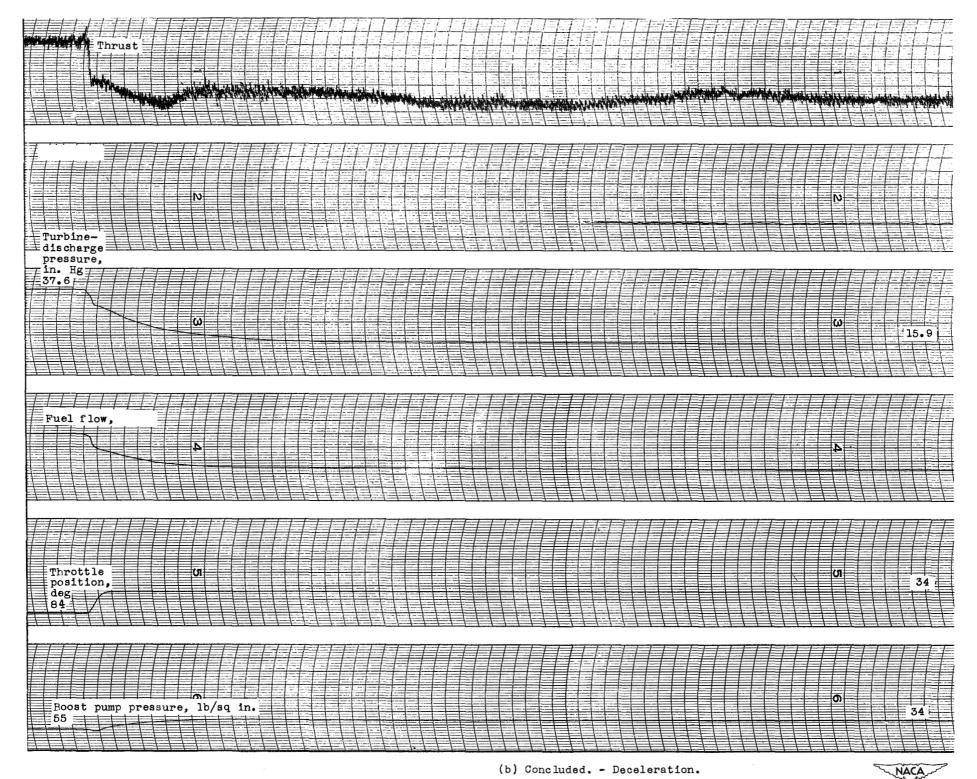


Figure 43. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 34° to 84°; altitude, 25,000 feet; nominal ram-pressure ratio, 1.6.

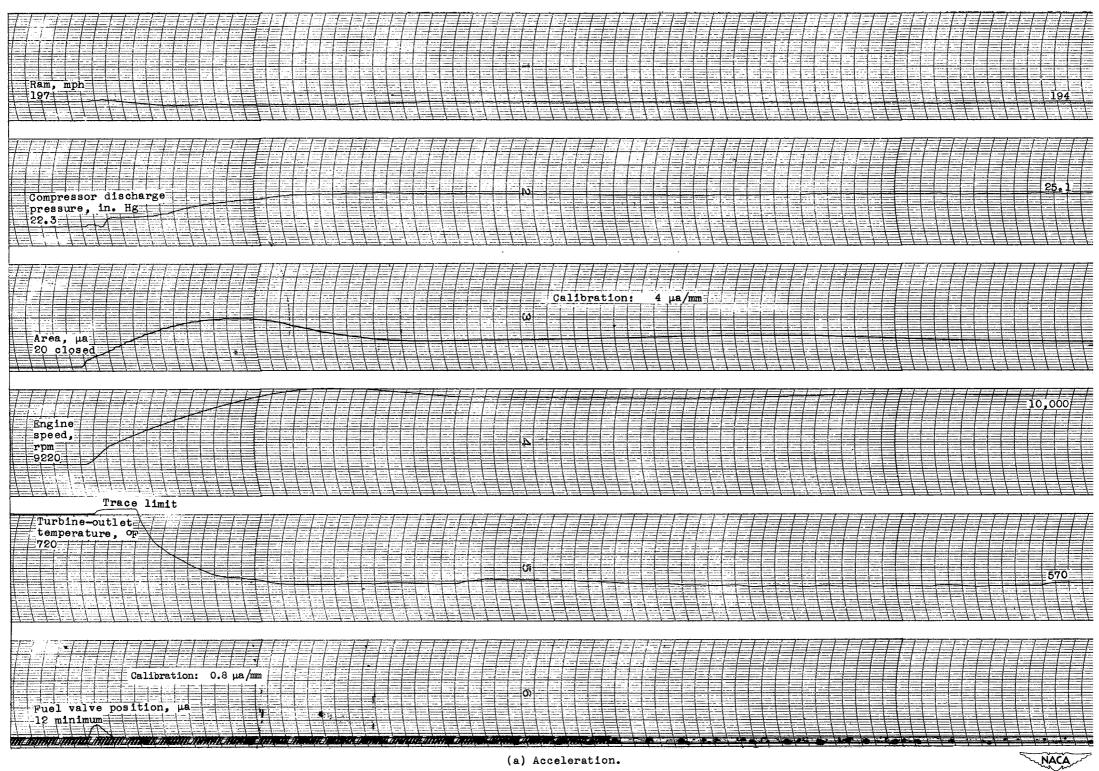


Figure 44. - Transient operation of automatically-controlled engine. Throttle position, 37° to 40°; altitude, 35,000 feet; nominal ram-pressure ratio, 1.2.

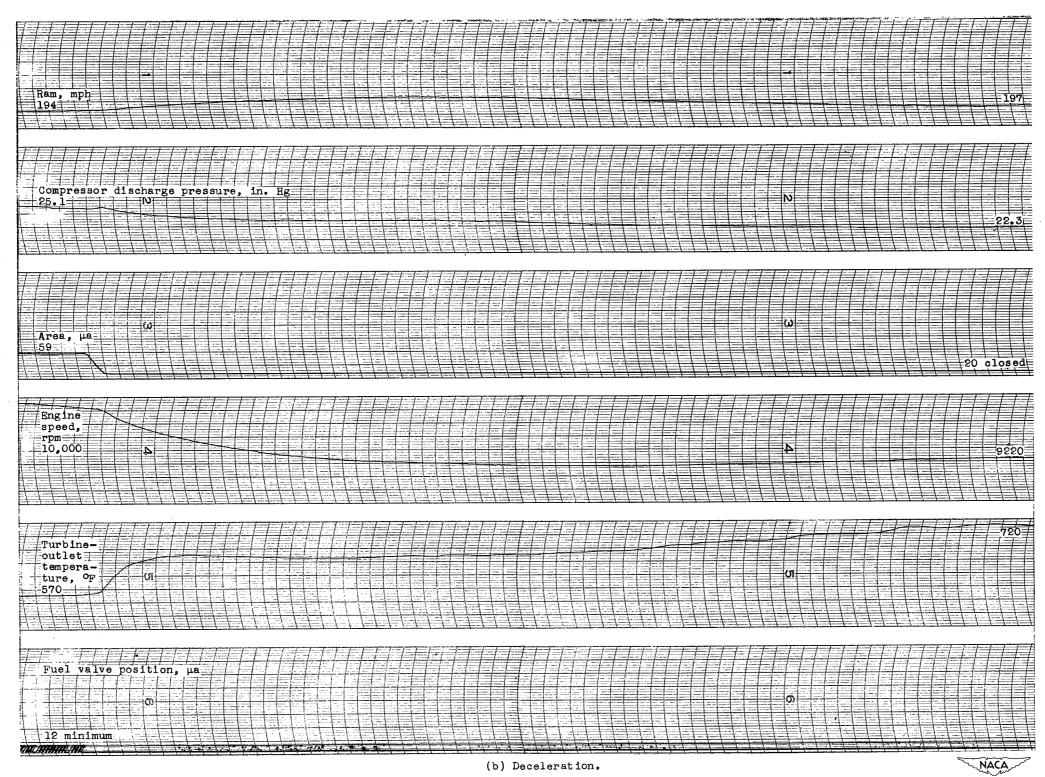


Figure 44. - Continued. Transient operation of automatically-controlled engine. Throttle position, 37° to 40°; altitude, 35,000 feet; nominal ram-pressure ratio, 1.2.

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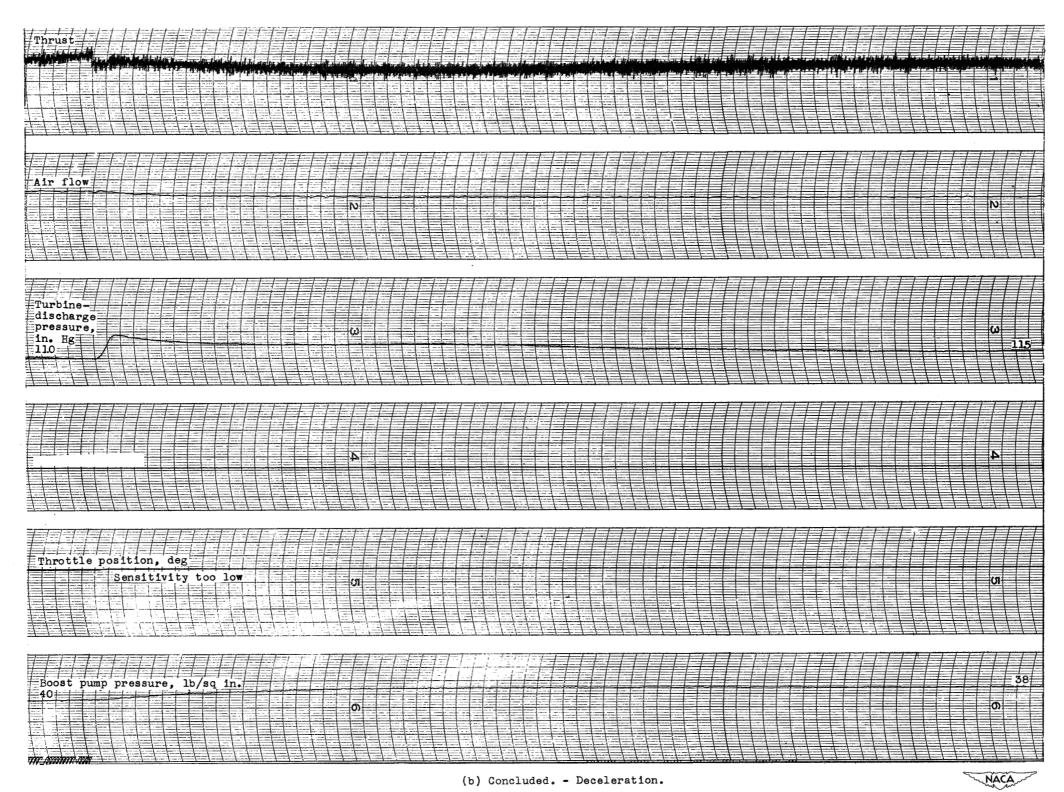


Figure 44. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 37° to 40°; altitude, 35,000 feet; nominal ram-pressure ratio, 1.2.

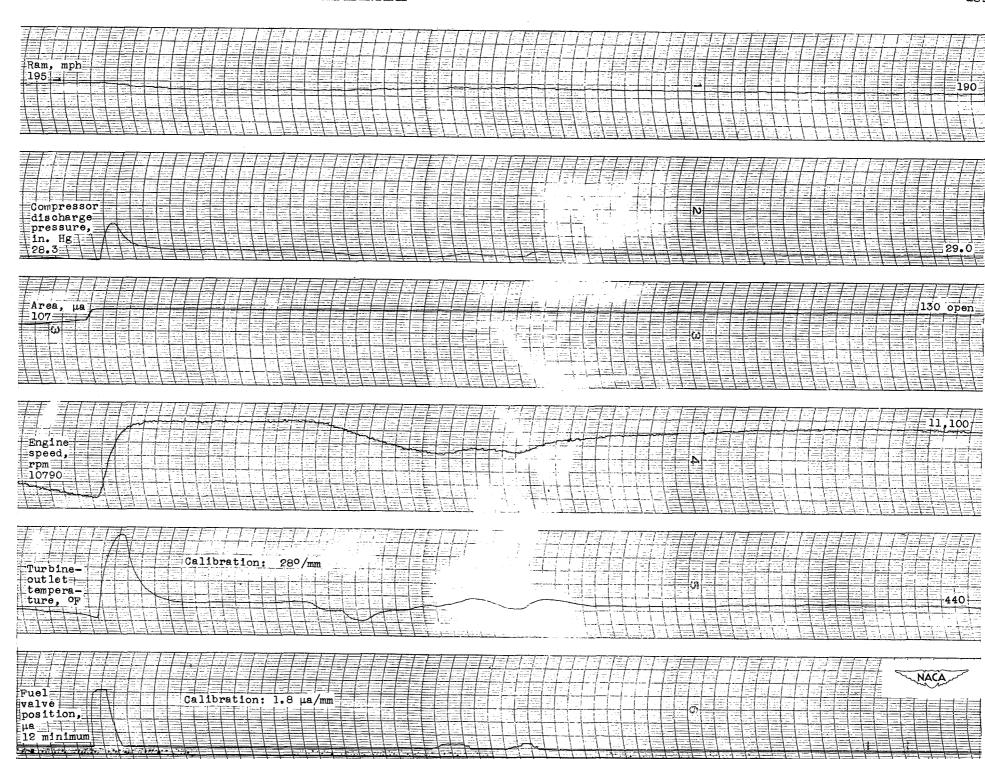


Figure 45. - Transient acceleration of automatically-controlled engine. Throttle position, 41.5° to 44.5°; altitude, 35,000 feet; nominal ram-pressure ratio, 1.2.

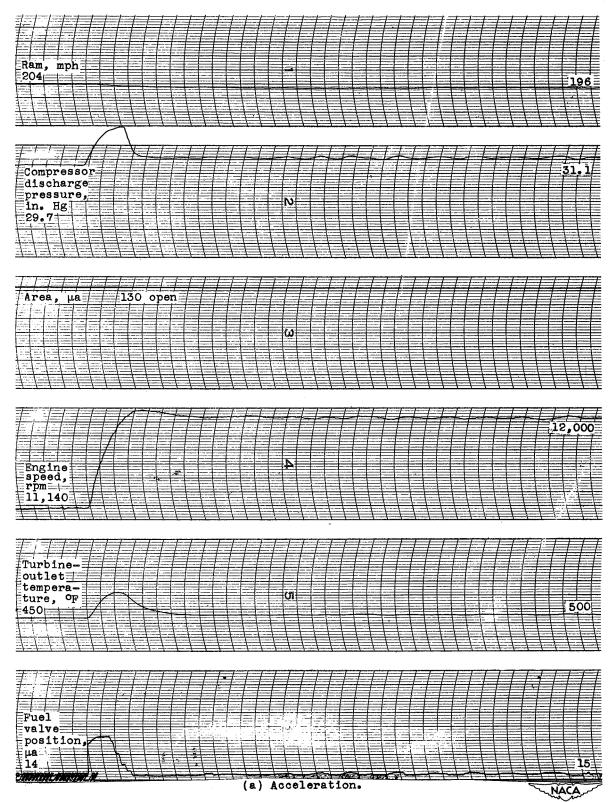


Figure 46. - Transient operation of automatically-controlled engine. Throttle position, 46.5° to 64.5°; altitude, 35,000 feet; nominal ram-pressure ratio, 1.2.

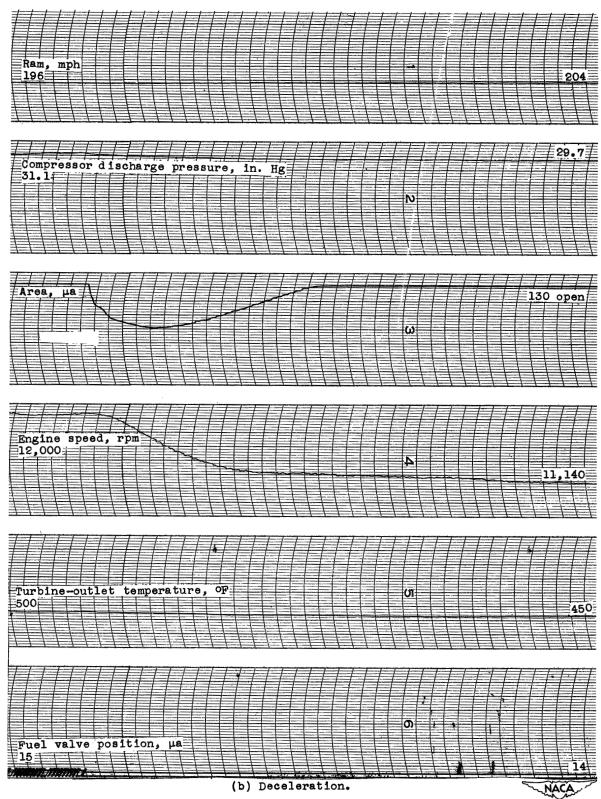


Figure 46. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 46.5° to 64.5°; altitude, 35,000 feet; nominal ram-pressure ratio, 1.2.

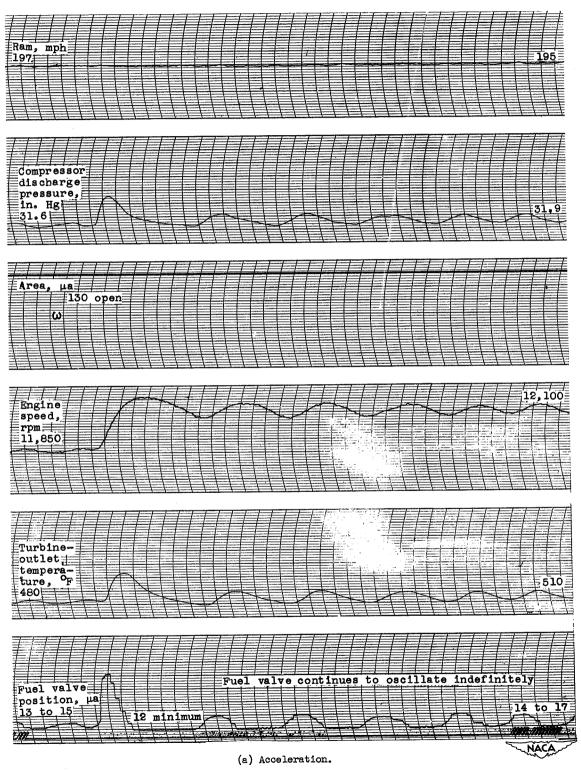


Figure 47. - Transient operation of automatically-operated engine. Throttle position, 60.5° to 64°; altitude, 35,000 feet; nominal ram-pressure ratio, 1.2.

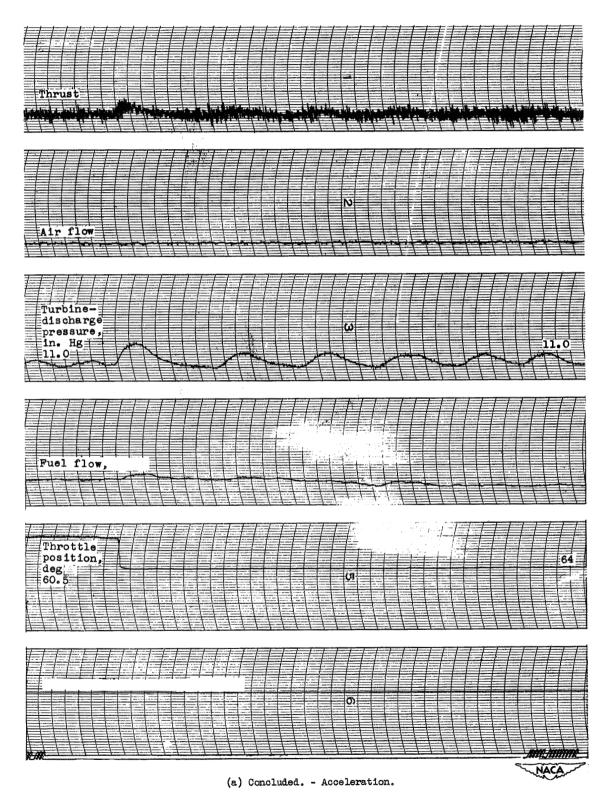


Figure 47. - Continued. Transient operation of automatically-controlled engine. Throttle position, 60.5° to 64°; altitude, 35,000 feet; nominal ram-pressure ratio, 1.2.

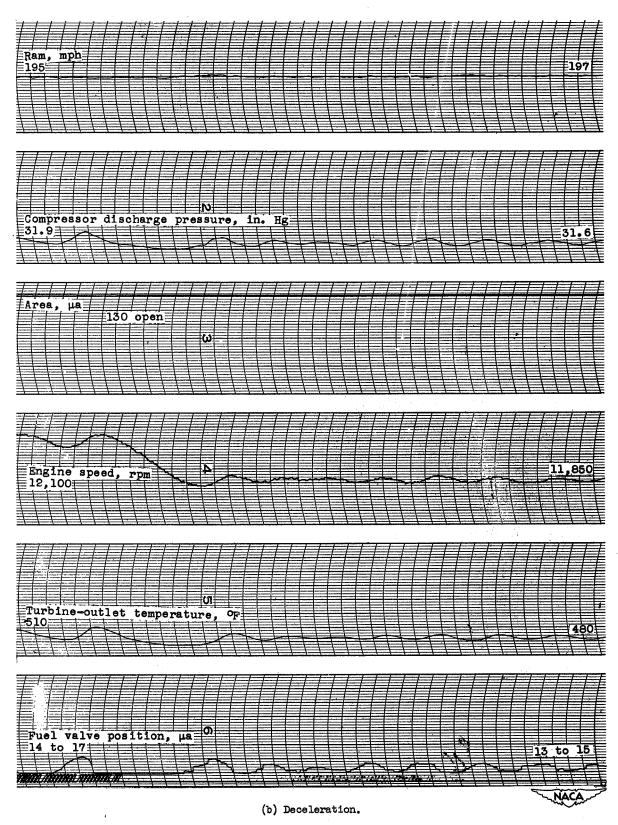


Figure 47. - Continued. Transient operation of automatically-controlled engine. Throttle position, 60.5° to 64°; altitude, 35,000 feet, nominal ram-pressure ratio, 1.2.

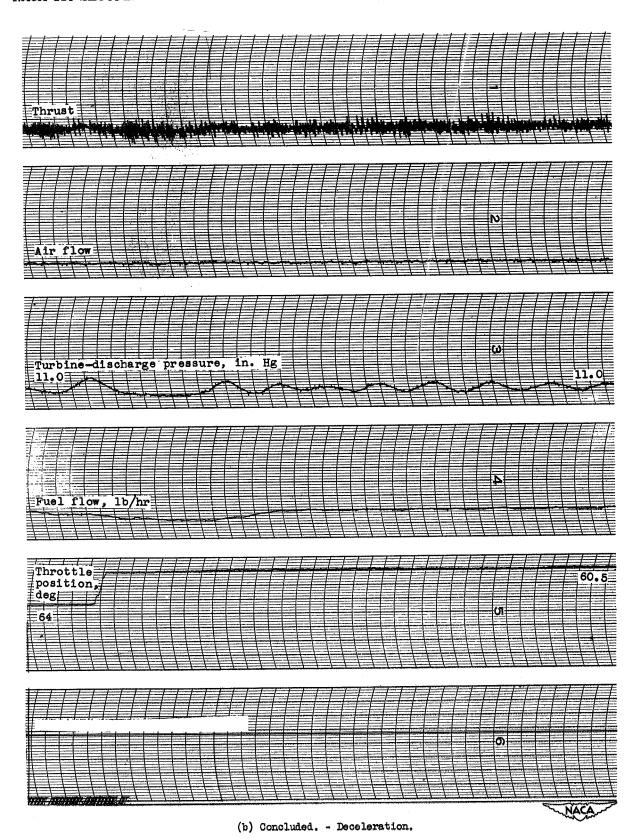
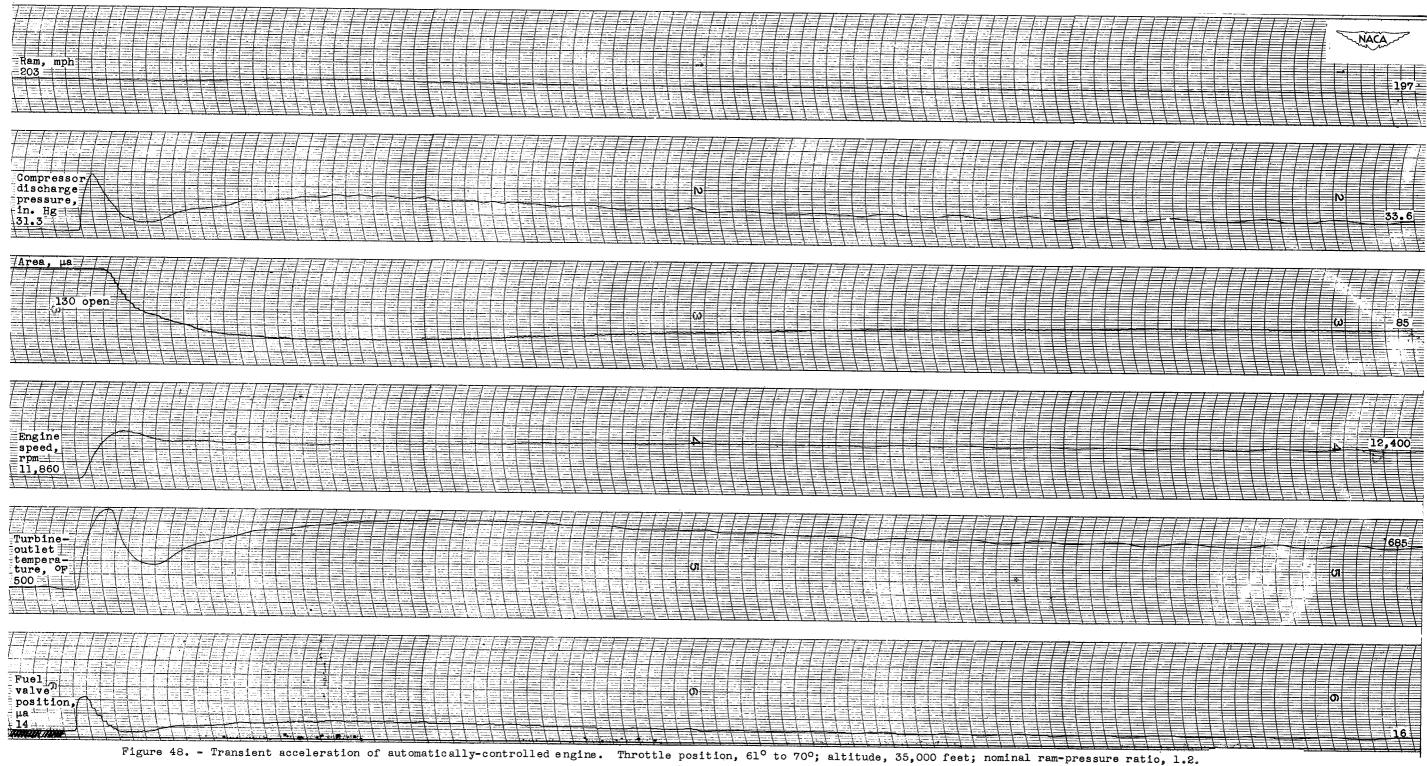


Figure 47. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 60.5° to 64°; altitude, 35,000 feet; nominal ram-pressure ratio, 1.2.



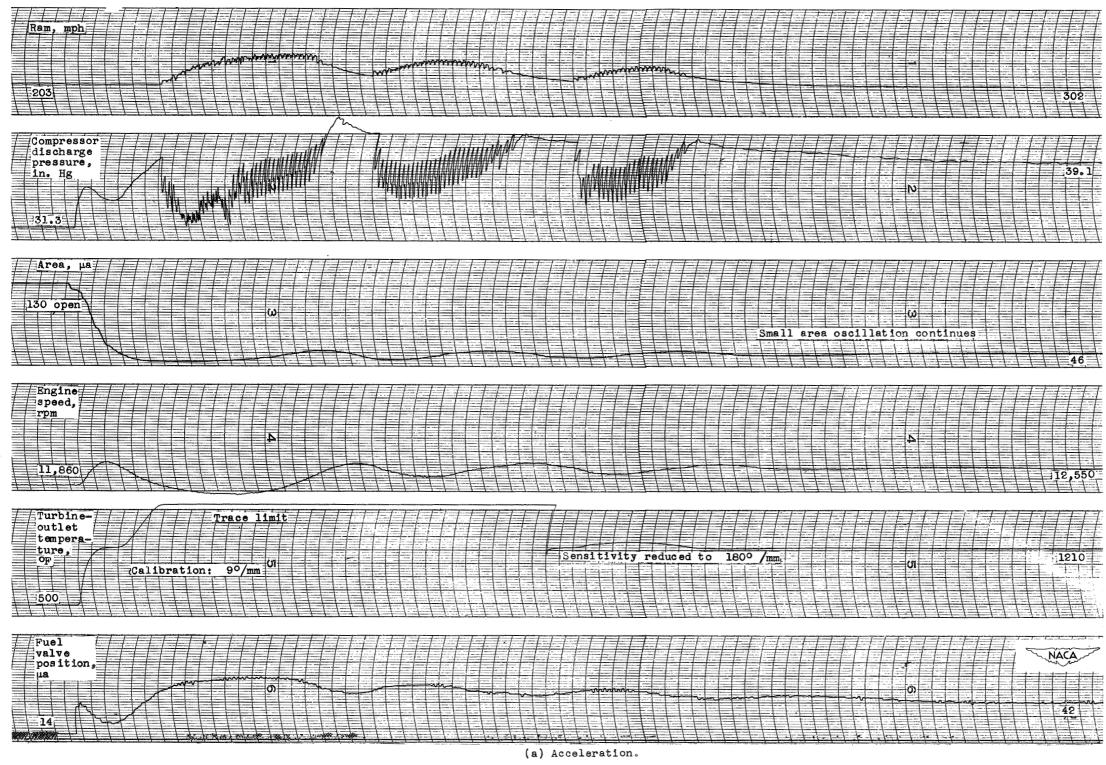


Figure 49. - Transient operation of automatically-controlled engine. Throttle position, 61° to 84°; altitude, 35,000 feet; nominal ram-pressure ratio, 1.2.

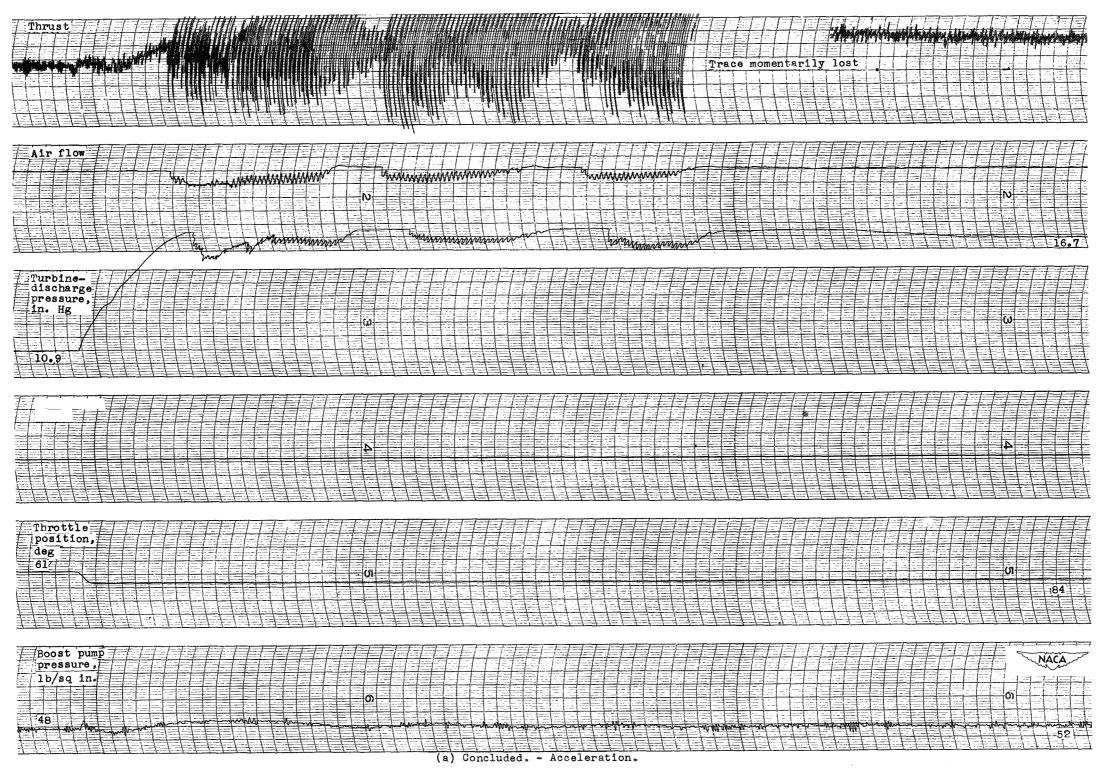


Figure 49. - Continued. Transient operation of automatically-controlled engine. Throttle position, 61° to 84°; altitude, 35,000 feet; nominal ram-pressure ratio, 1.2.

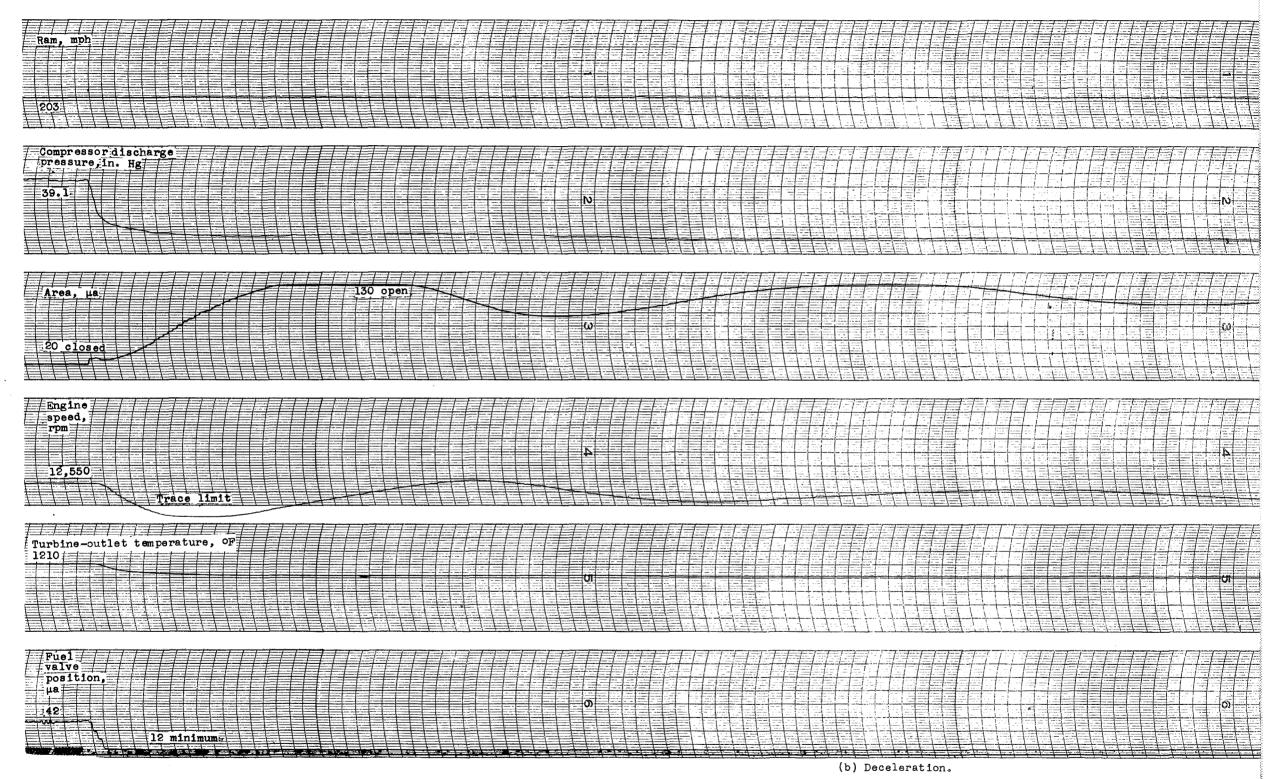


Figure 49. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 61° to 84°; altitude, 35,000 feet; nomis

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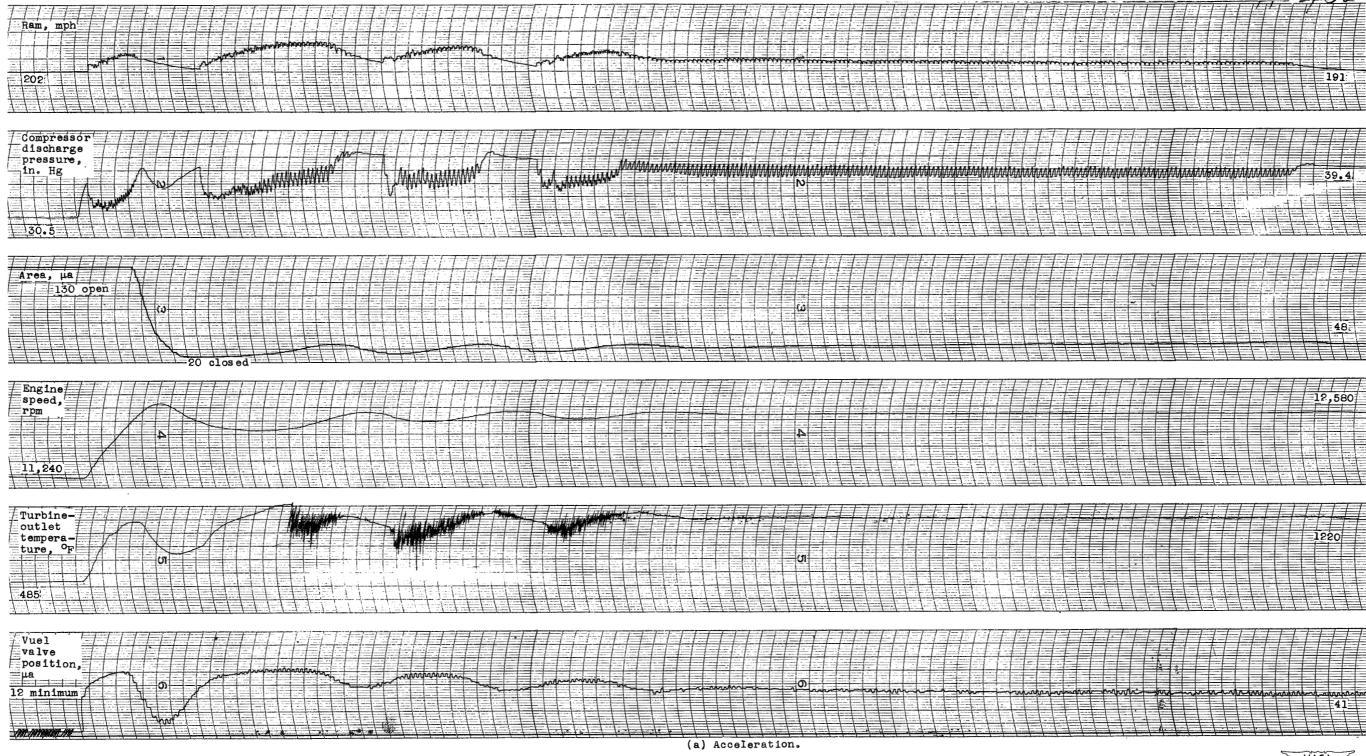


Figure 50. - Transient operation of automatically-controlled engine. Throttle position, 47° to 84°; altitude, 35,000 feet; nominal ram-pressure ratio, 1.2.

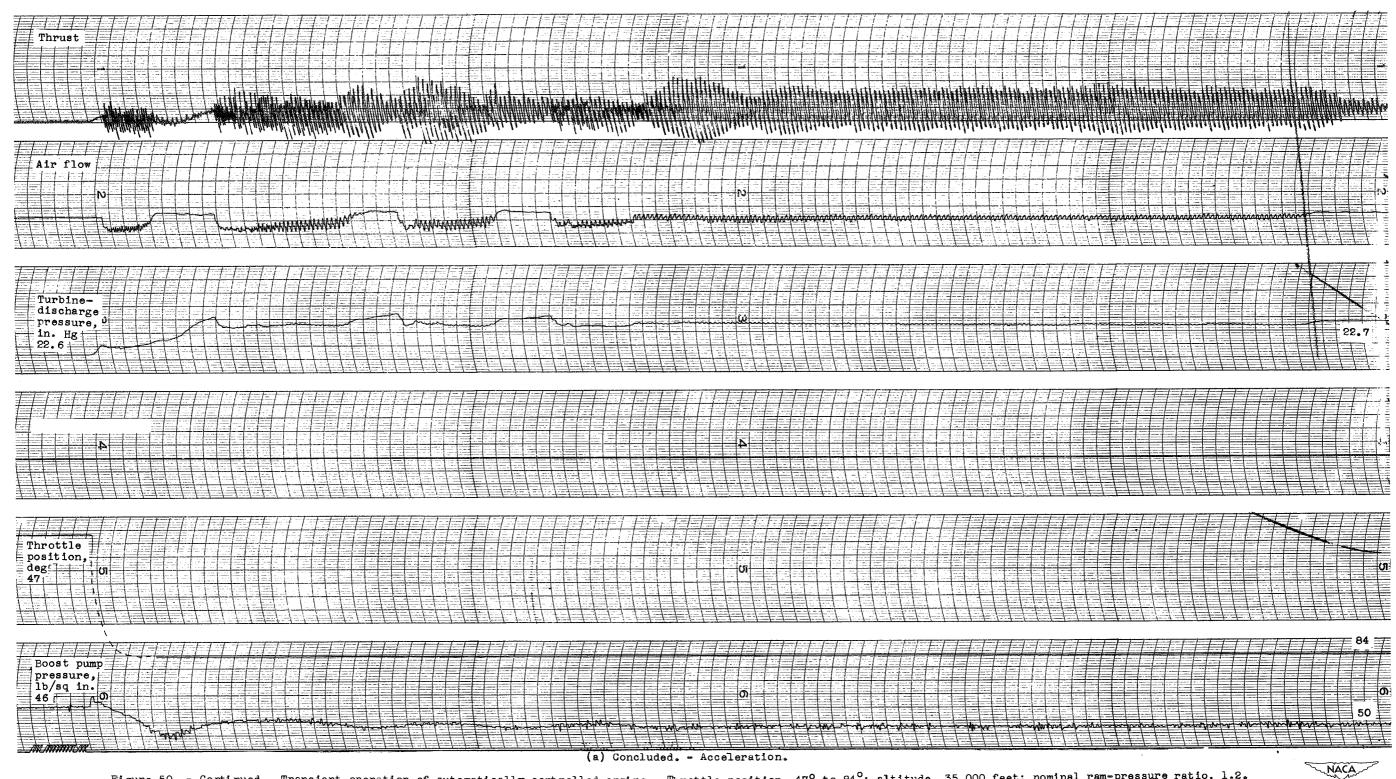


Figure 50. - Continued. Transient operation of automatically-controlled engine. Throttle position, 470 to 840; altitude, 35,000 feet; nominal ram-pressure ratio, 1.2.

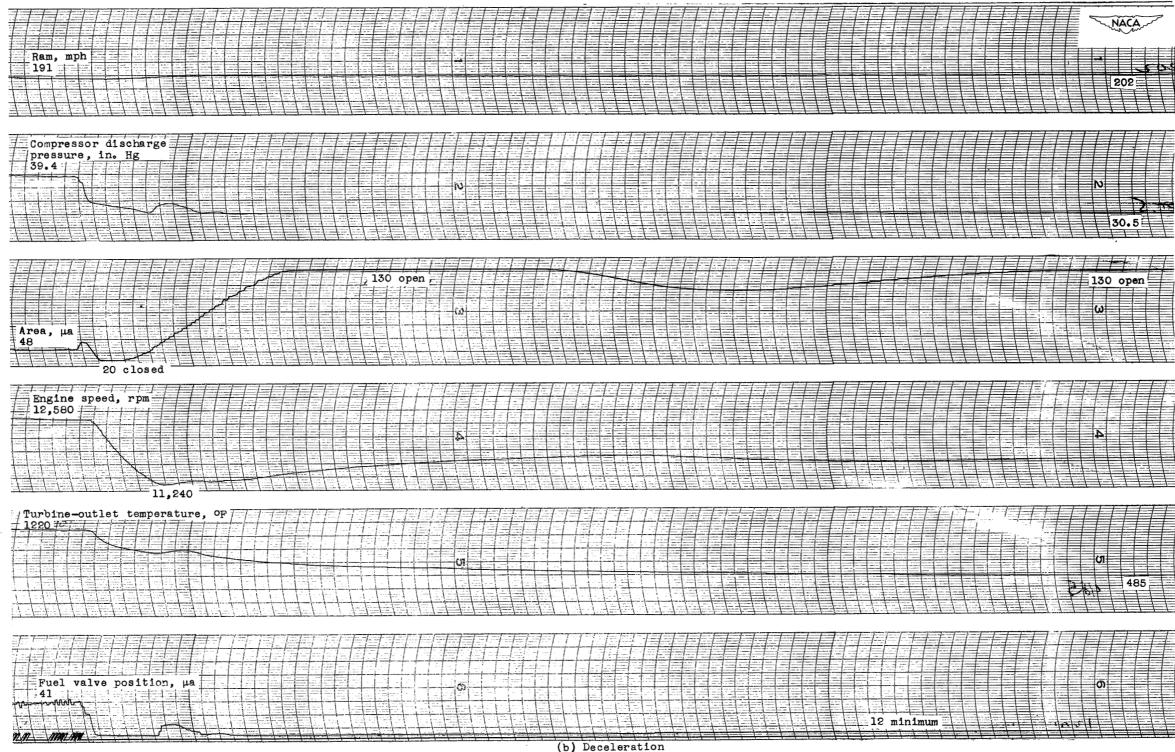


Figure 50. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 47° to 84°; altitude, 35,000 feet: nominal ram-pressure ratio, 1.2.

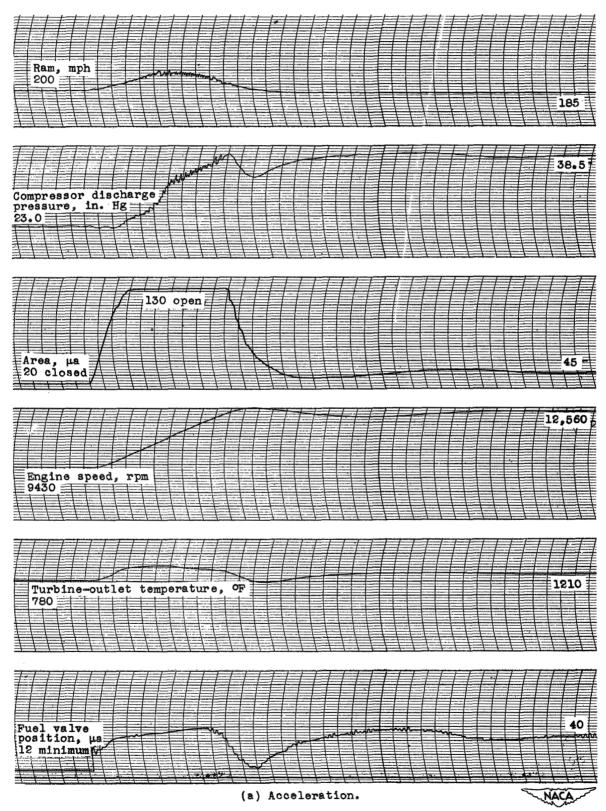


Figure 51. - Transient operation of automatically-controlled engine. Throttle position, 36.5° to 84°; altitude, 35,000 feet; nominal ram-pressure ratio, 12.

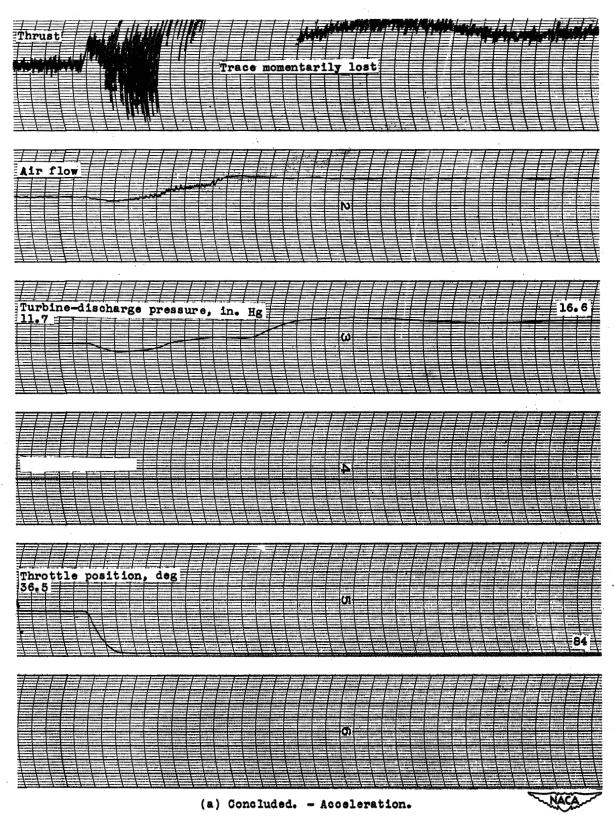


Figure 51. - Continued. Transient operation of automatically-controlled engine. Throttle position, 36.50 to 840; altitude, 35,000 feet; nominal rem-pressure ratio, 1.2.

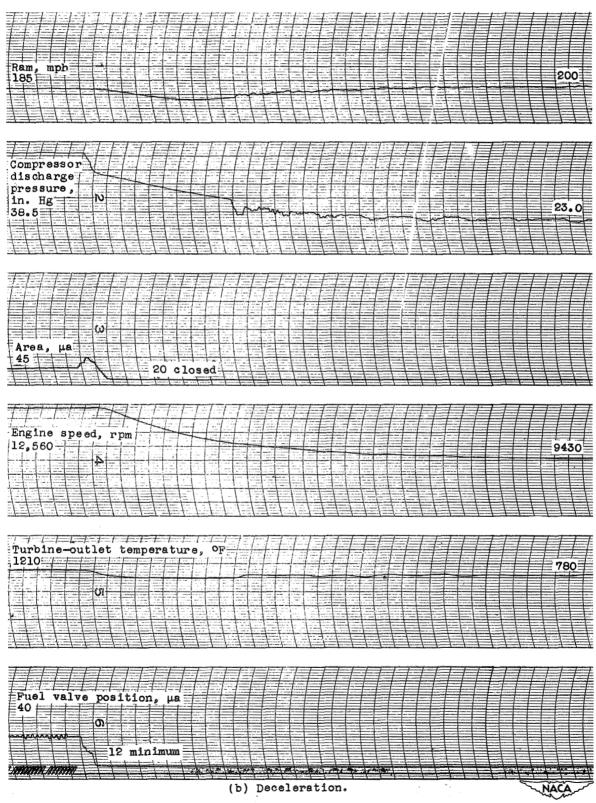


Figure 51. - Continued. Transient operation of automatically-controlled engine. Throttle position, 36.5° to 84°; altitude, 35,000 feet; nominal ram-pressure ratio, 1.2.

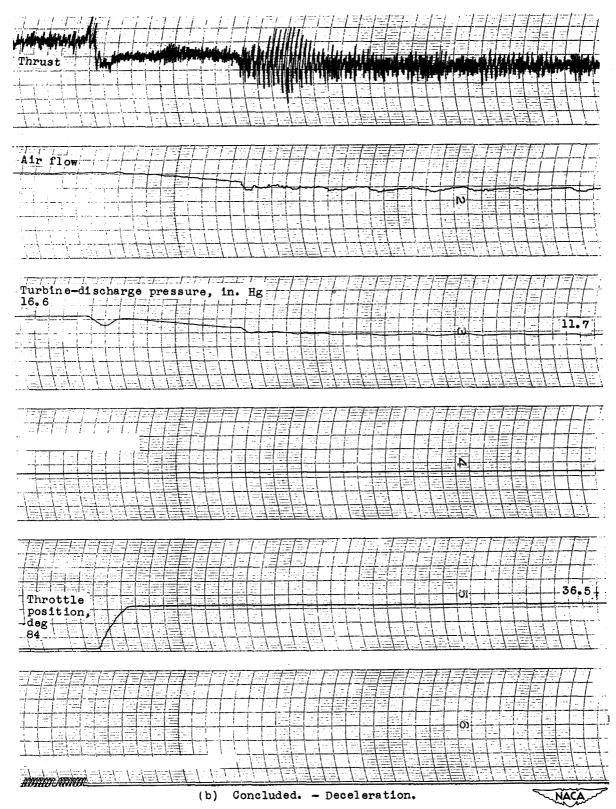
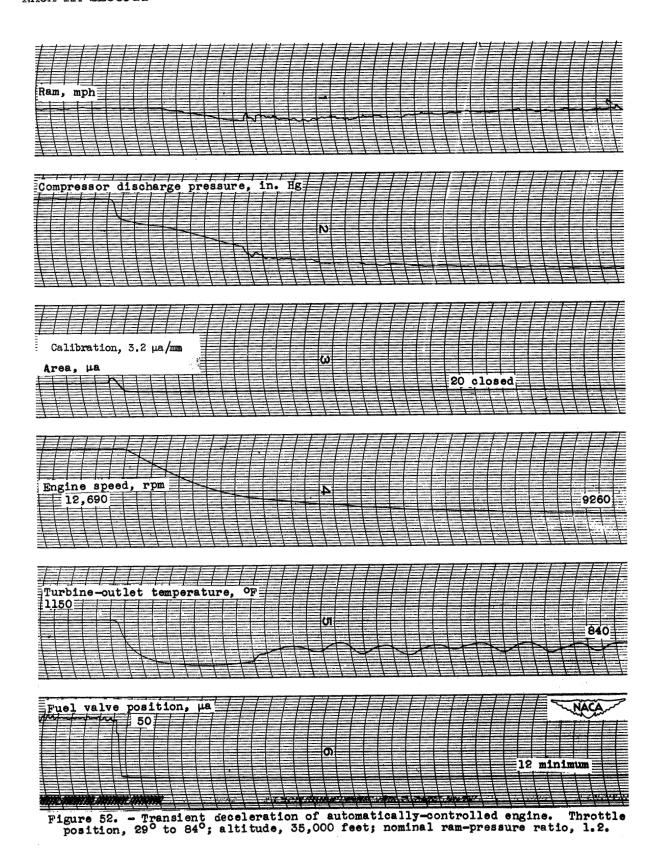


Figure 51. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 36.5° to 84°:, altitude, 35,000 feet; nominal ram-pressure ratio, 1.2.



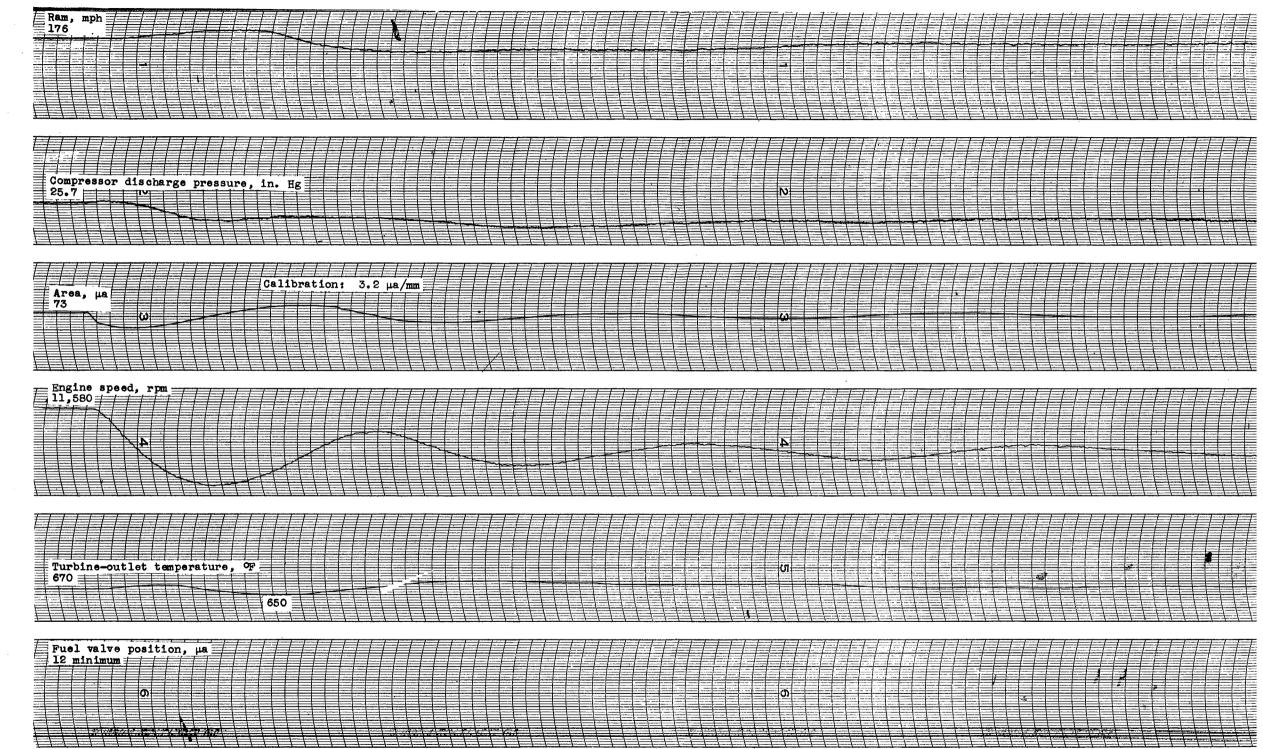
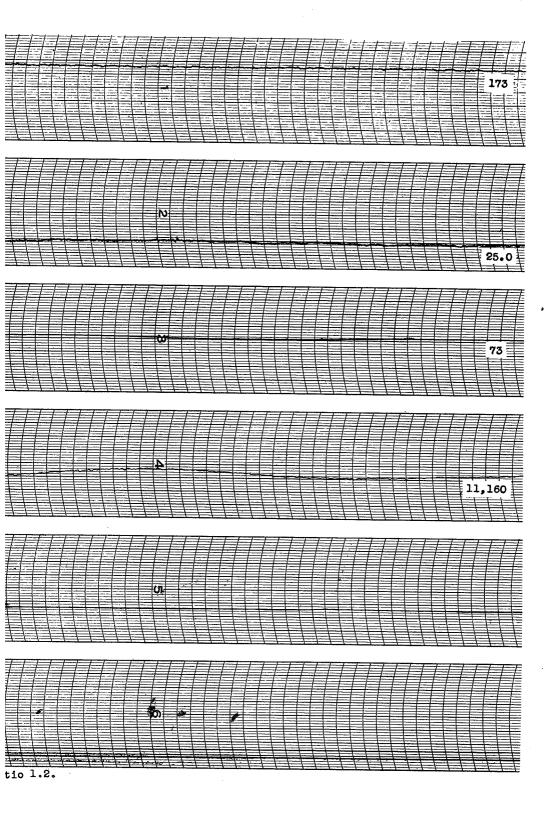


Figure 53. - Transient deceleration of automatically-controlled engine. Throttle position, 42° to 43.5°; altitude, 40,000 feet; nominal ram-pressure



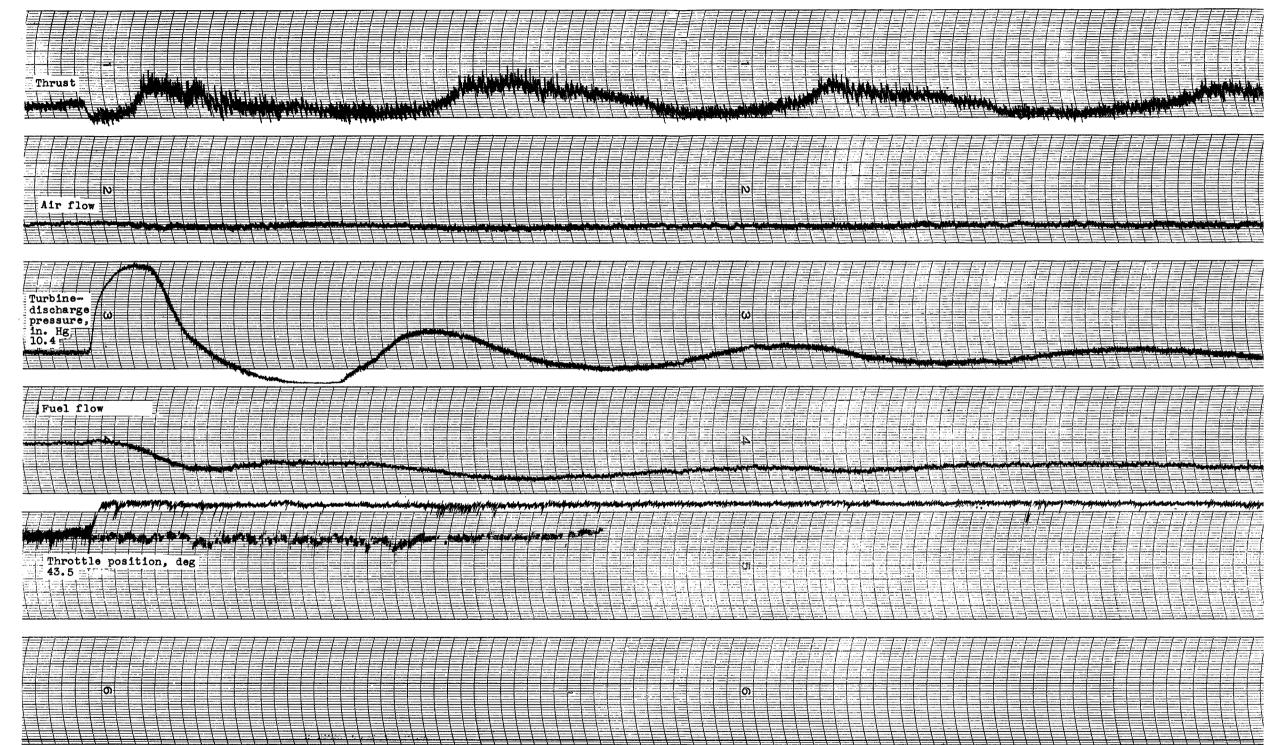
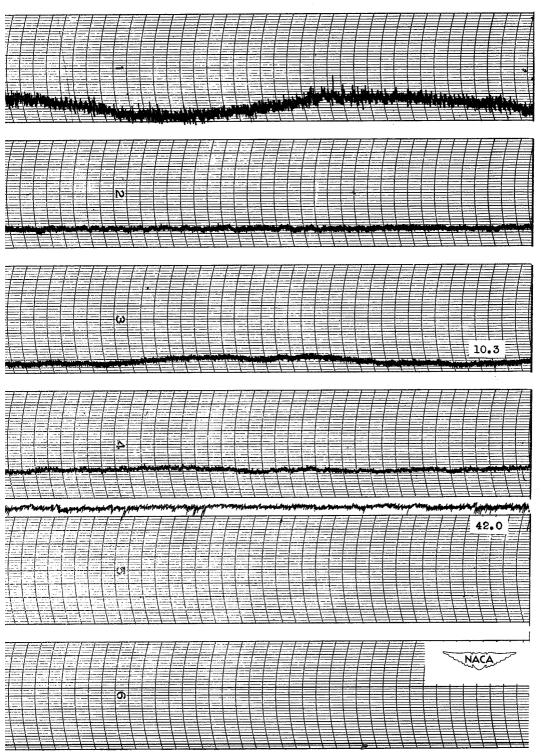


Figure 53. - Concluded. Transient deceleration of automatically-controlled engine. Throttle position, 42° to 43.5°; altitude, 40,000 feet; nominal



ram-pressure ratio, 1.2.

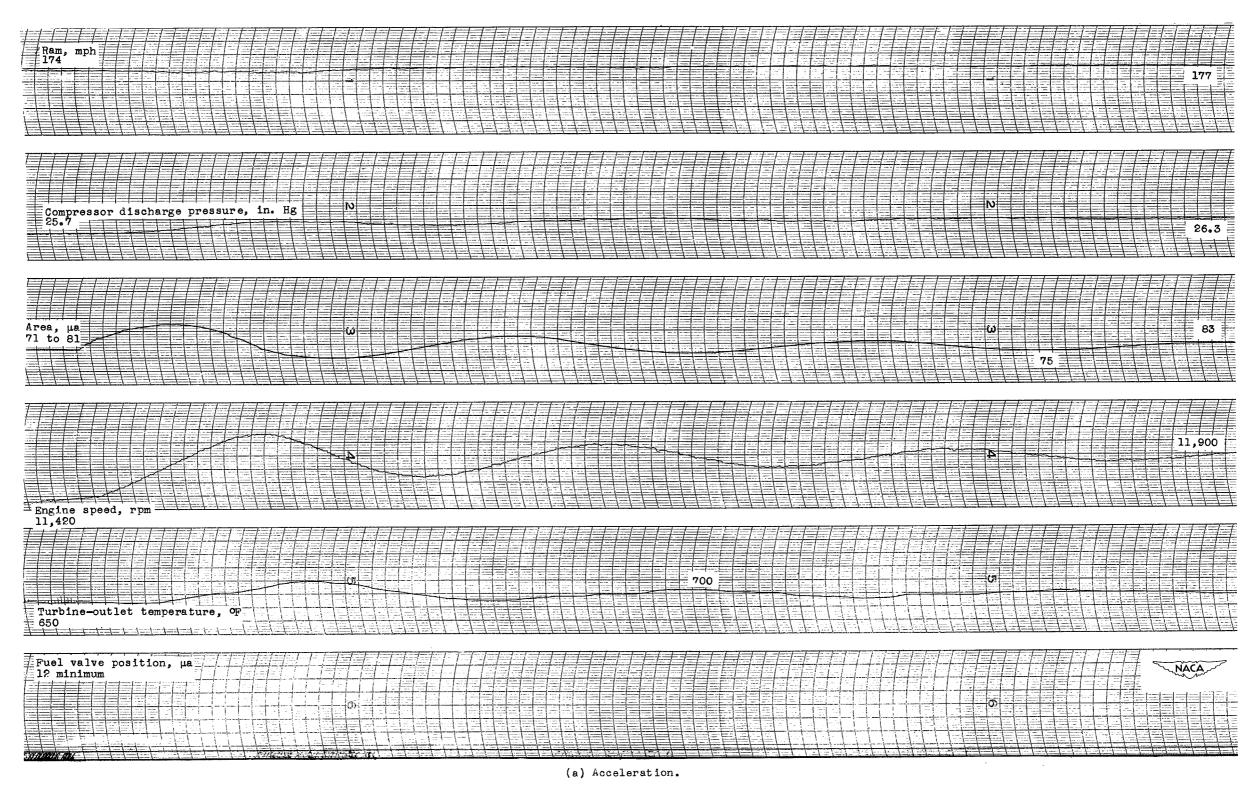


Figure 54. - Transient operation of automatically-controlled engine. Throttle position, 42° to 50.5°; altitude, 40,000 feet; nominal ram-pressure ratio, 1.2.

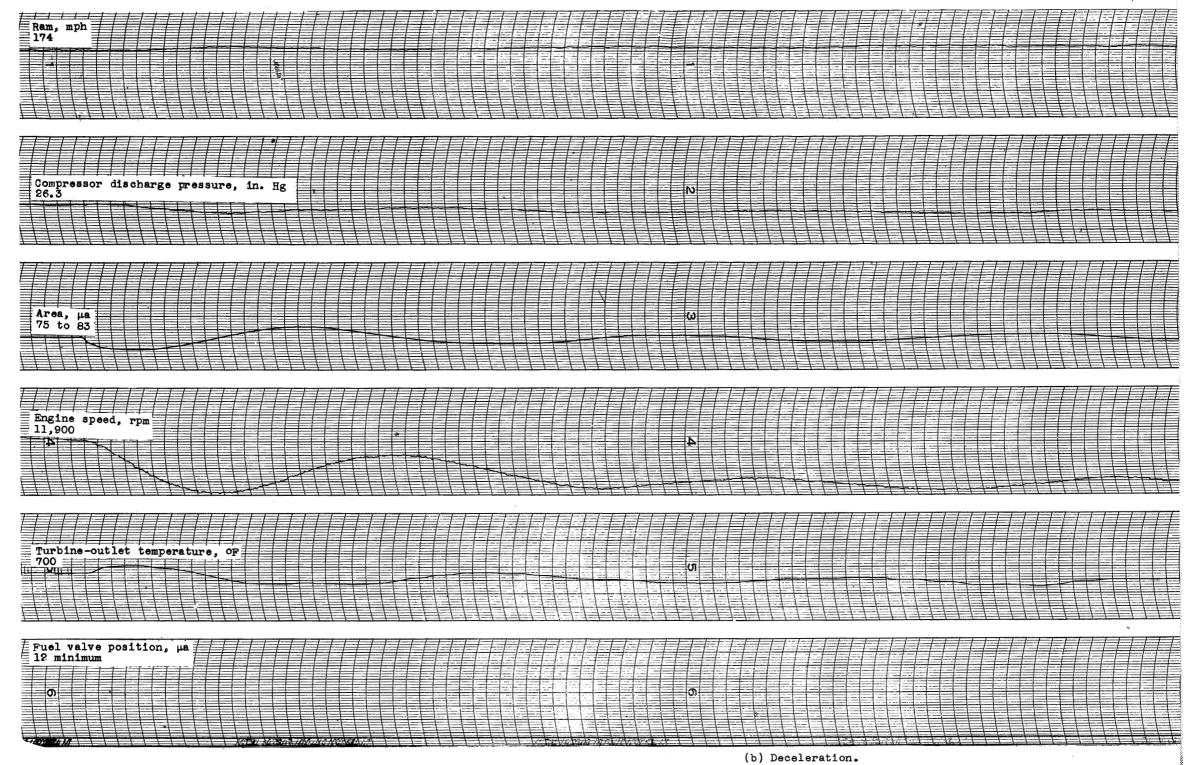
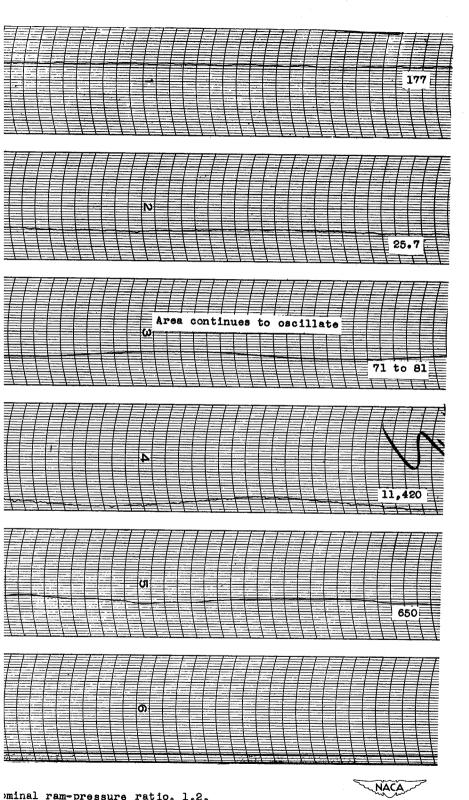
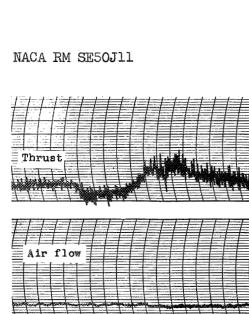


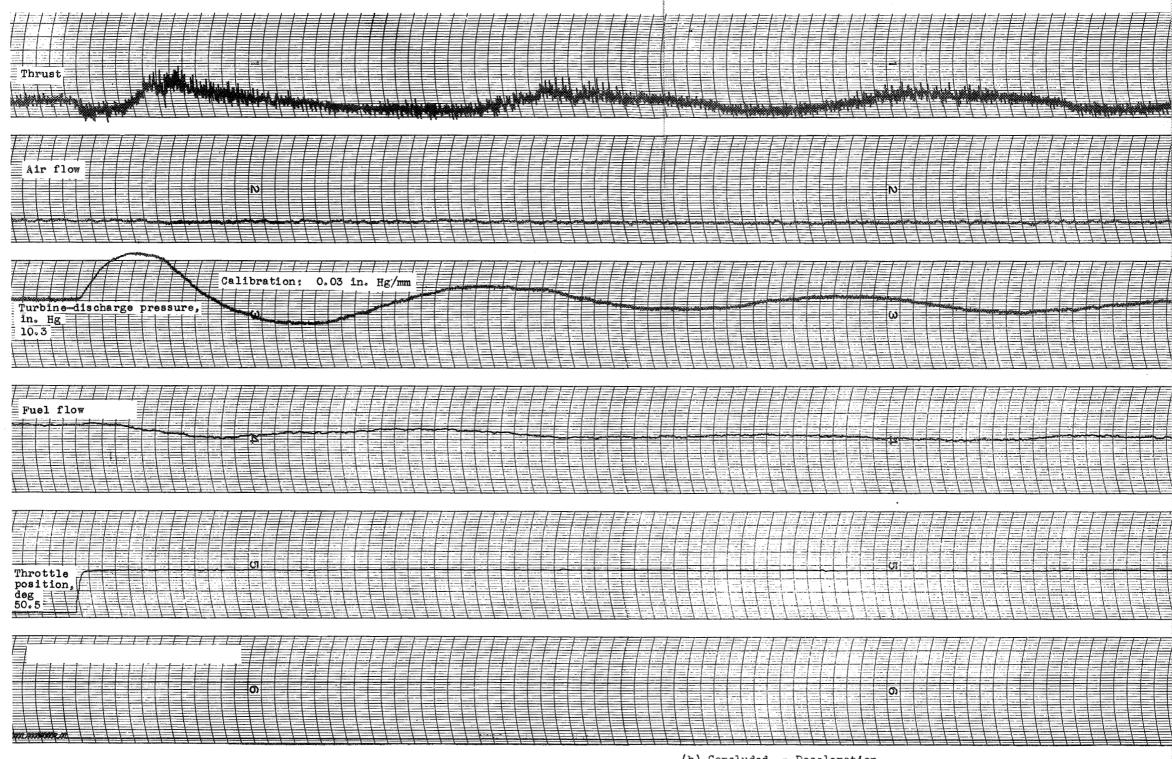
Figure 54. - Continued. Transient operation of automatically-controlled engine. Throttle position, 42° to 50.5°; altitude, 40,000 feet;



minal ram-pressure ratio, 1.2.

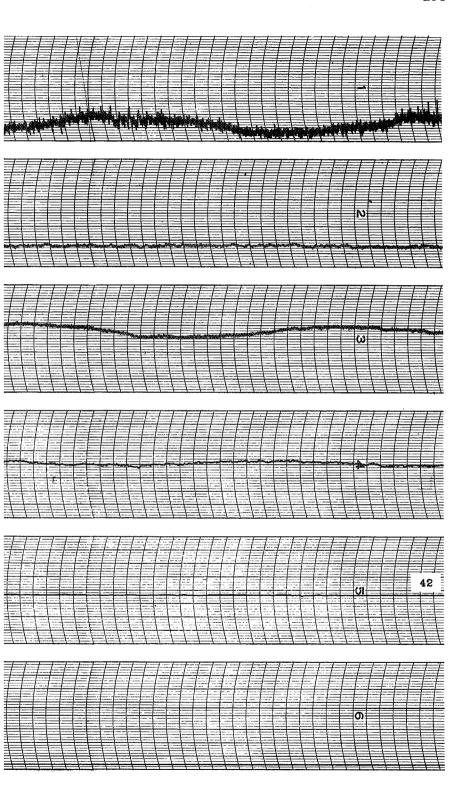


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(b) Concluded. - Deceleration.

Figure 54. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 42° to 50.5°; altitude, 40,000 feet; r



ominal ram-pressure ratio, 1.2.

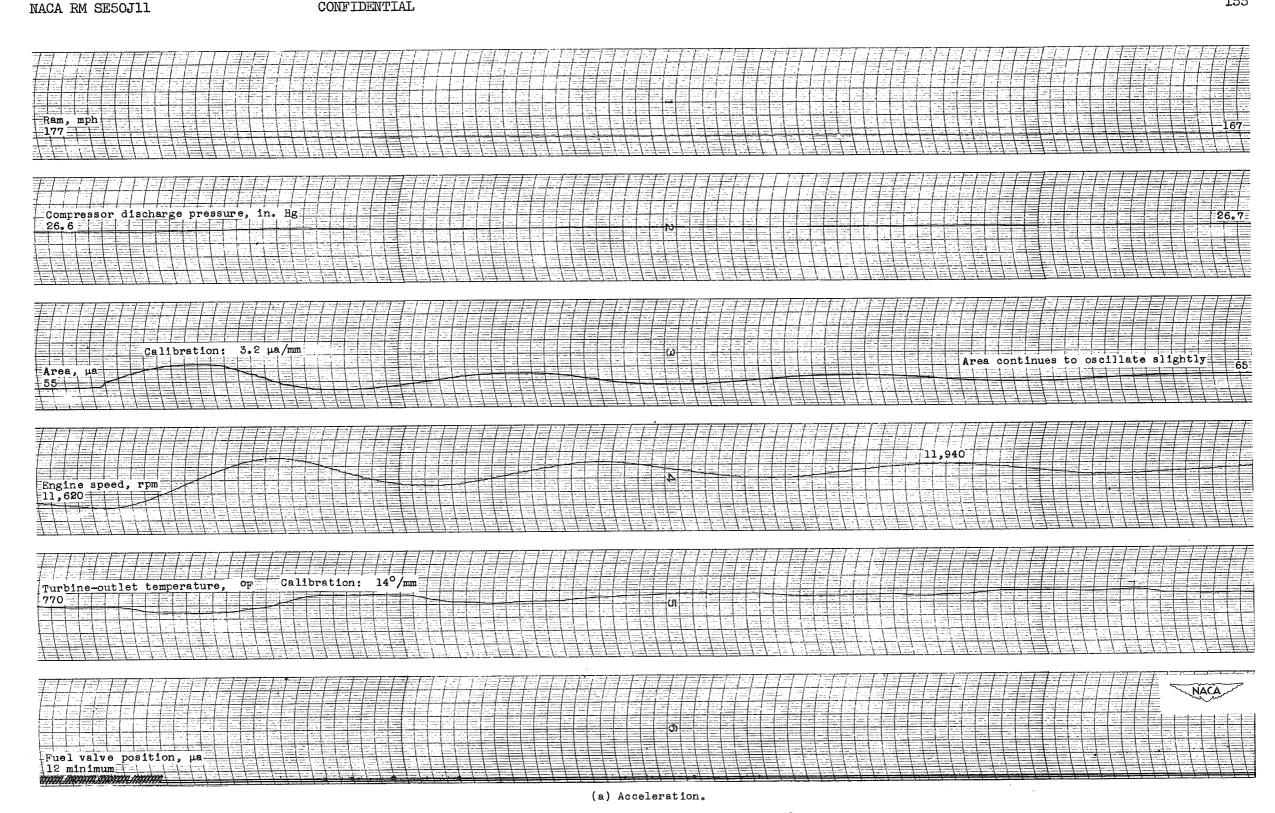


Figure 55. - Transient operation of automatically-controlled engine. Throttle position, 46.50 to 550; altitude, 40,000 feet; nominal ram-pressure ratio, 1.2.

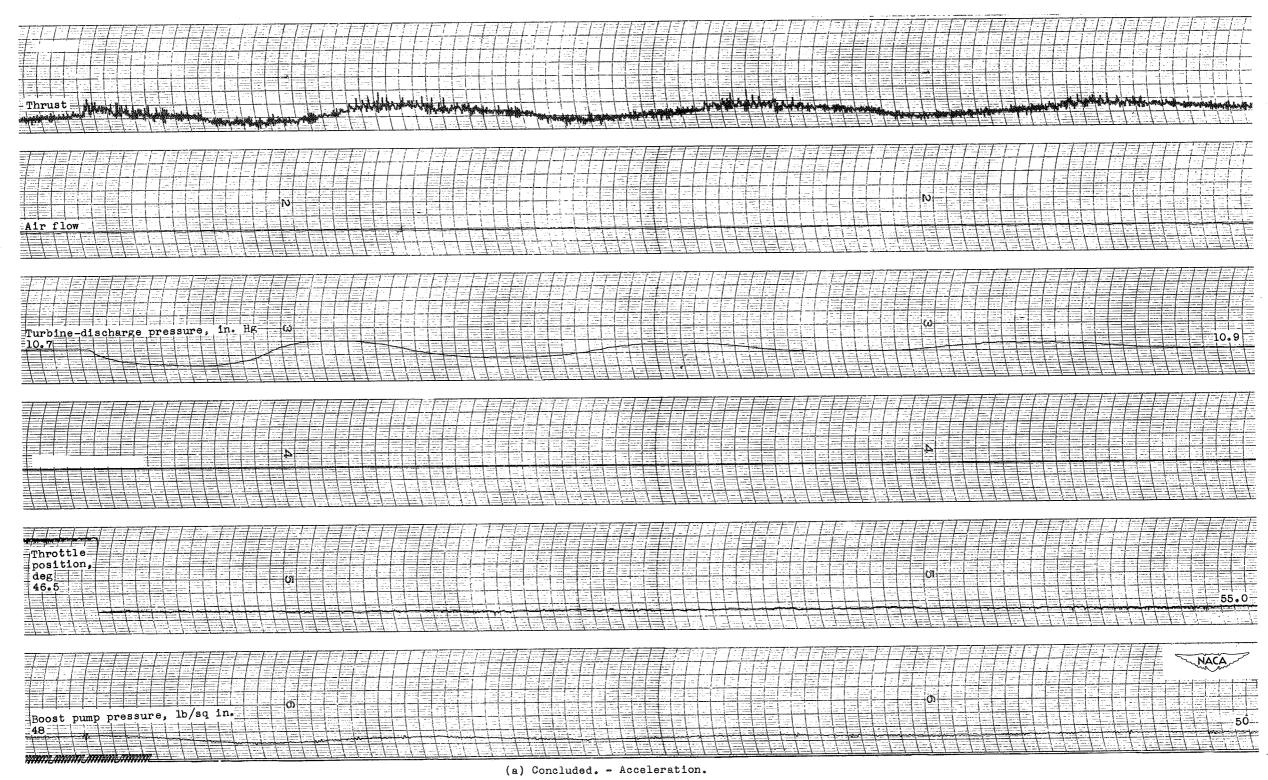


Figure 55. - Continued. Transient operation of automatically-controlled engine. Throttle position, 46.5° to 55°; altitude, 40,000 feet; nominal ram-pressure ratio, 1.2.

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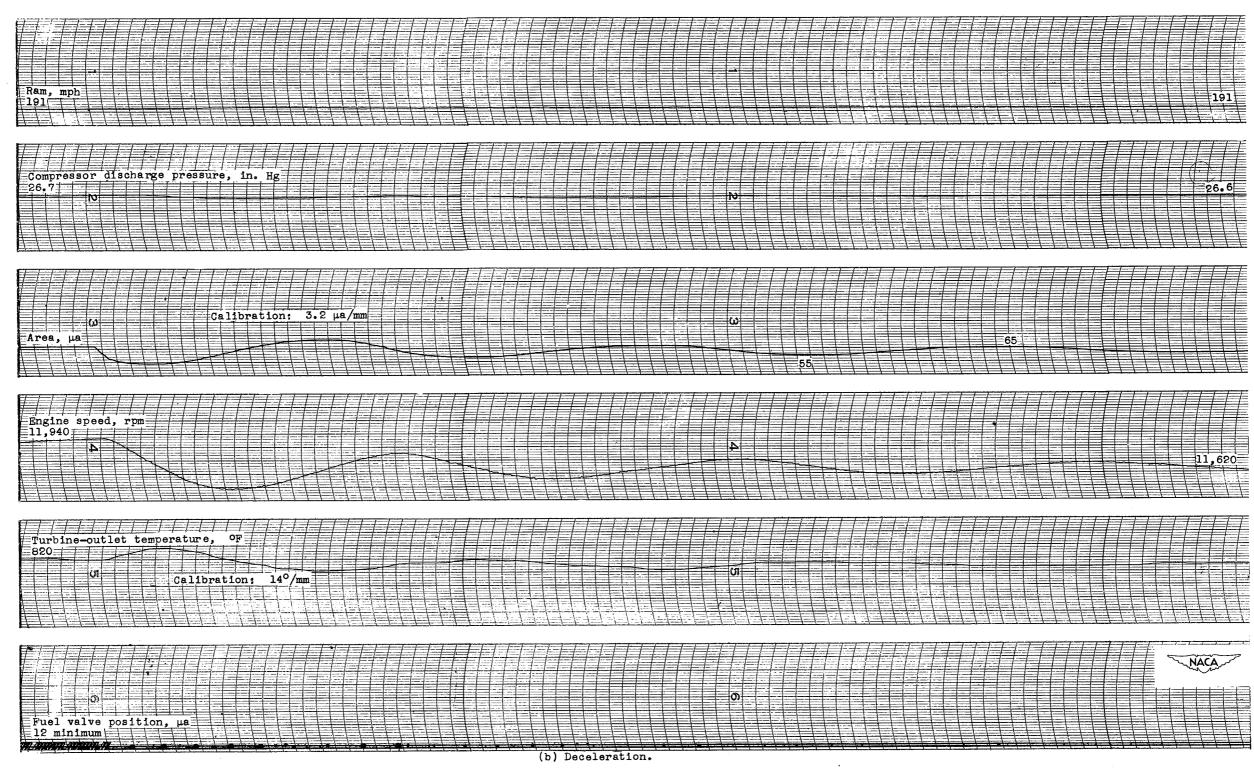


Figure 55. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 46.5° to 55°; altitude, 40,000 feet; nominal ram-pressure ratio, 1.2.

Figure 56. - Transient operation of automatically-controlled engine. Throttle position, 47° to 66°; altitude, 40,000 feet; nominal ram-pressure ratio, 1.2.

(a) Acceleration.

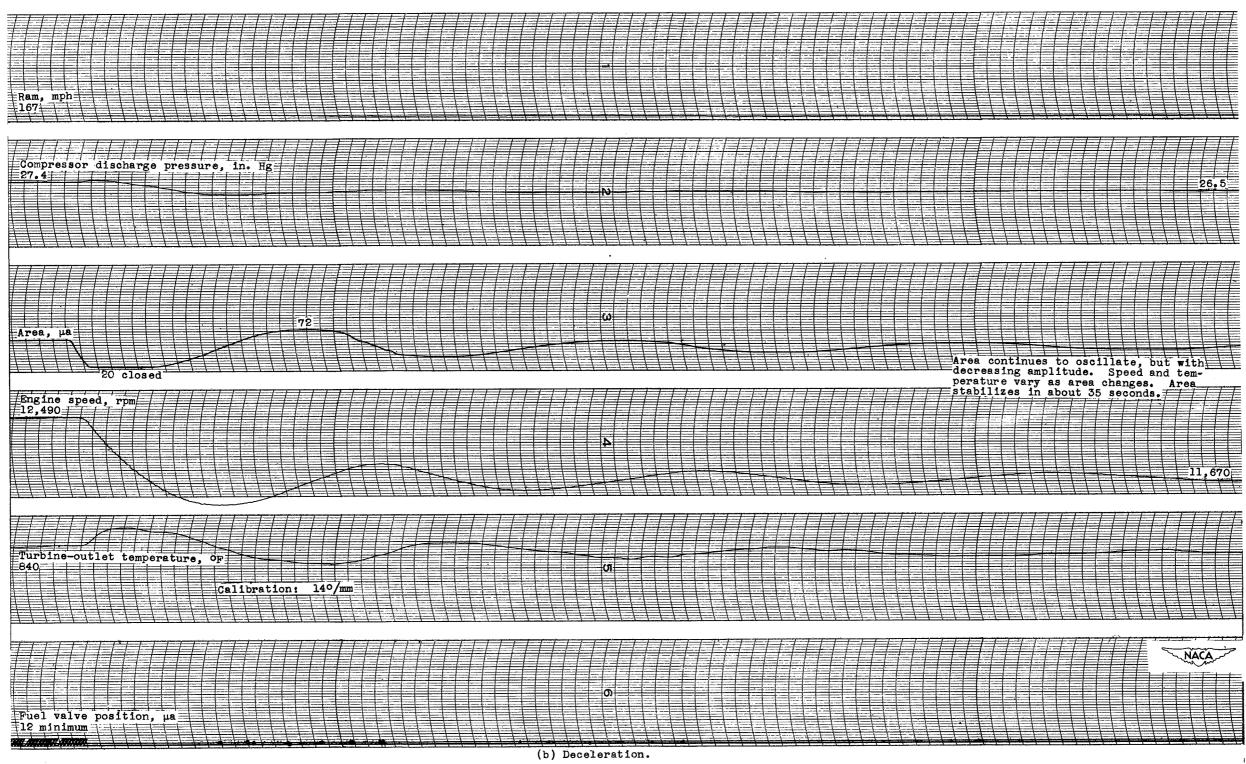


Figure 56. - Continued. Transient operation of automatically-controlled engine. Throttle position, 470 to 660; altitude, 40,000 feet; nominal ram-pressure ratio, 1.2.

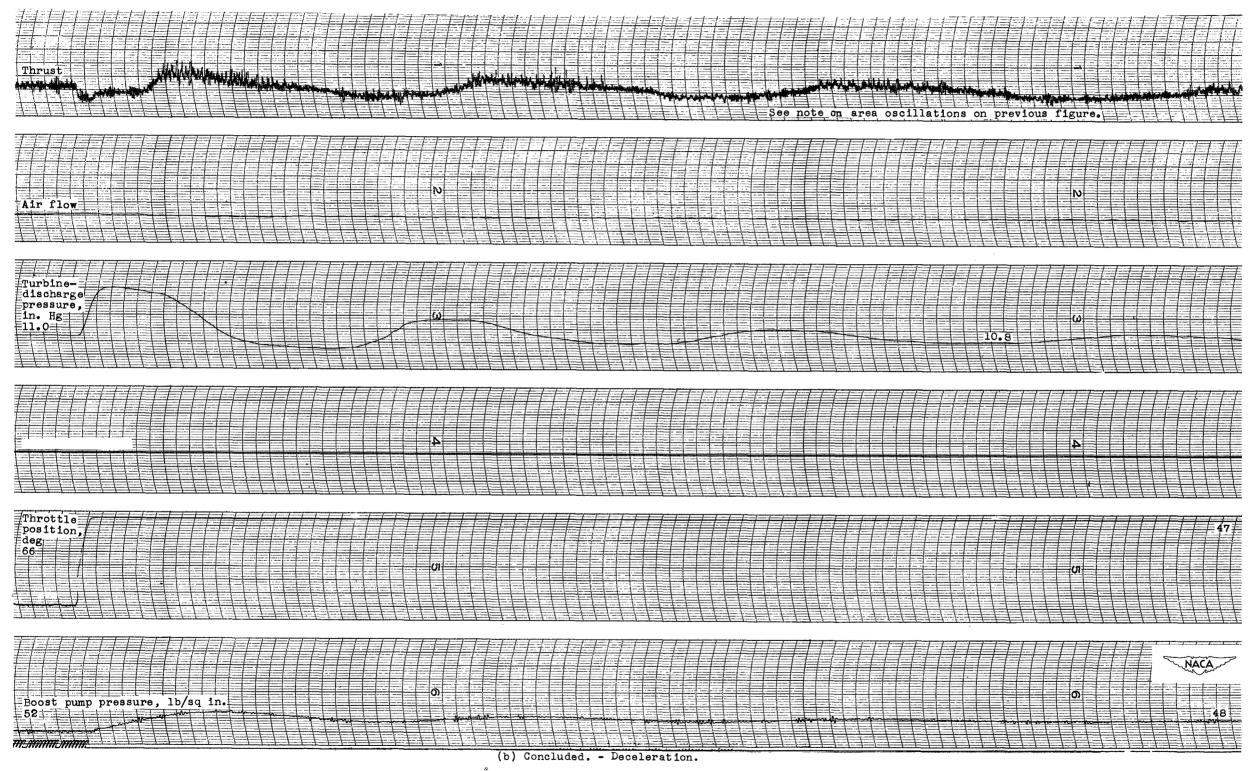


Figure 56. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 47° to 66°; altitude, 40,000 feet; nominal ram-pressure ratio, 1.2.

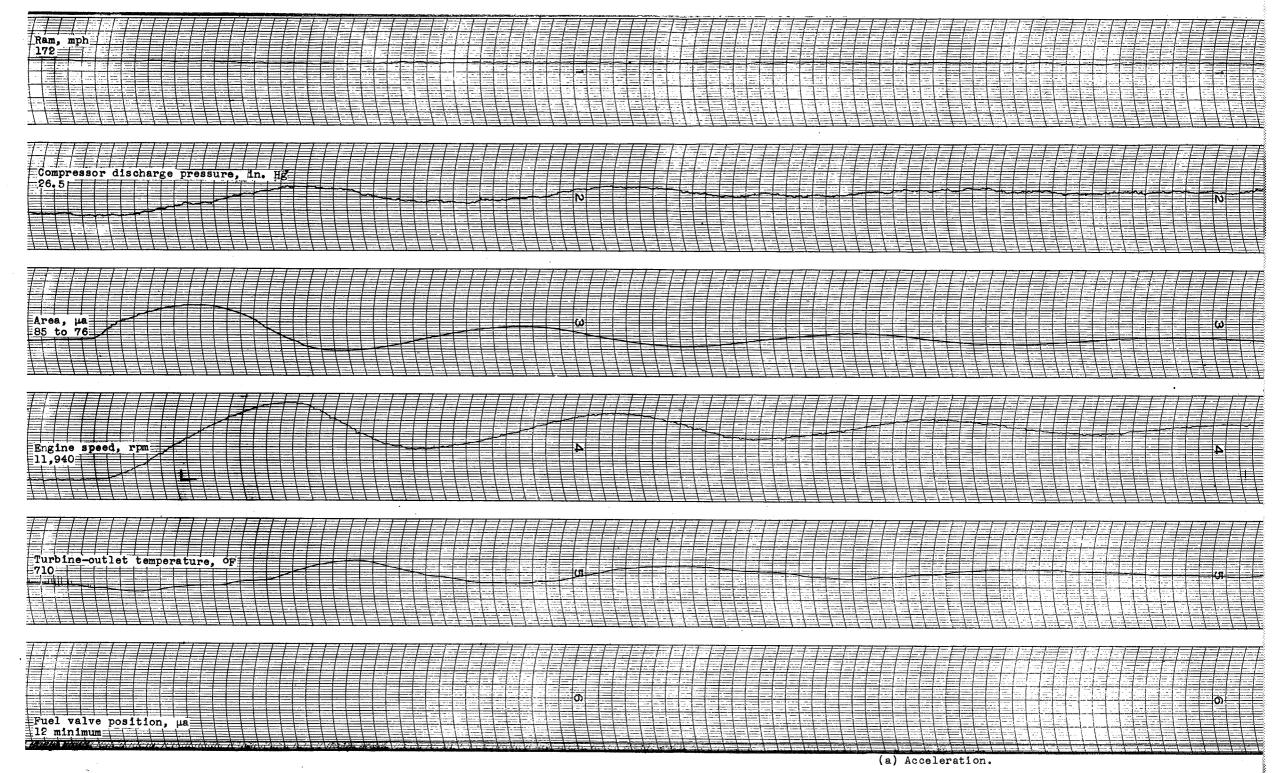
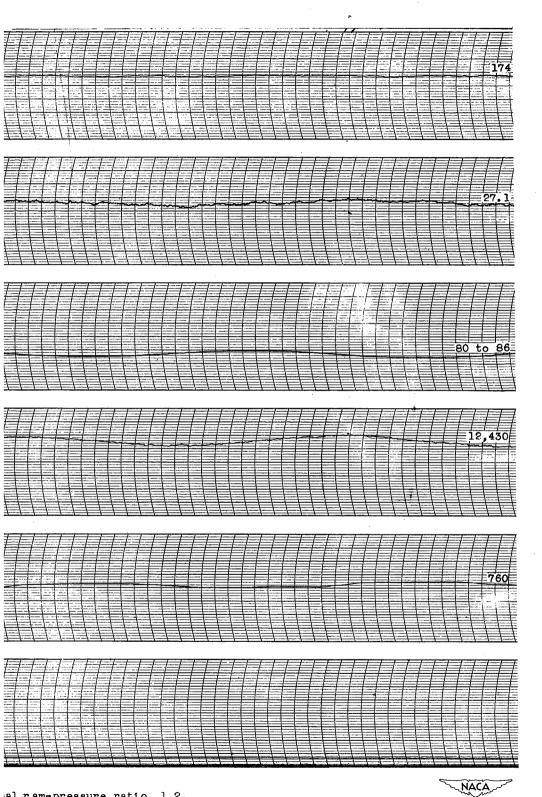


Figure 57. Transient operation of automatically-controlled engine. Throttle position, 50° to 62°; altitude, 40,000 feet; nomi



al ram-pressure ratio, 1.2.

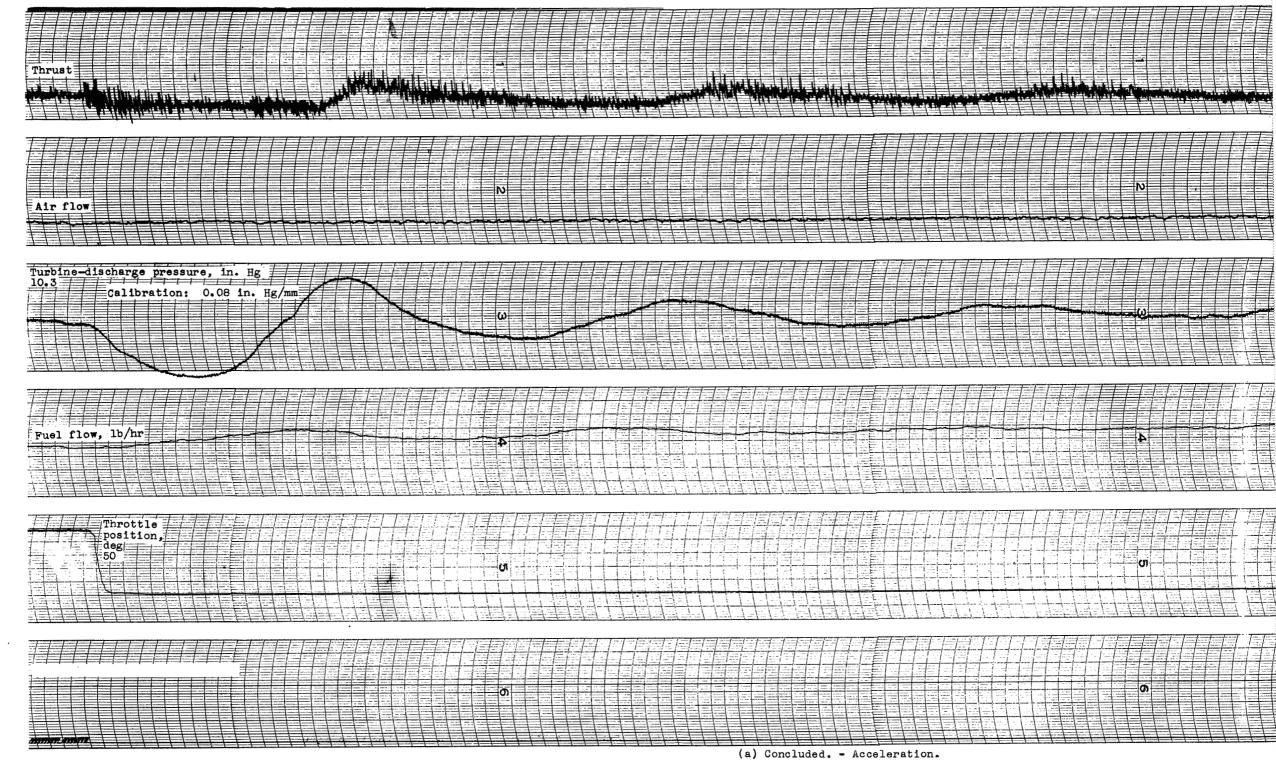
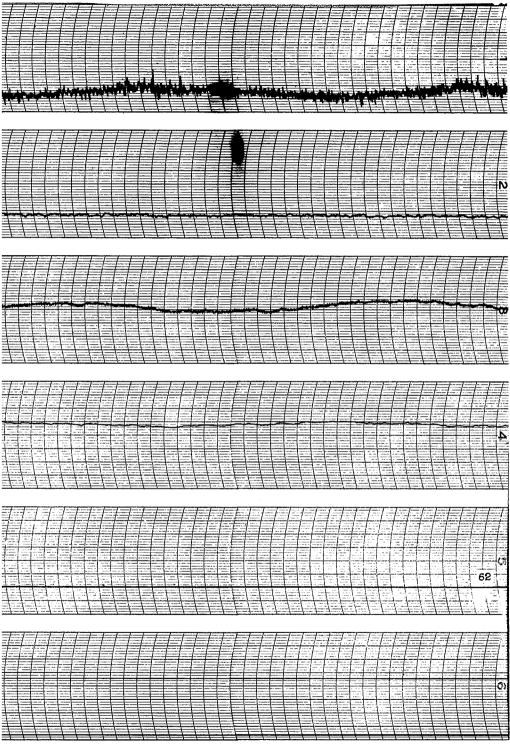


Figure 57. - Continued. Transient operation of automatically-controlled engine. Throttle position, 50° to 62°; altitude, 40,000 feet; nominal re





-pressure ratio, 1.2.

Figure 57. - Concluded. Transient operation of automatically-controlled engine. Throttle position, 50° to 62°; altitude, 40,000 feet; nominal ram-pressure ratio, 1.2.

(b) Deceleration.

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Figure 58. - Transient deceleration of automatically-controlled engine. Throttle position, 47° to 84°; altitude, 40,000 feet; nominal ram-pressure ratio, 1.2.

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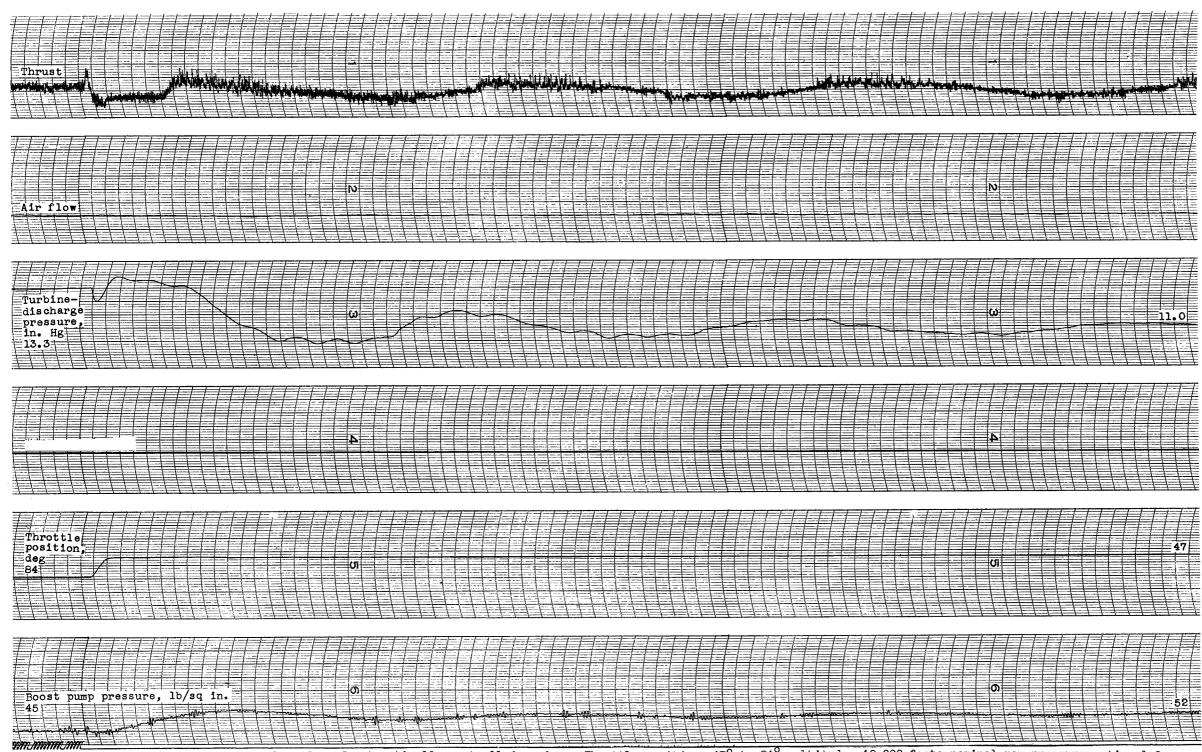


Figure 58. - Concluded. Transient deceleration of automatically-controlled engine. Throttle position, 47° to 84°; altitude, 40,000 feet; nominal ram-pressure ratio, 1.2.

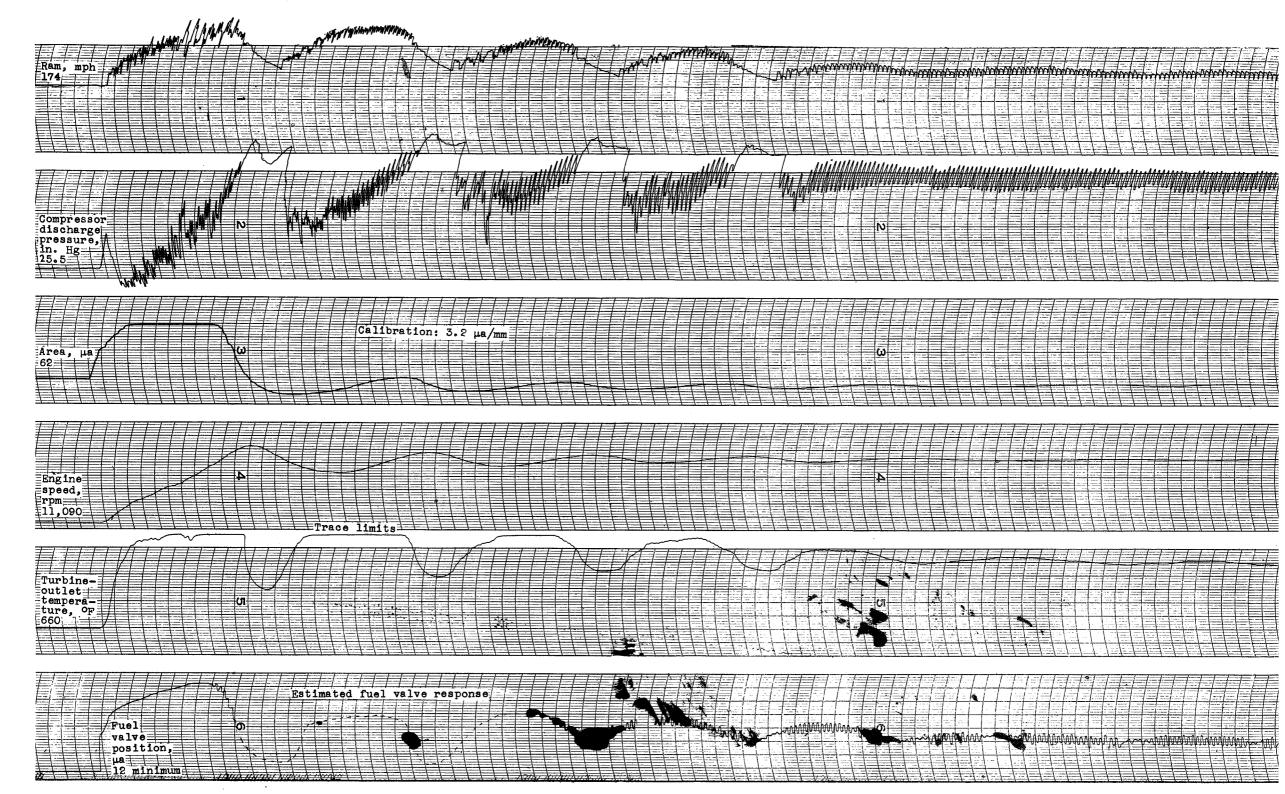


Figure 59. - Transient acceleration of automatically-controlled engine. Throttle position, 41° to 84°; altitude, 40,000 feet; nominal ram-pres

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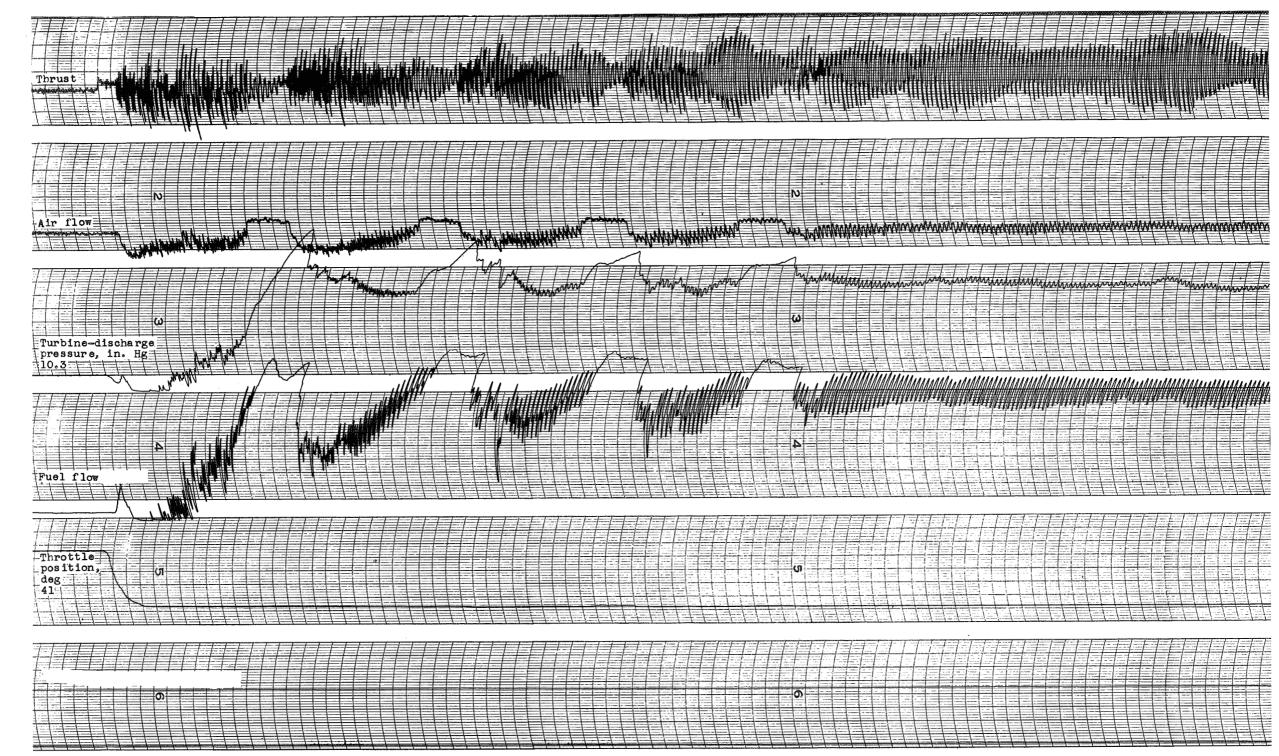
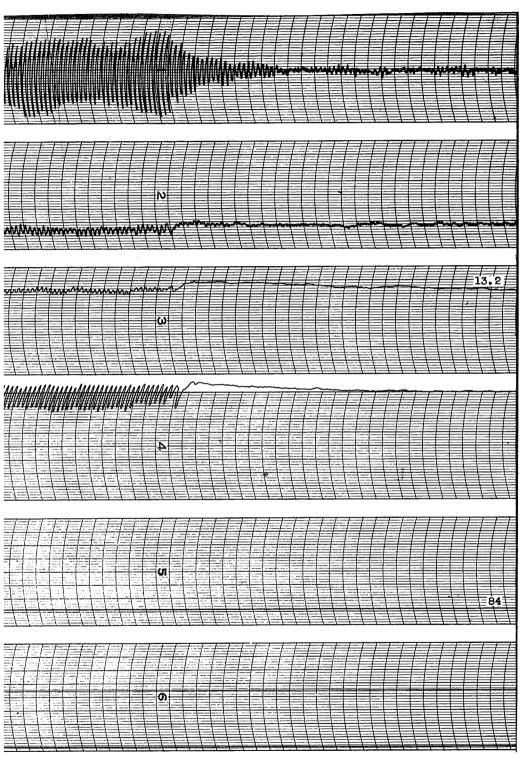


Figure 59. - Concluded. Transient acceleration of automatically-controlled engine. Throttle position, 41° to 84°; altitude, 40,000 feet; nomin



al ram-pressure 1.2.

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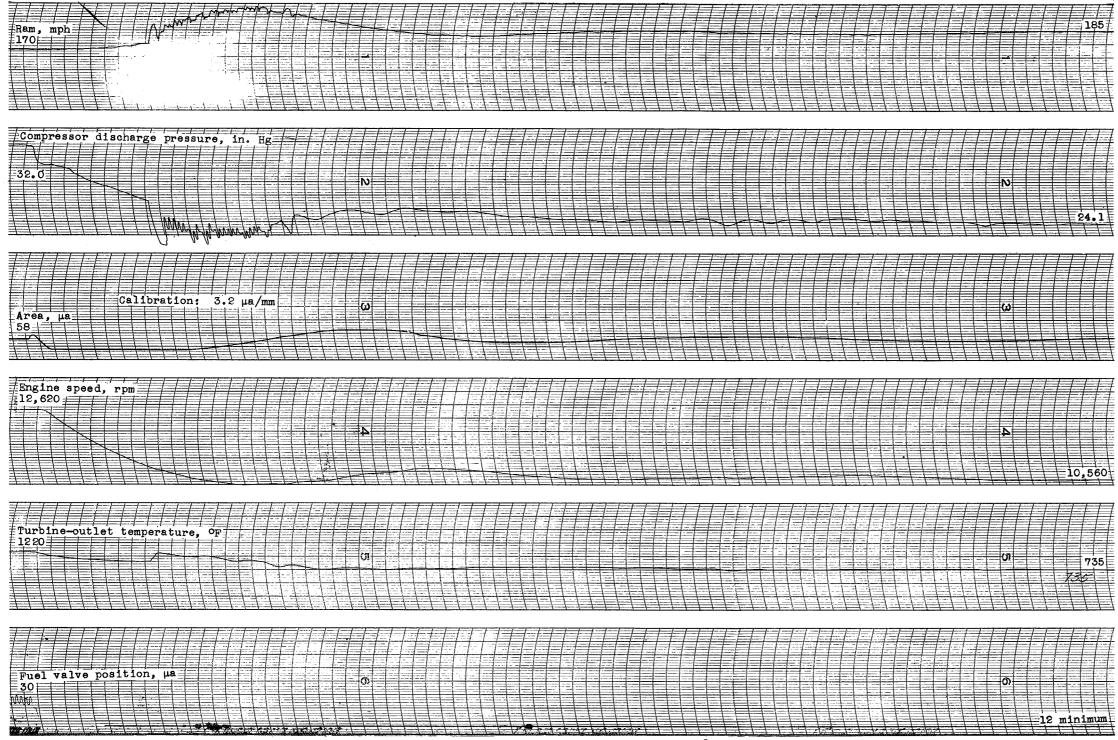


Figure 60. - Transient deceleration of automatically-controlled engine. Throttle position, 39° to 84°; altitude, 40,000 feet; nominal ram-pressure ratio, 1.2.

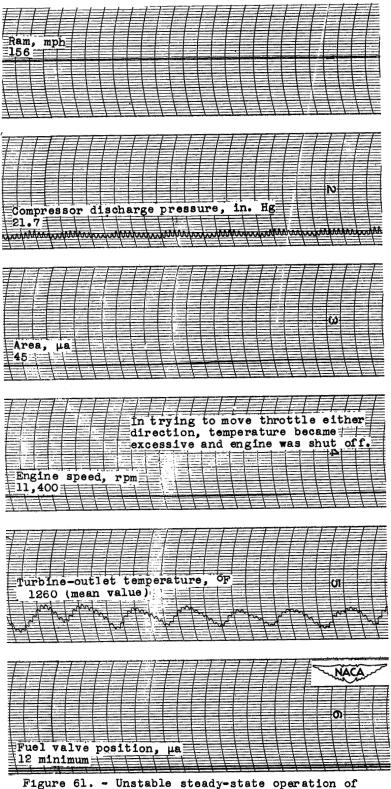


Figure 61. - Unstable steady-state operation of automatically-controlled engine. Throttle position, 47°; altitude, 45,000 feet; nominal ram pressure ratio, 1.2.

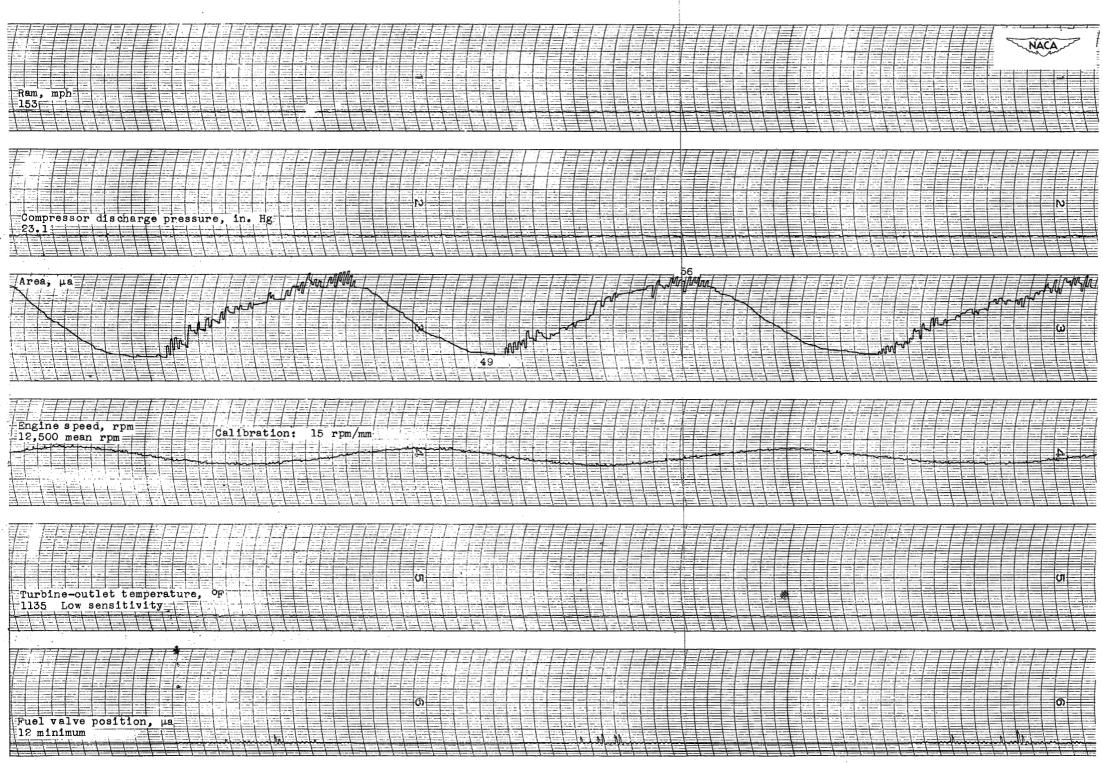


Figure 62. - Unstable steady-state operation of automatically-controlled engine. Throttle position, 80°; altitude, 45,000 feet; nominal ram pressure ratio, 1.2; engine inlet temperature, 47°F.

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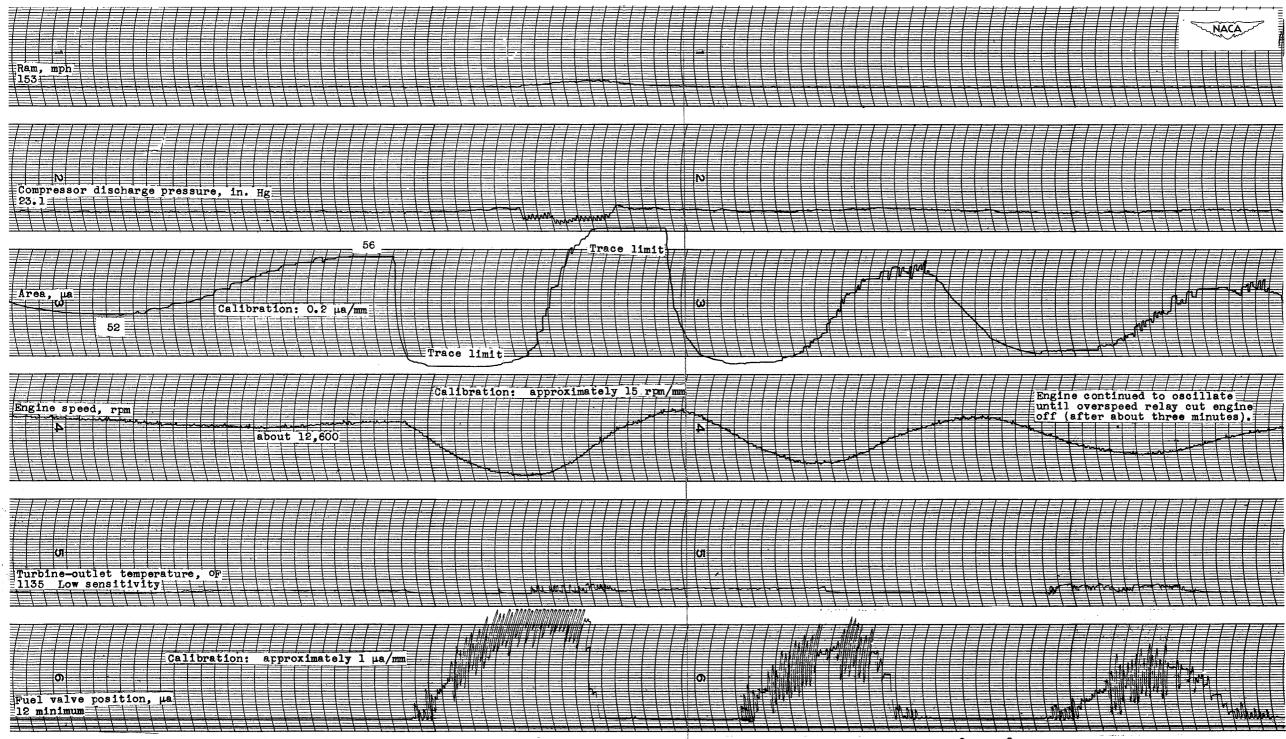


Figure 63. - Transient acceleration of automatically-controlled engine. Steady-state points unstable. Throttle position, 80° to 84°; altitude, 45,000 feet; nominal ram pressure ratio, 1.2; engine inlet temperature, 47°F.

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